

Cornell University
ENGRD 202
Professor Andy Ruina
Spring 2003 Lecture Notes

<u>Page(s)</u>	<u>Date(s)</u>	<u>Topics Covered</u>
2 – 9	01/20 – 01/22	Vectors (Notation, Algebra, etc.)
10 – 18	01/24 – 01/27	Equivalent Force Systems; Couples
19 – 21	01/29 – 01/31	Free Body Diagrams; 2 – D Statics
22 – 24	02/03	3 – D Statics
25 – 29	02/05 – 02/10	2 – D Trusses
30 – 37	02/12 – 02/17	3 – D Trusses; Matrix Algebra
38 – 47	02/17 – 02/24	Machines and Structures
48 – 53	02/24 – 02/26	Center of Mass; Centroids
54 – 57	03/03	Friction
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62 – 70	03/07 – 03/10	Hydrostatics
71 – 74	03/14	Review of Statics
75	03/24	Friction Hints
76 – 85	03/24 – 03/28	Stress and Strain
86 – 92	03/31 – 04/02	Bars in Tension
93 – 97	04/07 – 04/09	Stress on Inclined Sections
98 – 110	04/09 – 04/16	Torsion
111 - 119	04/16 – 04/23	Bending; Shear and Moment Diagrams
120 – 122	04/25	Stress and Strain in Beams
123 – 125	04/28	Moment of Inertia, Deflection of Beams
126 – 129	05/02	Deflection of Beams

ENGRD/TAM 202

JAN 20, 2003

①

ANDY RUINA

see WWW page (google: ruina)

This Course

STATICS: Force & Moment Balance

key skills:

Vectors
Free body diagrams
Solving vector equations

STRENGTH of SOLIDS

How long narrow things stretch,
twist & bend.

Concepts: strain & stress

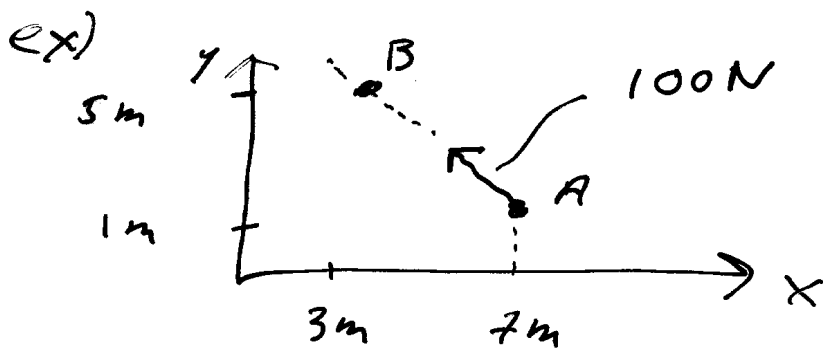
VECTORS: Things represented w/ arrows.

2 basic vectors: 1) relative position $\mathbf{r}_{A/C}$
2) Force \mathbf{F}_D

All other vectors are derived from these 2.

ADVERTISEMENT:

Read about Vectors, Free Body Diagrams
& Statics in on-line book by Ruina & Pratap.



$$\begin{aligned} \underline{F} &= F \cdot \hat{a} \\ &= (100\text{N}) \left(\frac{\underline{r}_{AB}}{|\underline{r}_{AB}|} \right) \\ &= 100\text{N} \frac{-\hat{i} + \hat{j}}{\sqrt{2}} \end{aligned} \quad (3)$$

$$\underline{F} = F_x \hat{i} + F_y \hat{j} = \underbrace{\frac{-100\text{N}}{\sqrt{2}}}_{F_x} \hat{i} + \underbrace{\frac{100\text{N}}{\sqrt{2}}}_{F_y} \hat{j}$$

x & y components of \underline{F}

Vector Notation

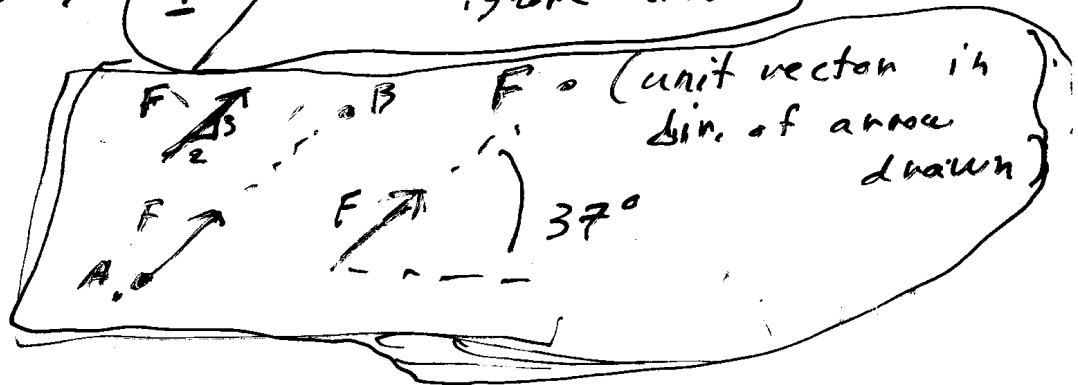
In equations: $\underline{F}, \vec{F}, \hat{F}, \mathbf{F}, \mathbb{F}, (\underline{F})$

unit vectors: $\hat{i}, \hat{j}, \underline{i}, \underline{j}, \hat{a}, \underline{a}$

In pictures:

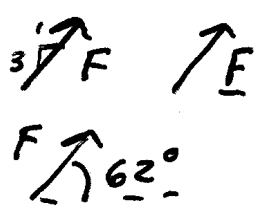
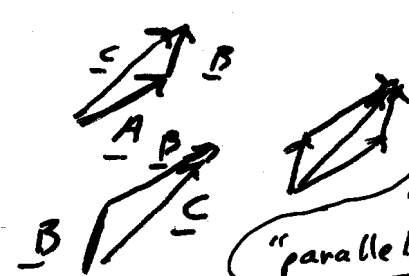

\underline{F} ignore arrow

dir. needs to be clear from picture.



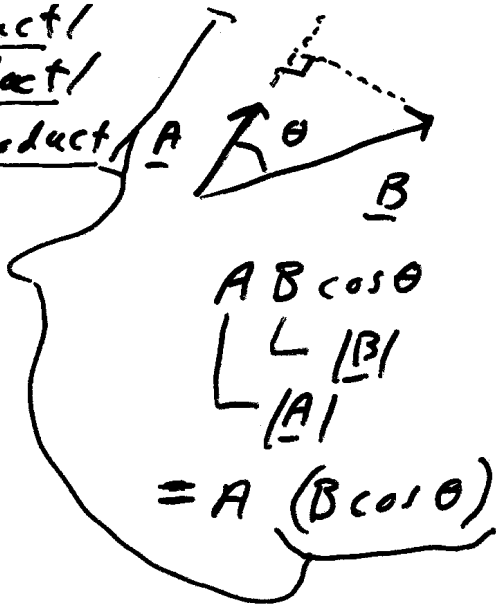
by slope, angle or dotted line between pts. w/ known locations

★ If you come early please sit towards middle of rows. (Less climbing at start of class) ④
TODAY: Vectors, Vectors, Vectors JAN 22, 2003

	<u>Picture</u>	<u>abstract/eqn.</u>	<u>Comp</u>
<u>Vector</u>		$\underline{F}, \hat{F}, \underline{F}$ \underline{F}	$F_x \hat{i} + F_y \hat{j}$ $+ F_z \hat{k}$
<u>Addition</u>		$\underline{A} + \underline{B} = \underline{C}$	$C_x = A_x + B_x$ $C_y = A_y + B_y$ $C_z = A_z + B_z$
<u>Multi by a scalar.</u>	 <p>a times longer than A</p>	$\underline{C} = a \underline{A}$	$C_x = a A_x$ $C_y = a A_y$ $C_z = a A_z$

"triangle" rule,
 "parallelogram" rule, "tip to tail" rule

dot product / inner product / scalar product



$$= A (B \cos \theta)$$

$$= A \cdot (\text{proj. of } B \text{ in } A \text{ dir.})$$

$$= B \cdot (\text{proj. of } A \text{ in } B \text{ dir.})$$

$$C = \underline{A} \cdot \underline{B}$$

↳ a scalar

$$C = A_x B_x + A_y B_y + A_z B_z$$

$$\underline{A} \cdot (\underline{B} + \underline{D}) = \underline{A} \cdot \underline{B} + \underline{A} \cdot \underline{D}$$

2D:

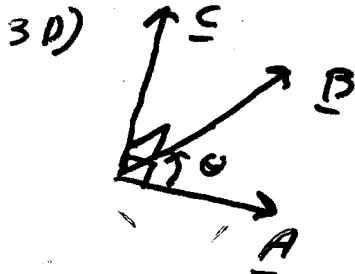
$$C = A_x B_x + A_y B_y$$

$$\begin{cases} \hat{i} \cdot \hat{i} = 1 \\ \text{etc} \\ \hat{i} \cdot \hat{j} = 0 \end{cases}$$

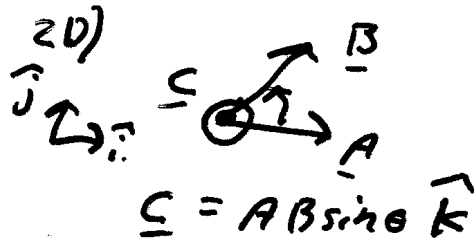
ex) $\underline{A} = 2\hat{i} + 4\hat{j}$
 $\underline{B} = \hat{i} - 3\hat{j}$

$$\underline{A} \cdot \underline{B} = 2 \cdot 1 + 4 \cdot (-3) = -10$$

cross product / vector product



$$|C| = AB \sin \theta$$



$$\underline{C} = \underline{A} \times \underline{B}$$

$$\hat{i} \times \hat{j} = \hat{k}$$

etc

$$\hat{i} \times \hat{i} = \underline{0}$$

$$\underline{A} \times (\underline{B} + \underline{D}) = \underline{A} \times \underline{B} + \underline{A} \times \underline{D}$$

$$\underline{C} = C_x \hat{i} + C_y \hat{j} + C_z \hat{k}$$

$$C_x = A_y B_z - A_z B_y$$

$$C_y = -A_x B_z + A_z B_x$$

$$C_z = A_x B_y - A_y B_x$$

$$\text{ex) } \underline{A} = 3\hat{i} + 4\hat{j} + 6\hat{k}$$

$$\underline{B} = \hat{i} - 2\hat{j} + 9\hat{k}$$

$$\underline{A} \times \underline{B} = (3\hat{i} + 4\hat{j} + 6\hat{k}) \times (\hat{i} - 2\hat{j} + 9\hat{k})$$

$$= (\quad) \hat{i}$$

$$+ (\quad) \hat{j}$$

$$+ (3 \cdot 2 - 4 \cdot 1) \hat{k}$$

RULES of Vector algebra

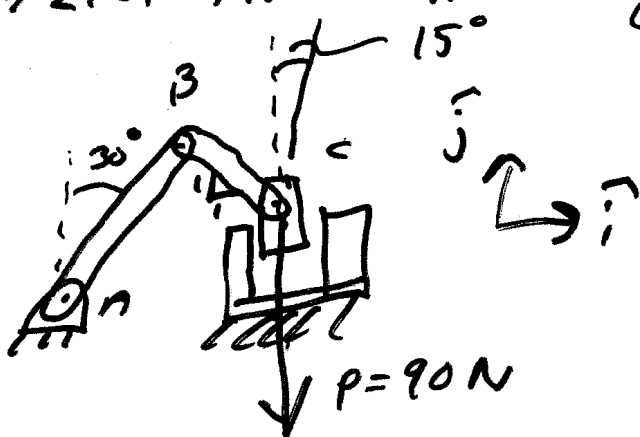
All distributive, associative, commutative rules that you know apply to

$\underline{A} + \underline{B}$, $a\underline{A}$, $\underline{B} \cdot \underline{A}$, $\underline{B} \times \underline{A}$ except

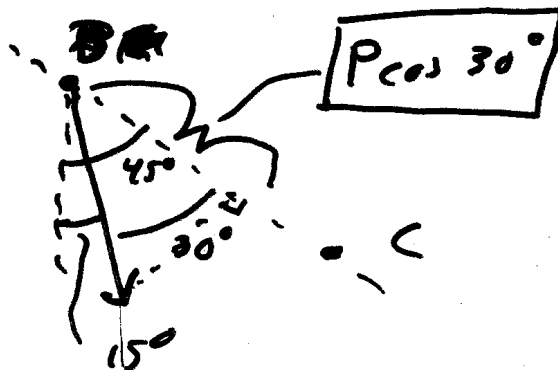
$$\underline{A} \times \underline{B} = -\underline{B} \times \underline{A}.$$

{WATCH OUT: $\frac{\underline{A}}{\underline{A}}$ is nonsense}

ex) 2.24 from book



Comp. of \underline{P} along AB (↑)



Alt. Approach:

$$\begin{aligned}
 \left. \begin{array}{l} \text{Comp. of } \underline{P} \text{ in} \\ \text{dir. } BC \end{array} \right\} &= \underline{P} \cdot \hat{j}_{BC} \\
 &= \cancel{P(\cos 15^\circ \hat{j} + \sin 15^\circ \hat{i})} \\
 &= P(\cos 15^\circ (-\hat{j}) + \sin 15^\circ \hat{i}) \\
 &\quad \cdot \left(\frac{\hat{i} - \hat{j}}{\sqrt{2}} \right) \\
 &= P \left(\frac{\sin 15^\circ}{\sqrt{2}} + \frac{\cos 15^\circ}{\sqrt{2}} \right) \\
 &\quad \underbrace{\hspace{10em}}_{\cos 30^\circ} \\
 &= P \cos 30^\circ
 \end{aligned}$$

$$\begin{aligned}
 \text{Trig aside! } \cos 30^\circ &= \cos(45^\circ - 15^\circ) \\
 &= \cos 45^\circ \cos 15^\circ + \sin 45^\circ \sin 15^\circ \\
 &= \frac{\cos 15^\circ}{\sqrt{2}} + \frac{\sin 15^\circ}{\sqrt{2}}
 \end{aligned}$$

That is: With vectors we effectively derived the trig identity for $\cos(\theta - \phi)$!

Moment of a Force (w.r.t. some pt.) (8)



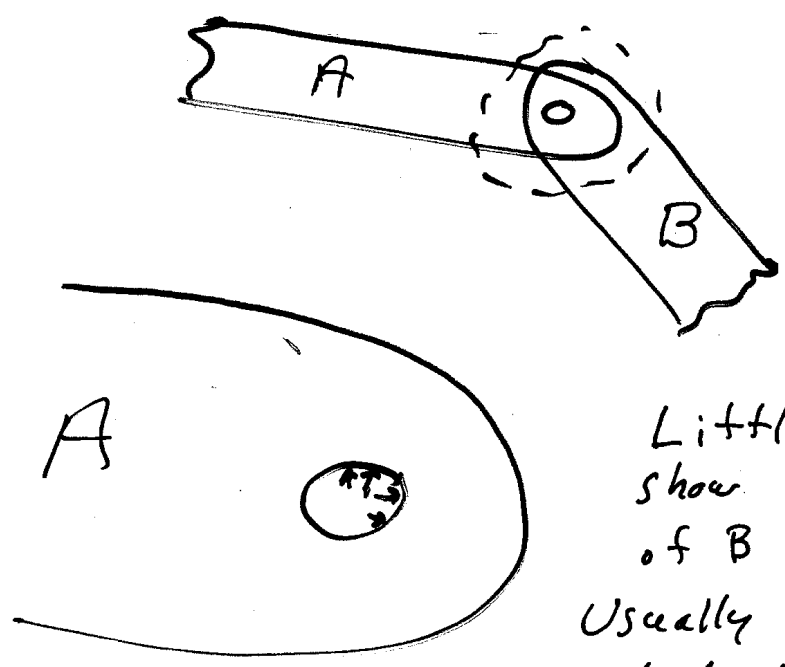
$$\underline{M}_{/C} = \underline{r}_{A/C} \times \underline{F}$$

$$\uparrow \text{Vector} = \underline{r}_{CA} \times \underline{F}$$

$$|\underline{M}_{/C}| = d F$$

Note: moment of a force is dependent on ref. pt.

ex)

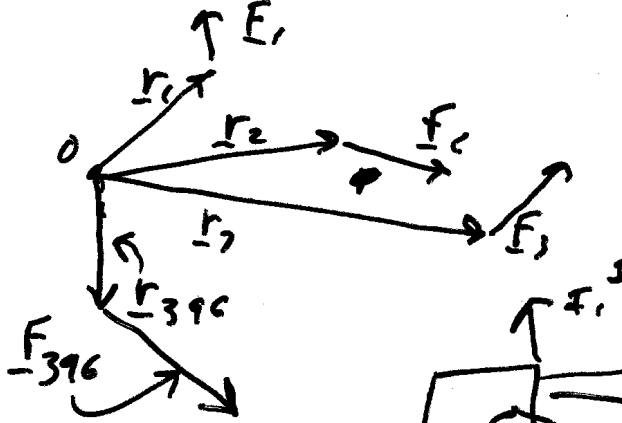


Little arrows show contact forces of B on A. Usually we don't want to know such detail but instead want the net effect.

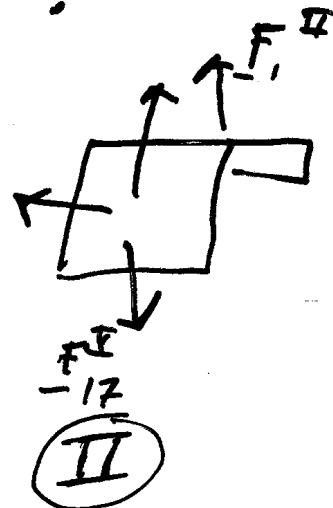
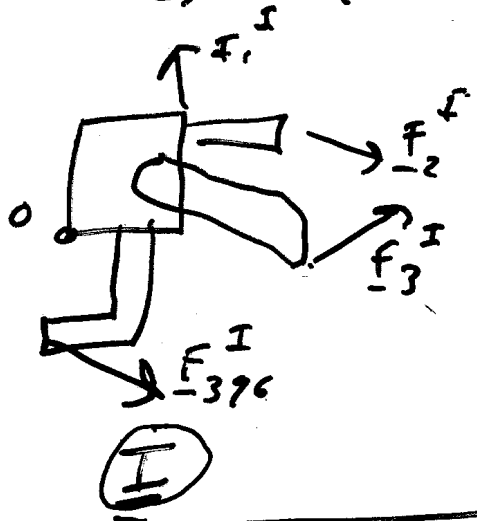
SO (cont'd next lecture)

TODAY : Equiv. force systems, couples
 (Please read Ruina & Pratap S2.6 online) 9
JANU.
2003

Force System : A collection of forces \underline{F}_i & locations $\underline{r}_i = \underline{r}_{i/o} = \underline{r}_{o/i}$



Question : When are 2 force systems I & II "equivalent".



I & II are equivalent iff $\sum \underline{F}_i^I = \sum \underline{F}_i^{II}$

C is any one point.

and $\sum \underline{r}_{i/c} \times \underline{F}_i^I = \sum \underline{r}_{i/c} \times \underline{F}_i^{II}$

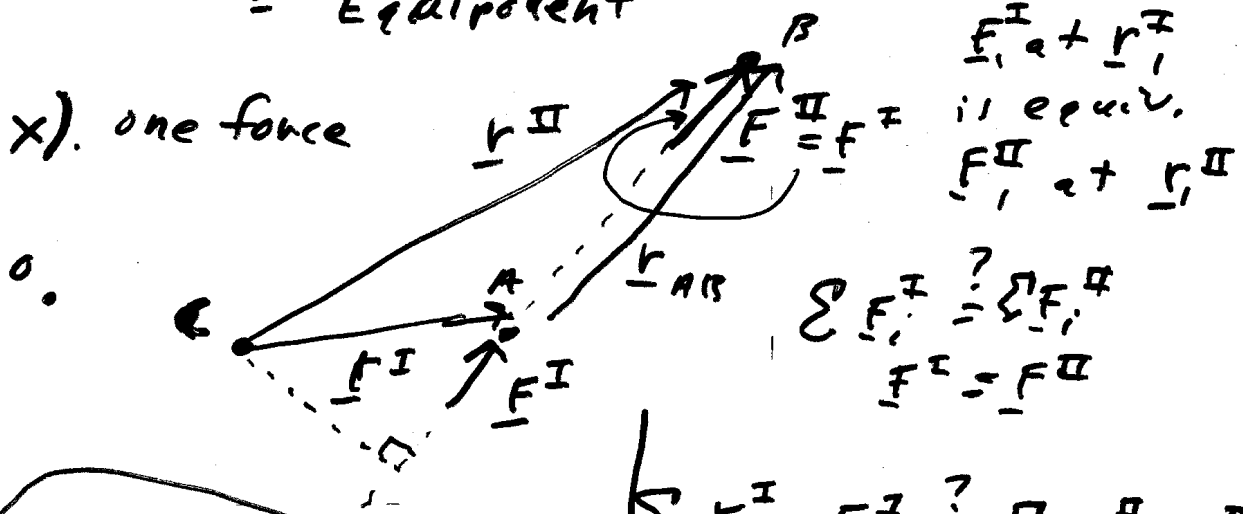
Forces & positions in system 1

Forces & positions in system 2.

Why use this def? Because all that the laws of mechanics know about a force system is the $\sum \underline{F}_i$ & $\sum \underline{r}_{i0} \times \underline{F}_i$.

"Equivalent" = "statically equiv." = "mechanically equiv." = "Equipollent"

ex) one force



$$\underline{F}_1^I + \underline{r}_1^I$$

is equiv.

$$\underline{F}_1^{II} + \underline{r}_1^{II}$$

$$\sum \underline{F}_i^I = \sum \underline{F}_i^{II}$$

$$\underline{F}^I = \underline{F}^{II}$$

$$\underline{r}^I \times \underline{F}^I = (\underline{r}_I + \underline{r}_{AB}) \times \underline{F}_I$$

$$= \underline{r}_I \times \underline{F}_I + \underline{r}_{AB} \times \underline{F}_I$$

$$= \underline{r}_I \times \underline{F}_I + \underline{0}$$

$$\underline{r}^I \times \underline{F}^I = \underline{r}_I \times \underline{F}_I$$

$$\underline{r}^I \times \underline{F}^I = \underline{r}^{II} \times \underline{F}^{II}$$

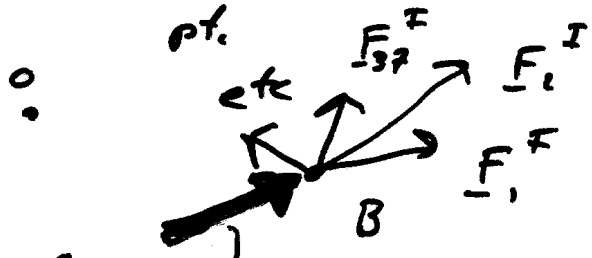
$$\underline{r}^I \times \underline{F}^I = \underline{r}^{II} \times \underline{F}^I$$

$$\underline{r}^{II} = \underline{r}_I + \underline{r}_{AB}$$

⇒ A force is equiv. to same force slid to a new location on its line of action.

Key Fact

ex) A collection of forces acts at one pt. (1)



is equiv to

A single force at that pt. = sum.

$$\underline{F}^R = \sum \underline{F}_i^I$$

$$\sum \underline{F}_i^F = ? \sum \underline{F}_i^R$$

$$\sum \underline{F}_i^I = \underline{F}^R$$

$$\sum \underline{M}_{i/c}^I = ? \sum \underline{M}_{i/c}^R$$

$$\sum \underline{r}_{i/c}^I \times \underline{F}_i^I = ? \sum \underline{r}_{i/c}^R \times \underline{F}_i^R$$

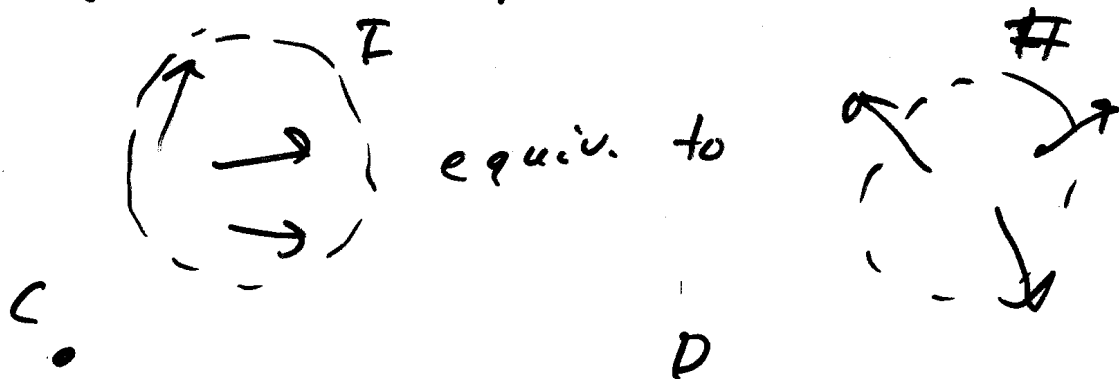
Key Fact

$$\underline{r}_{B/c} \times \sum \underline{F}_i^I = \underline{r}_{B/c} \times \underline{F}^R$$

subex)



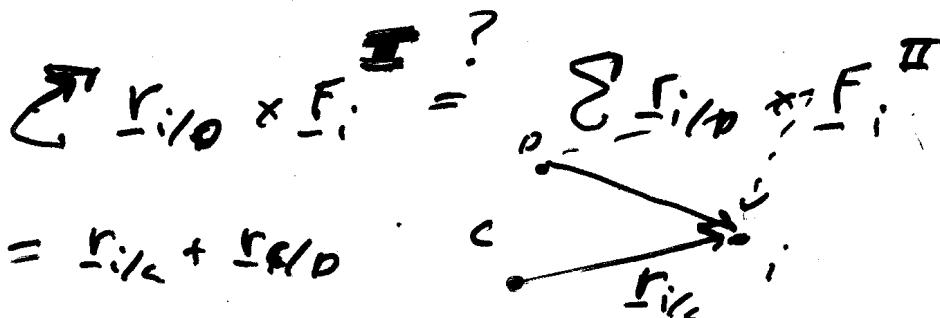
FACT: If Syst I is equiv. to Syst II (1D) for Ref. pt. C then it is necessarily equiv. for all pts. in the Universe.



if equiv. for C necessarily equiv. for D

⇒ You only need to check equivalence for one ref. pt. & then equiv. for all ref. pts is assured.

Check: $\sum F_i^I = \sum F_i^{II}$ ind. of ref. pt.



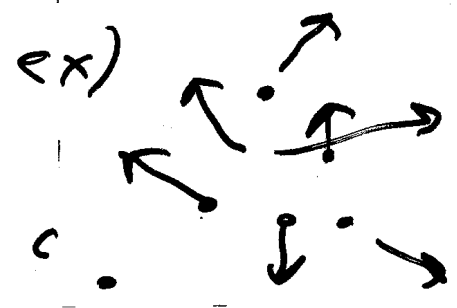
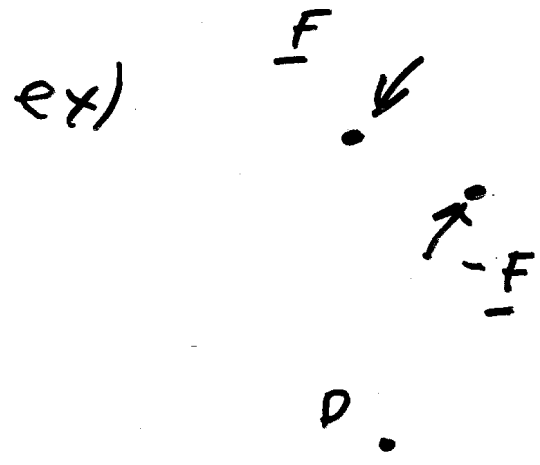
$r_{i/0} = r_{i/c} + r_{c/0}$

$$\sum r_{i/c} \times F_i^I + r_{c/0} \times \sum F_i^I = \sum r_{i/c} \times F_i^I + r_{c/0} \times \sum F_i^I$$

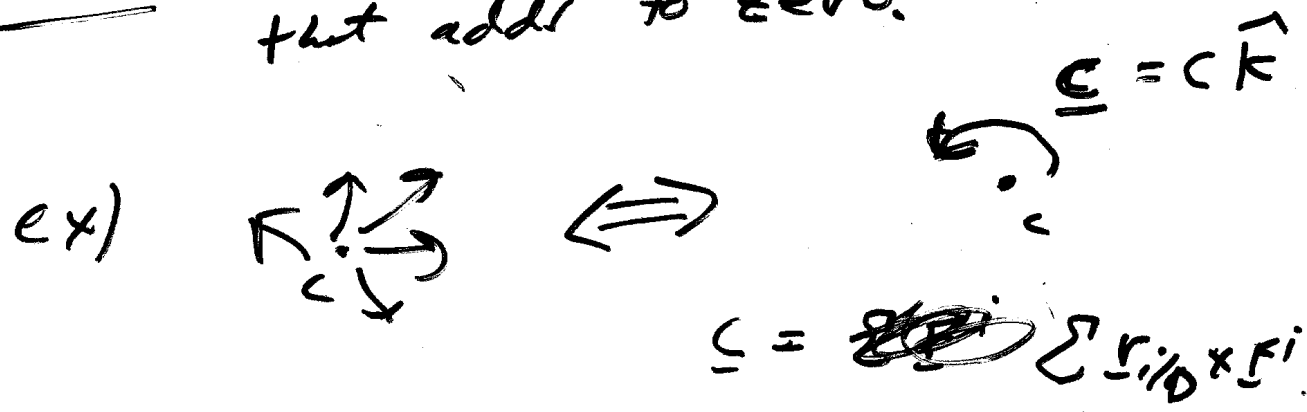
FACT: If a force system has $\sum \underline{F}_i = \underline{0}$ then

$$\sum \underline{M}_{/c} = \sum \underline{M}_{/o}$$

for all pairs of pts. C & D



DEF.: A couple is any force distr. that adds to zero.



FACT: Any force syst. is equiv to a \underline{F} and a \underline{C} at D

\perp force

\perp couple

TODAY: Couples etc. cont'd.

JAN 27, 2003

(14)

Recall: Any force system is equiv. to a force and a

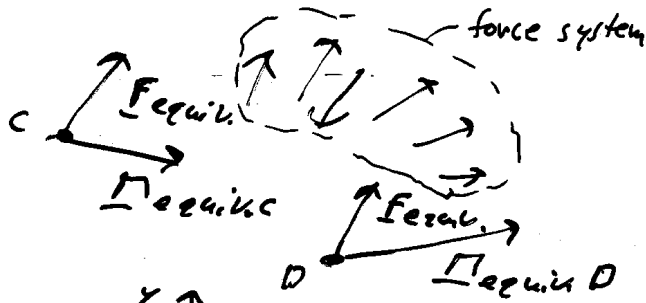
couple: force acts at C. $\underline{F}_{equiv} = \sum \underline{F}_i$

You pick the pt. C,

$$\underline{M}_{equiv} = \sum \underline{r}_{i/C} \times \underline{F}_i$$

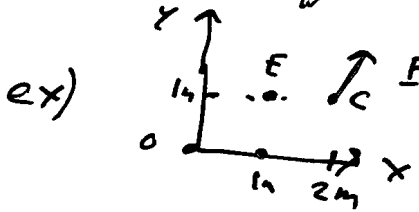
Change the point &

you change \underline{M}_{equiv} , but not \underline{F}_{equiv} .



$$2D: \underline{M}_{equiv} = M_{equiv} \hat{k}$$

$$\underline{M}_{equiv D} = \underline{M}_{equiv C} + \underline{r}_{C/D} \times \underline{F}_{equiv}$$



$\underline{F} = (3\hat{i} + 4\hat{j}) \text{ N}$ can be replaced by a force-couple syst. at origin that is "equivalent".

FACT: In 2D: Every force system is equiv. to either

(15)

a) just a force at an appropriate location
(\neq that whole line of action)
when $\sum \vec{F}_i \neq \vec{0}$

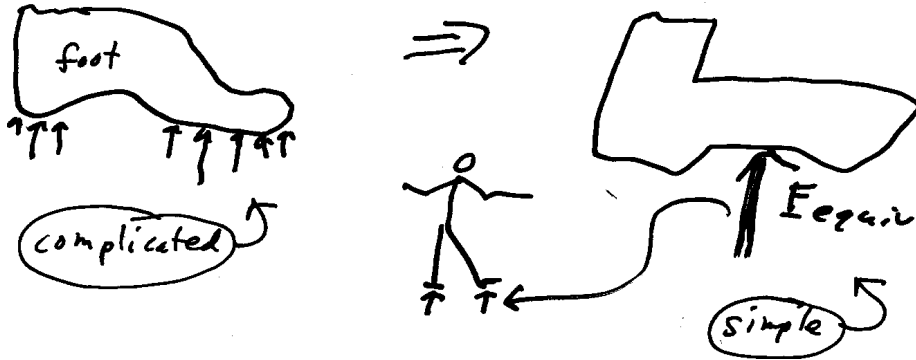
or b) just a couple applied any place
when $\sum \vec{F}_i = \vec{0}$

in 3D: Cannot generally reduce a force system to just a force,

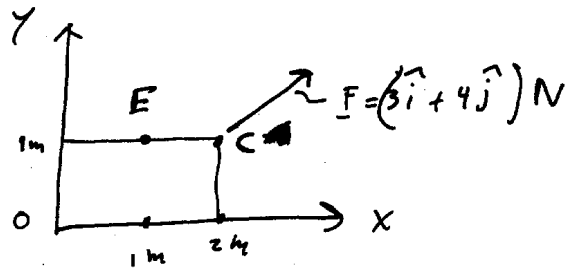
[can reduce to a "wrench"]

↑ see books

We like this because:



ex)



What is equiv. Force-Couple system at 0?

$$F_{equiv} = F = (3\hat{i} + 4\hat{j}) N$$

$$\begin{aligned} \underline{M}_{equiv} &= \underline{M}_O = \sum r_{iO} \times F_i = (2m\hat{i} + 1m\hat{j}) \times (3\hat{i} + 4\hat{j}) N \\ &= (r_{O} \parallel F) \sin \theta \quad (\text{too hard to use in this ex.}) \\ &= (2m \cdot 3N) \cdot 0 + (2m) \cdot (4N) \hat{k} + (1m \cdot 3N) (-\hat{k}) \\ &\quad + (1m) \cdot (4N) 0 \end{aligned}$$

$$\underline{M}_{equiv} = 5mN \hat{k}$$

What is $\underline{M}_{equiv} E$?

Use new force-couple system.

$$\begin{aligned} \underline{M}_{equiv E} &= \underline{M}_{equiv} + r_{O/E} \times F_{equiv} \\ &= 5mN \hat{k} + [(-\hat{i} - \hat{j})m] \times (3\hat{i} + 4\hat{j}) N \\ &= [5 - 4 + 3] mN \hat{k} \\ &= 4mN \hat{k} \end{aligned}$$

Given F_{equiv} & \underline{M}_{equiv} is there a place where $\underline{M}_{equiv} = 0$?

Try pt. C:

$$\underline{M}_{equiv E} = \underline{M}_{equiv}$$

$$\begin{aligned} &+ r_{E/C} \times F_{equiv} \\ &= 4mN \hat{k} + \\ &\quad -1m\hat{i} \times (3\hat{i} + 4\hat{j}) N \\ &= [4 - 4] mN \hat{k} \\ &= 0 \end{aligned}$$

{ as we knew had }
{ to work. }

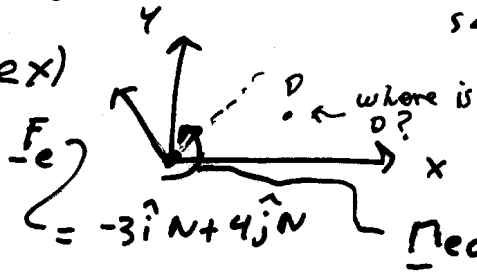
How to find locations where force system is equiv. to just a force (w/ no couple). Find pt. D

(17)



so that

ex)



$$\underline{0} = \underline{M}_{equiv.} = \underline{M}_{e0} + \underline{r}_{0D} \times \underline{F}_e$$

$$= 6 \text{ Nm } \hat{k} + \underline{r}_{0D} \times (-3 \text{ N } \hat{i} + 4 \text{ N } \hat{j})$$

(1 eqn. in 2 unknowns)

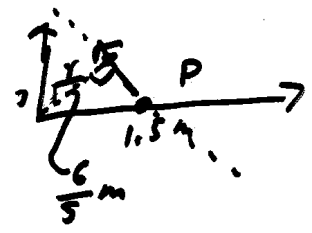
Try to find one point on line of solns.

$$\underline{r}_D = x_D \hat{i} + 0 \hat{j}$$

$$\underline{0} = \underline{M}_{equiv. D} = 6 \text{ Nm } \hat{k} + -x_D \hat{i} \times (-3 \text{ N } \hat{i} + 4 \text{ N } \hat{j})$$

$$\{ \underline{0} = [6 \text{ Nm} \quad -x_D \cdot 4 \text{ N}] \hat{k} \} \Rightarrow \{ 3 \cdot \hat{k} \} \Rightarrow x_D = \frac{6 \text{ Nm}}{4 \text{ N}} = 1.5 \text{ m}$$

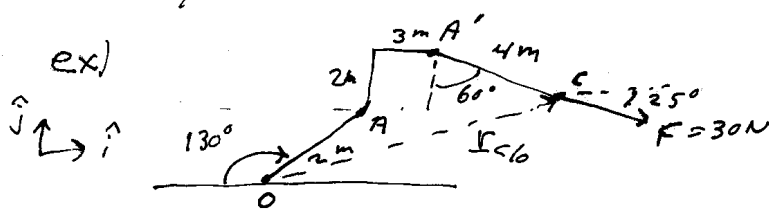
⇒



Ans. one good pt. D is on x axis 1.5 m from origin.

~~TODAY: 2 warnings 1) FBD & Statics~~ JAN 29, 2003
 TODAY: a) 2 warnings b) FBDs & Statics (18)

Warning 1: In life, engineering & HW problems geometry what you need is often not handed to you on a silver platter,
 ⇒ Don't let your geometry/trig issues get tangled with your mechanics issues.



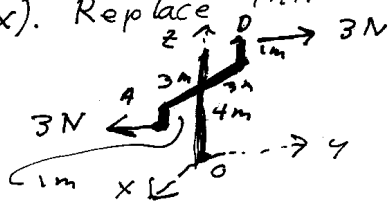
$$M_O = r_{C/O} \times F = r_{A/O} + r_{A'/A} + r_{C/A'}$$

$$r_{A/O} = 2m(\cos 50^\circ \hat{i} + \sin 50^\circ \hat{j}), \quad r_{A'/A} = (3m\hat{i} + 2m\hat{j})$$

$$r_{C/A'} = 4m(-\cos 60^\circ \hat{j} + \sin 60^\circ \hat{i}), \quad \hat{a} = \cos 25^\circ \hat{i} - \sin 25^\circ \hat{j}$$

Warning 2: Moment about a pt. in 3D is a 3D vector

ex). Replace this force system w/ a force & a couple
 $\sum F_i = 0 \Rightarrow$ equiv force is 0

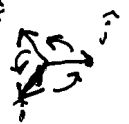


$$\sum M_O = \sum M/A = r_{D/A} \times 3N \hat{j}$$

↑ system is a couple

$$M = (6m\hat{i} + 2m\hat{k}) \times 3N\hat{j} = \boxed{-18Nm\hat{k} + 6Nm\hat{i}}$$

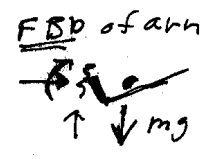
↑ The equiv. couple



FREE BODY DIAGRAMS (FBDs) read Chapter 4 in Rain/Fred

(17)

A picture of a collection of material w/ all external forces & couples showing.

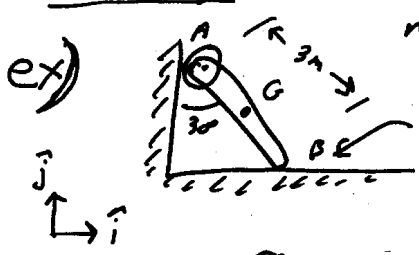


↳ "chainsaw on scalpet"

Laws of Mechanics: Apply to any FBD of any system

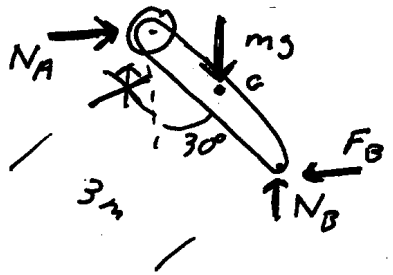
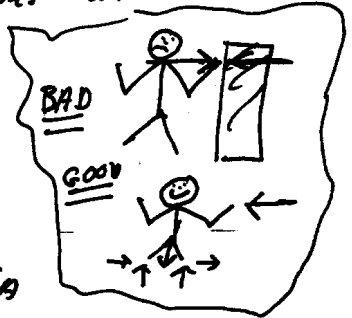
in static equilibrium: $\sum \underline{F}_i = 0$, $\sum \underline{M}_i = 0$
 For every pt. C
external external

Generally: 2D \Rightarrow 3 indep. scalar eqs., 3D \Rightarrow 6 ind. scalar eqs.



$m = 10\text{kg}$
 $\downarrow g = 10\text{N/kg}$
 assume no slip at B

Reactions at A & B



2D \Rightarrow 3 eqs.
 3 unknowns:
 N_B, F_B, N_A

$$0 = \sum \underline{M}_B = \int_{G/B} \times (mg \underline{j}) + \int_{A/B} \times N_A \underline{i}$$

$$\{ 0 = \frac{3}{4} m \cdot 100\text{N} \hat{k} + \frac{3\sqrt{3}}{2} m N_A (\hat{k}) \}$$

$$\{ \} \cdot \hat{k} \Rightarrow 0 = 75\text{Nm} - \frac{3\sqrt{3}}{2} N_A m$$

$$N_A = \frac{50}{\sqrt{3}} \text{N}$$

$N_A =$ force comp.
 $N = 1$ Newton.

(20)

(cont'd next lecture)

TODAY: FBDs & Statics (cont'd), 2) Q&A, 3) QUIZ 0 practice (13/10) & H.W.

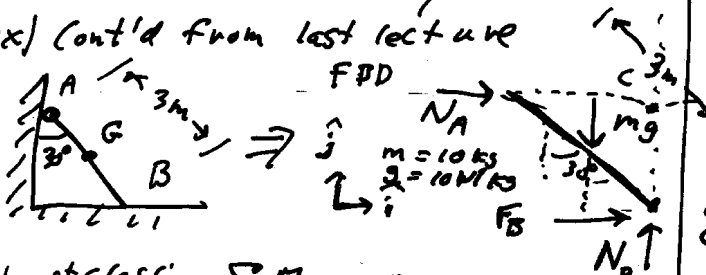
FBD: Pic. of system & all forces on it.

Mechanics: $\sum F_i = 0, \sum \tau_{ic} = 0$

sum over all forces/couples on FBD and no others.

Note: "Chainsaw", External forces only

ex) Cont'd from last lecture



Last class: $\sum \tau_{iB} = 0$

$\sum \tau_{iB} = 0 \Rightarrow N_A = \frac{2\sqrt{3}}{3} 100N$

Going on...

Method 1: $\sum F_i = 0$

$\sum F_i = 0 \Rightarrow$ 2 eqs for F_B & N_B

$\{ N_A \hat{i} + N_B \hat{j} + F_B \hat{i} - mg \hat{j} = 0 \}$ (2)

$\{ \sum \hat{i} \Rightarrow N_A + F_B = 0 \}$ 2 eqs for F_B & N_B
 $\{ \sum \hat{j} \Rightarrow N_B - mg = 0 \}$
 we know N_A already

Method 2: Can we find F_B w/out having found N_A first?

$\sum \tau_{iC} = 0$ ("kill" N_B & N_A)

$\Rightarrow \left\{ \left(\frac{3}{4} m \right) mg \hat{k} + \left(3m \frac{\sqrt{3}}{2} \right) F_B \hat{k} = 0 \right\}$

$\{ \sum \hat{k} \Rightarrow F_B = \frac{-3/4}{3\sqrt{3}/2} 100N$

Q&A

QUIZ 0

Feb 3, 2003 | TODAY: 3D Statics

Recall FBD = picture of isolated system and all external forces & couples that act on it.

Laws of Mechanics (statics)

For all forces/couples on a FBD

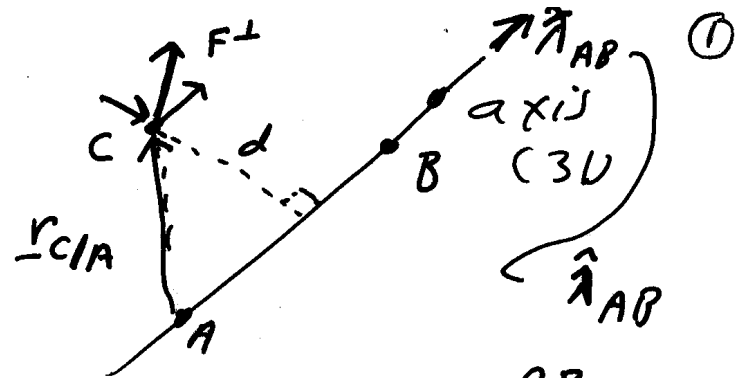
$$\sum \underline{F}_i = \underline{0} \quad \text{and} \quad \sum \underline{M}_C = \underline{0}$$

C any pt. you like

generally 6 ind. scalar eqs in 3D

Moment about an axis

tendency of forces (& couples) to cause rotation about an axis (given force syst. has different moment about diff. axes)



Moment about axis AB

$$= d F^\perp$$

↳ hard to evaluate from geometry (often)

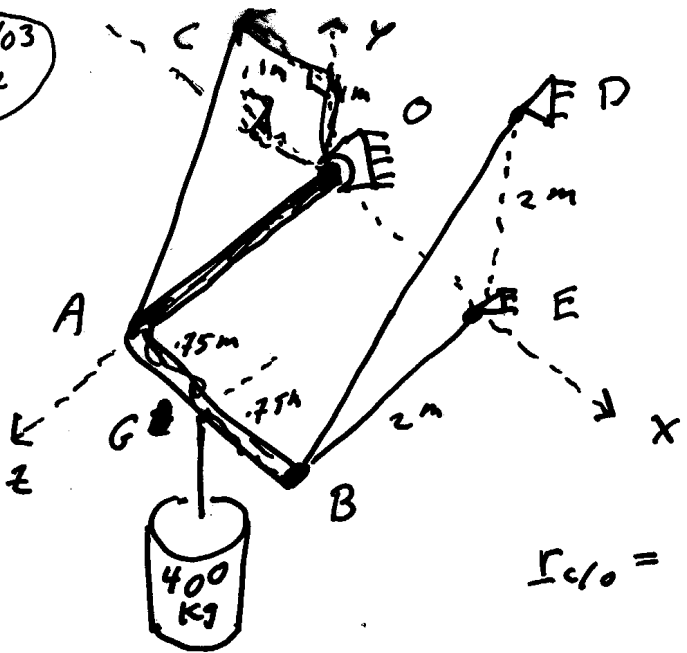
$$= (\underbrace{\underline{r}_{C/A}}_{\underline{r}_{C/A}} \times \underline{F}) \cdot \hat{n}_{AB}$$

$$= (\underline{r}_{C/B} \times \underline{F}) \cdot \hat{n}_{AB}$$

ex) 3.67 from Periam

~

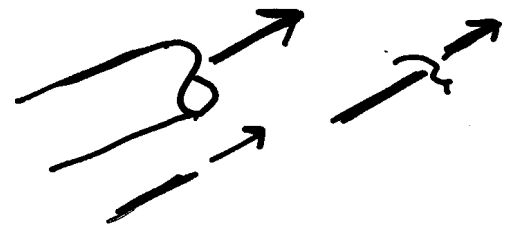
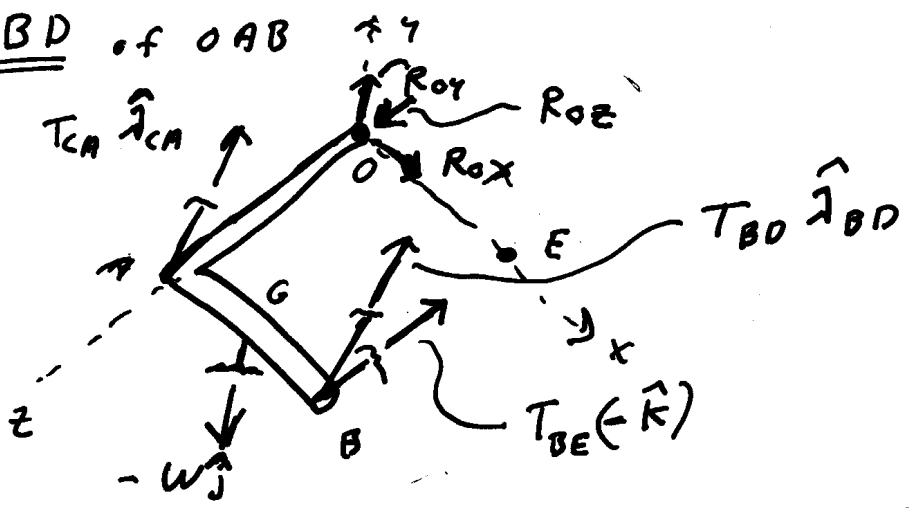
2/3/03
172



O, C, D, E
are co-planar.
Bent bar OABD.
Wires AC, BD,
rod BE.
Ball & socket
joints at
O, A, B, C, D, E,
G

$$r_{c/o} = -1m\hat{i} + 1m\hat{j}$$

FBD of OAB



Find unknown forces
& reactions: $T_{BE}, T_{BD},$
 $T_{CA}, R_{Ox}, R_{Oy}, R_{Oz}$ (6
unknowns)

$$\sum \underline{F}_i = \underline{0}, \quad \sum \underline{M}_i/H = \underline{0} \quad (1)$$

6 eqs.

Method 1: Pick any pt
if you like. Write eqs.
1. Break into comp.
solve 6 eqs for 6
unknowns.

Method 2: 2/3/03 p13

Try to get 1 eq. in 1 unknown.

$$\sum M_{axis AB} = 0$$

only R_{Oy} contributes

$$\Rightarrow \boxed{R_{Oy} = 0}$$

$$\sum M_{axis OB} = 0$$

only T_{CA} & W contribute

$$\left(\sum \underline{M}_{/B} \right) \cdot \frac{\underline{r}_{OB}}{|\underline{r}_{OB}|} = 0$$

$$\left(\underline{r}_{G/B} \times W(-\hat{j}) + \underline{r}_{A/B} \times T_{CA} \hat{\lambda}_{AC} \right) \cdot \hat{\lambda}_{OB} = 0$$

$$\underline{r}_{G/B} = -\frac{3}{4} m \hat{i}, \quad \underline{r}_{A/B} = -7.5 m \hat{i}$$

$$\hat{\lambda}_{AC} = \frac{\underline{r}_{AC}}{|\underline{r}_{AC}|} = \frac{-2\hat{k} - 1\hat{j}}{\sqrt{6}}$$

$$\hat{\lambda}_{OB} = (1.5m\hat{i} + 2m\hat{k}) / \sqrt{(1.5m)^2 + (2m)^2}$$

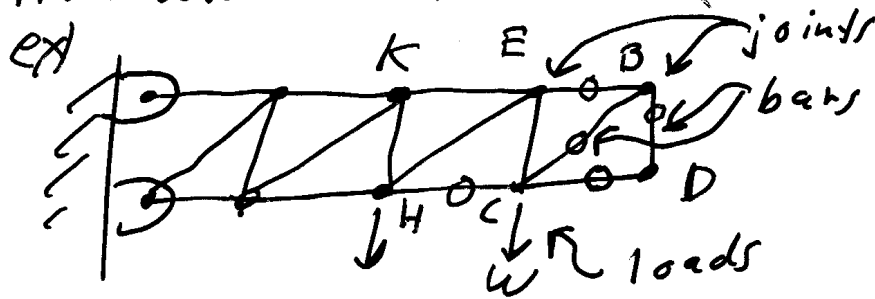
Plug in, crank $\Rightarrow T_{AC} = \dots$

$$\sum M_{axis OA} \Rightarrow T_{BD} = \dots$$

Feb. 5, 2007 ①
TRUSSES:

Truss: A collection of "bars" connected at ends by hinges (2D), or ball & socket joints (3D).

All loads must be at "joints".



TRUSSES are GOOD.

- a) useful
- b) agreeable

Primary Goal: Given geometry, loads find tensions in bars, & reactions.

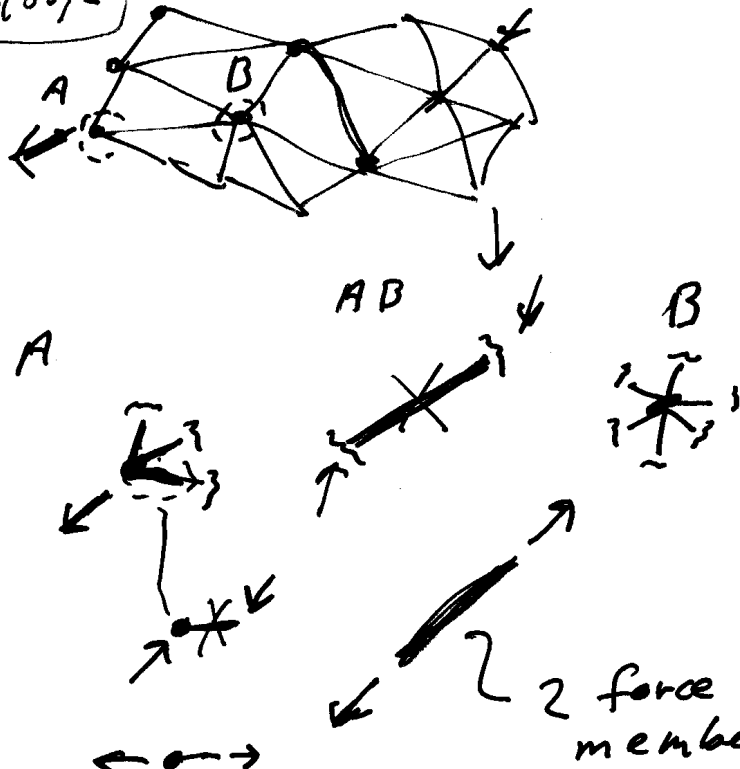
Method:

- a) joints
- b) part of truss

$$\sum \underline{F}_i = \underline{0}, \quad \left\{ \sum \underline{M}_i = \underline{0} \right\}$$

Key idea: bars are "force members".

2/5/03, 2

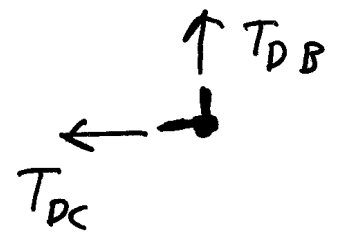


If only two forces act on a body in static equil.
 \Rightarrow equal in magnitude
 opposite in dir.
 point along the line connecting the 2 pts

of action.



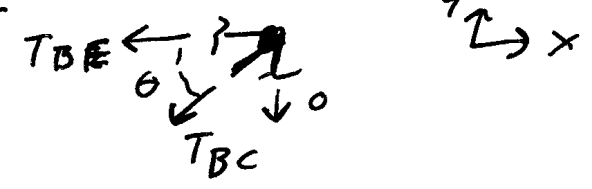
FBD of joint D from ex



$$\sum F_x = 0 \Rightarrow T_{DC} = 0$$

$$\sum F_y = 0 \Rightarrow T_{DB} = 0$$

Joint B



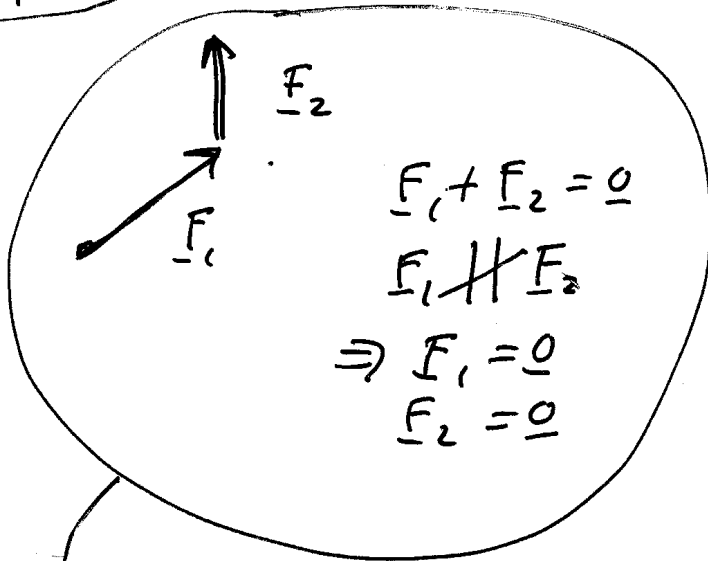
$$\sum F_{y'} = 0 \Rightarrow T_{BE} \sin \theta = 0$$

$$T_{BE} = 0$$

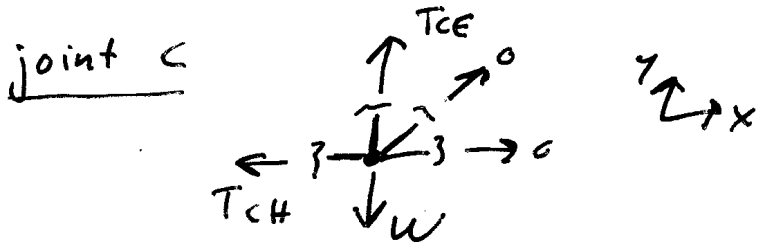
$$\sum F_x = 0 \Rightarrow T_{BC} \sin \theta = 0$$

$$T_{BC} = 0$$

2/5/03 p13



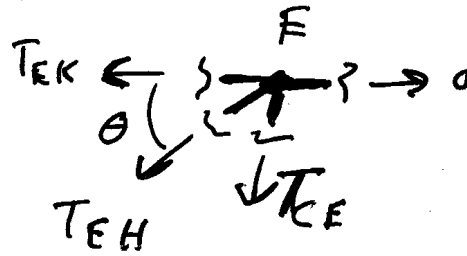
How to get zero-force members.



$$\sum F_x = 0 \Rightarrow T_{CH} = 0$$

$$\sum F_y = 0 \Rightarrow T_{CE} = W$$

joint E



$$\sum \underline{F}_i = \underline{0} \Rightarrow$$

$$\sum F_x = 0 \Rightarrow -T_{EK} - T_{EH} \cos \theta = 0$$

$$\sum F_y = 0 \Rightarrow -T_{CE} - T_{EH} \sin \theta = 0$$

* is 2 eqs. in 2 unknowns

~~$$\begin{aligned} T_{EK} + T_{EH} \cos \theta &= 0 \\ T_{CE} + T_{EH} \sin \theta &= 0 \end{aligned}$$~~

$$T_{EH} = -T_{CE} / \sin \theta$$

$$T_{EH} = -W / \sin \theta$$

"bar EH is in compression"

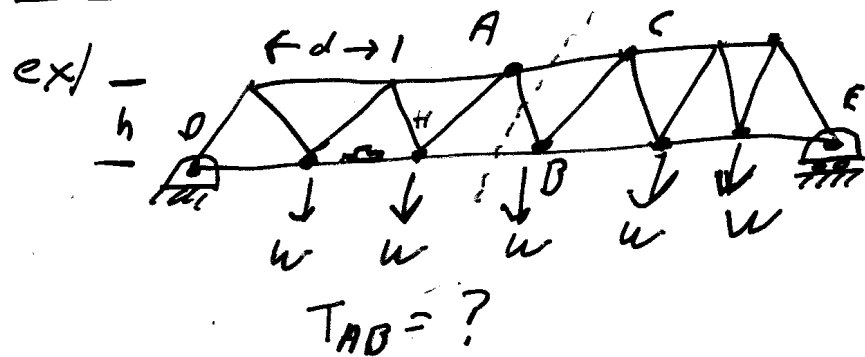
Feb. 10

TODAY: Trusses (cont'd)

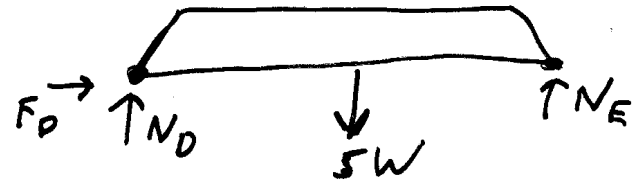
METHOD (most of time)

1. Draw FBD of whole structure
(use to find reactions if possible)
2. Find 0-force bars by inspection.
3. Draw FBDs of joints on sections as appropriate.

Method of "Sections"



FBD of whole structure \uparrow \rightarrow ?

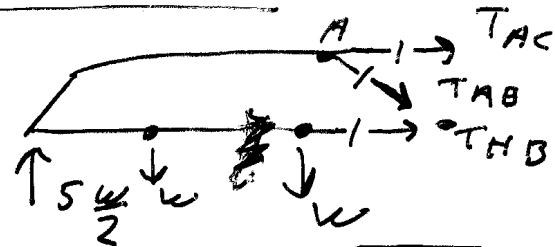


$$\{\sum F_x = 0\} \Rightarrow F_D = 0$$

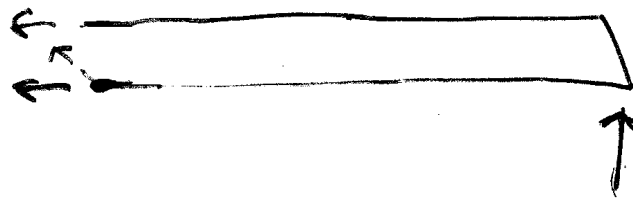
$$\sum \underline{M}_B = 0 \Rightarrow N_E = \frac{5w}{2}$$

$$\sum \underline{M}_C = 0 \Rightarrow N_D = \frac{5w}{2}$$

FBD of section



$$\sum F_y = 0 \Rightarrow T_{AB}$$



$$T_{AC} = ?$$

$$\sum \underline{M}_i = 0$$

$$\sum \tau_{iB} = 0 \quad \checkmark$$

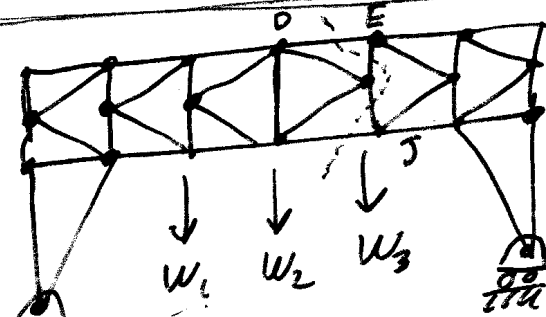
$$-(T_{AC} \cdot h) \hat{k} + (W \text{ terms}) \cdot (\text{horizontal distances}) = 0$$

$h \ll$ horiz. distances

$\Rightarrow |T_{AC}| \gg$ weight

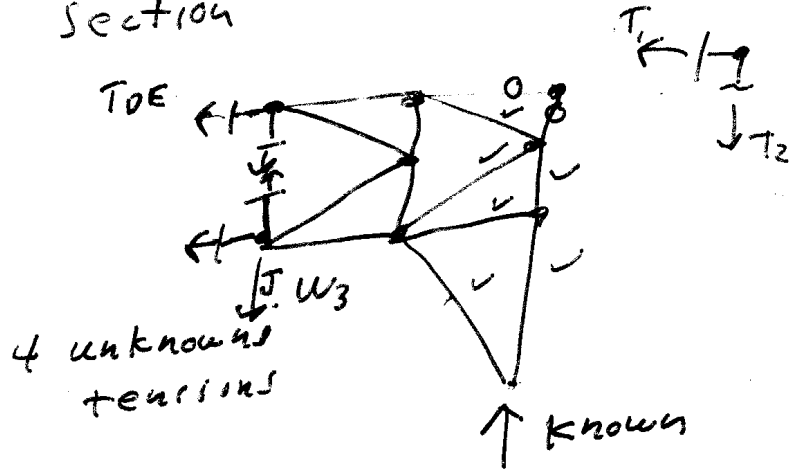
top in compression
bottom in tension

EX)
geometry given



$$T_{DE} = ?$$

FEB 10 p. 2
FBD of structure \Rightarrow reactions
section



$$\sum \tau_{ij} = 0 \Rightarrow T_{DE}$$

Sometimes

a) luck

b) no luck

\Rightarrow multiple section cuts.
~~set~~ set up sets of eqs.

$$\left(\sum \underline{r}_{i/c} \times \underline{F}_i \right) \cdot \hat{a}_{CD} = 0$$

Feb 12, 2003

TODAY: 3D TRUSSES

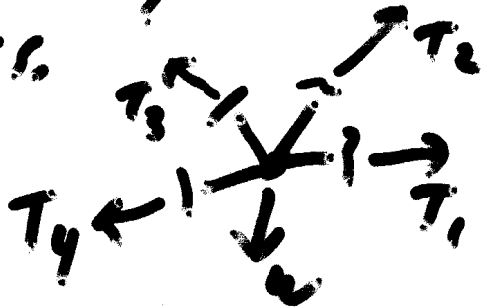
Methods (just like 2D)

* joints

Each FBD $\Rightarrow \sum \underline{F}_i = \underline{0}$

3 scalar eqs. \curvearrowright

one joint at a time
work way around
truss.



• c

• c'

* Sections

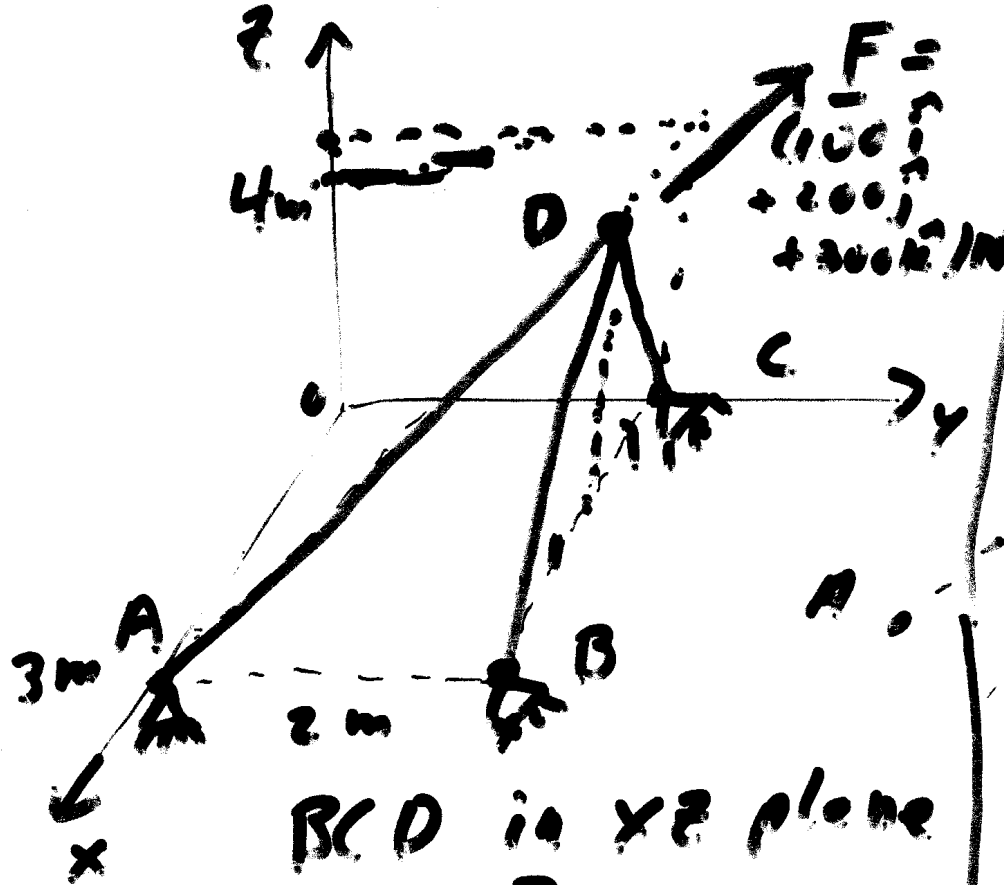
Each FBD \Rightarrow

$$\left\{ \begin{array}{l} \sum \underline{F}_i = \underline{0}, \sum \underline{M}_i = \underline{0} \\ \text{or} \\ \sum \underline{M}_i \text{ diff. axes} \\ \text{or} \end{array} \right.$$

6 ind. scalar
eqs.

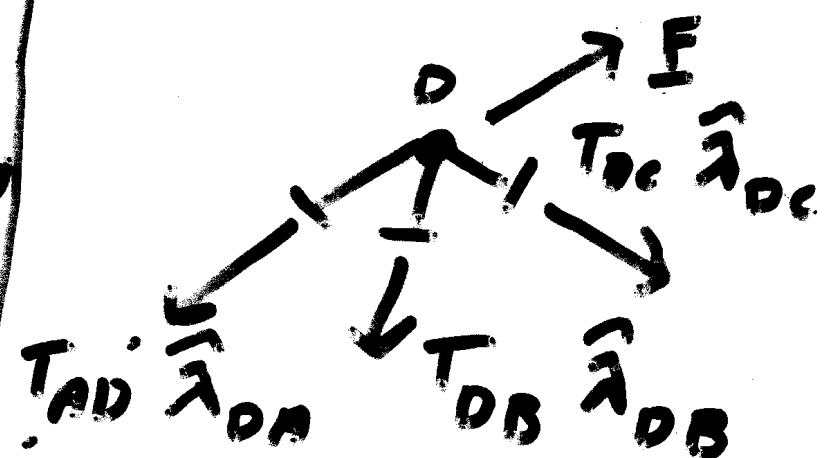
ex)

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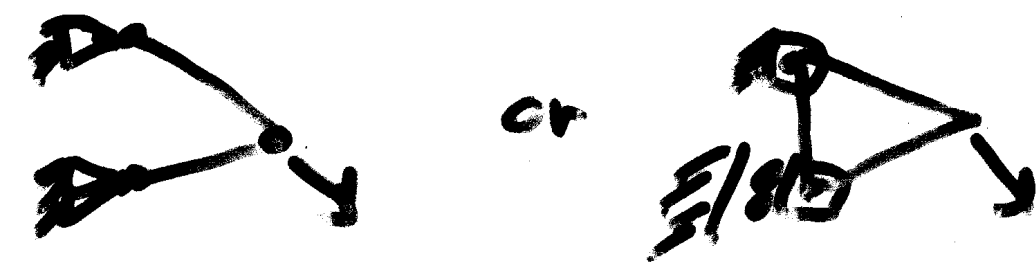
BCD is in xy plane
 $T_{DB} = ?$

FBD joint D



$$\sum \underline{F}_i = \underline{0}$$

$$T_{AD} \hat{\lambda}_{DA} + T_{DB} \hat{\lambda}_{DB} + T_{DC} \hat{\lambda}_{DC} + \underline{F} = \underline{0} \quad (1)$$



Side geometry: $\begin{matrix} 2 \\ 1 \\ 2 \\ 0 \\ 3 \end{matrix}$

$$\hat{\lambda}_{DA} = \frac{\mathbf{r}_{DA}}{|\mathbf{r}_{DA}|} = \frac{1.5\hat{i} - 2\hat{j} - 4\hat{k}}{\sqrt{1.5^2 + 2^2 + 4^2}}$$

$$= \lambda_{DAx} \hat{i} + \lambda_{DAy} \hat{j} + \lambda_{DAz} \hat{k}$$

$$\hat{\lambda}_{DC} = \dots$$

$$\hat{\lambda}_{DB} = \dots$$

$\{0\} \cdot \hat{i}, \{0\} \cdot \hat{j}, \{0\} \cdot \hat{k}$

3 eqs. in
3 unknowns.

How to solve?

Write in matrix
form

$$\lambda_{DAx} T_{DA} + \lambda_{DBx} T_{DB} + \lambda_{DCx} T_{DC} = -F_x$$

$$T_{CD} = -F_x$$

$$\lambda_{DAy} T_{DA} + \dots$$

$$\lambda_{DAz} T_{DA} + \dots$$

⇒

2/12 P. 4

$$\begin{bmatrix} \lambda_{DAx} & \lambda_{DBx} & \lambda_{DCx} \\ \lambda_{DAy} & \lambda_{DBy} & \lambda_{DCy} \\ \lambda_{DAz} & \lambda_{DBz} & \lambda_{DCz} \end{bmatrix} \begin{bmatrix} T_{DA} \\ T_{DB} \\ T_{DC} \end{bmatrix} = \begin{bmatrix} -F_x \\ -F_y \\ -F_z \end{bmatrix}$$

ready for computer soln.
calculator

⇒ $T_{BD} = \dots$

MATLAB soln.

2/12 pg 5

$$rDA = [1.5 \quad -2 \quad -4]';$$

$$rDB = [1.5 \quad 0 \quad -4]';$$

$$rDC = [-1.5 \quad 0 \quad -4]';$$

$$\text{lamDA} = rDA / \text{norm}(rDA);$$

$$\text{lamDB} = rDB / \text{norm}(rDB);$$

$$\text{lamDC} = rDC / \text{norm}(rDC);$$

$$F = [100 \quad 200 \quad 300]';$$

$$A = [\text{lamDA} \quad \text{lamDB} \quad \text{lamDC}];$$

Tensions = $A \setminus (F)$
 ↑
 backslash

done

Short cut

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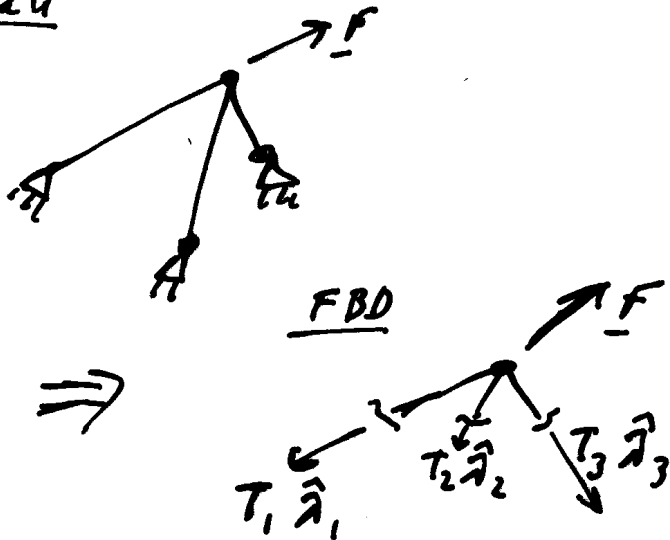
$$\{0\} \cdot (r_{on} \times r_{oc})$$

\Rightarrow 1 eq. for TDB

Feb. 17, 2003

TODAY: *3D trusses (cont, review)
* Machines & Structures
(First of a few lectures)

Recall



$$\sum \underline{F}_i = \underline{0} \Rightarrow \{ T_1 \hat{a}_1 + T_2 \hat{a}_2 + T_3 \hat{a}_3 + \underline{F} = \underline{0} \}$$

In matrix form

$$\left[\begin{array}{c|c|c} [\hat{a}_1] & [\hat{a}_2] & [\hat{a}_3] \end{array} \right] \begin{bmatrix} T_1 \\ T_2 \\ T_3 \end{bmatrix} = [\underline{F}]$$

↳ cols. are the comps of unit vectors

Solve by brute force (computer)
or...

$$\{ \underline{0} \} \cdot \underbrace{\hat{a}_1 \times \hat{a}_3}_{\substack{\text{is } \perp \text{ to } \hat{a}_1 \\ \text{" } \perp \text{ to } \hat{a}_3}}$$

$$T_2 \hat{a}_2 \cdot (\hat{a}_1 \times \hat{a}_3) = \underline{F} \cdot (\hat{a}_1 \times \hat{a}_3)$$

$$T_2 = \frac{\underline{F} \cdot \hat{a}_1 \times \hat{a}_3}{\hat{a}_2 \cdot \hat{a}_1 \times \hat{a}_3}$$

Math Aside

2/17 pg 2

$\underline{A} \cdot \underline{B} \times \underline{C}$ is called
mixed triple product
(used also for moment about
an axis: $\underline{M}_A = \underline{a} \cdot \underline{r} \times \underline{F}$)

Some facts:

$$\underline{A} \cdot \underline{B} \times \underline{C} = \underline{A} \times \underline{B} \cdot \underline{C}$$

Why? a) multi. out & check
b) Parallelepiped volume

Also: $\underline{A} \cdot \underline{B} \times \underline{C} = \det \begin{bmatrix} [\underline{A}]' \\ [\underline{B}]' \\ [\underline{C}]' \end{bmatrix}$

$$= \det \left[[\underline{A}] / [\underline{B}] / [\underline{C}] \right]$$

$$\underline{F} \cdot \hat{\underline{a}}_1 \times \hat{\underline{a}}_3 = \underline{F} \times \hat{\underline{a}}_1 \cdot \hat{\underline{a}}_3$$
$$= - \hat{\underline{a}}_1 \times \underline{F} \cdot \hat{\underline{a}}_3$$

$$\hat{\underline{a}}_2 \cdot \hat{\underline{a}}_1 \times \hat{\underline{a}}_3 = - \hat{\underline{a}}_1 \times \hat{\underline{a}}_2 \cdot \hat{\underline{a}}_3$$

$$\Rightarrow T_2 = \frac{\det \left[[\hat{\underline{a}}_1] / [\underline{F}] / \hat{\underline{a}}_3 \right]}{\det \left[[\hat{\underline{a}}_1] / [\hat{\underline{a}}_2] / [\hat{\underline{a}}_3] \right]}$$

Kramer's rule
(we've derived using
vectors)

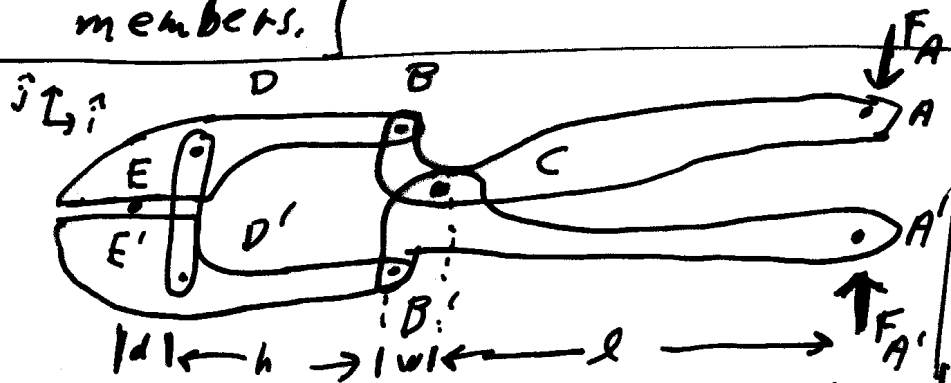
Machines:

2/17 P93

(& Structures)

Collections of objects connected in simple ways.

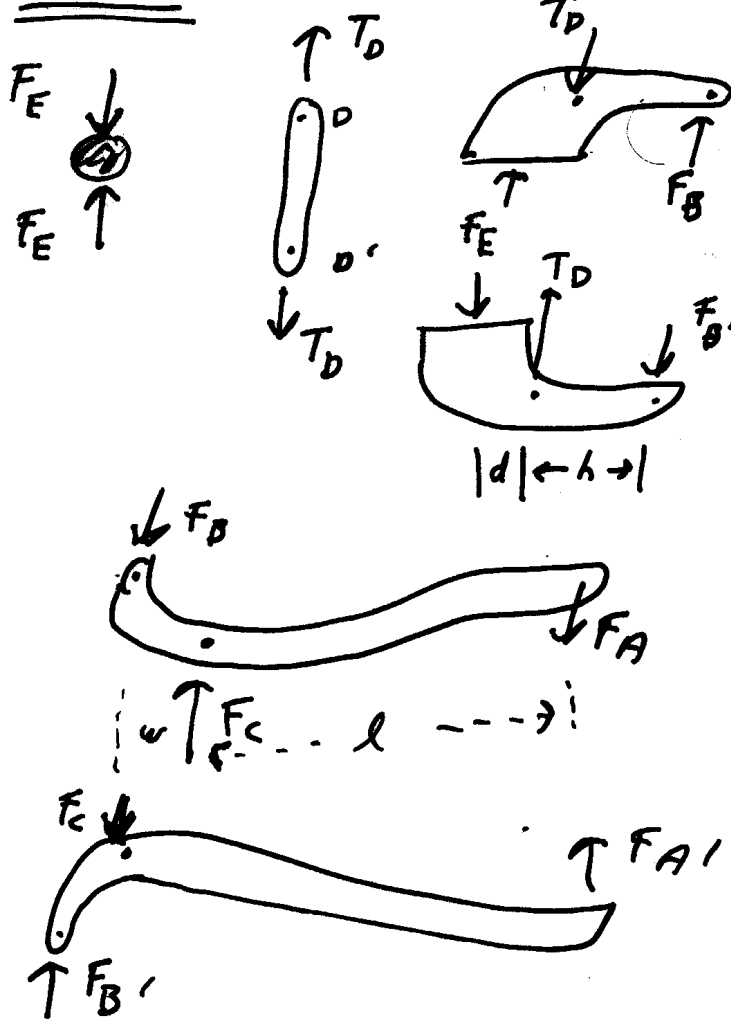
[Different from trusses in that some of objects are not 2-face members.]



Assume top-bottom symmetric above i FBD of bolt-cutter & bolt.

Neglect Weight

FBDs



$$\sum \underline{M}_C = \underline{0} \quad (\text{upper handle})$$

$$\Rightarrow -l F_A + F_B w = 0$$

$$\boxed{F_B = \frac{l}{w} F_A} \quad (1)$$

$$\sum \underline{M}_D = \underline{0} \quad (\text{cutter piece})$$

$$-h F_B + d F_E = 0$$

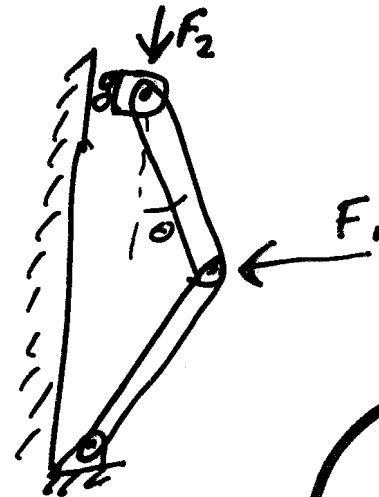
$$\boxed{F_E = \frac{h}{d} F_B} \quad (2)$$

$$(1) \& (2) \Rightarrow \boxed{F_E = \frac{h l}{w d} F_A}$$

$$\text{ex) } \frac{l}{w} \approx 40, \quad \frac{h}{d} \approx 4$$

$$\Rightarrow \boxed{F_E \approx 160 F_A !}$$

Note: "toggle mechanism"



θ is small

$$\Rightarrow F_2 \gg F_1$$

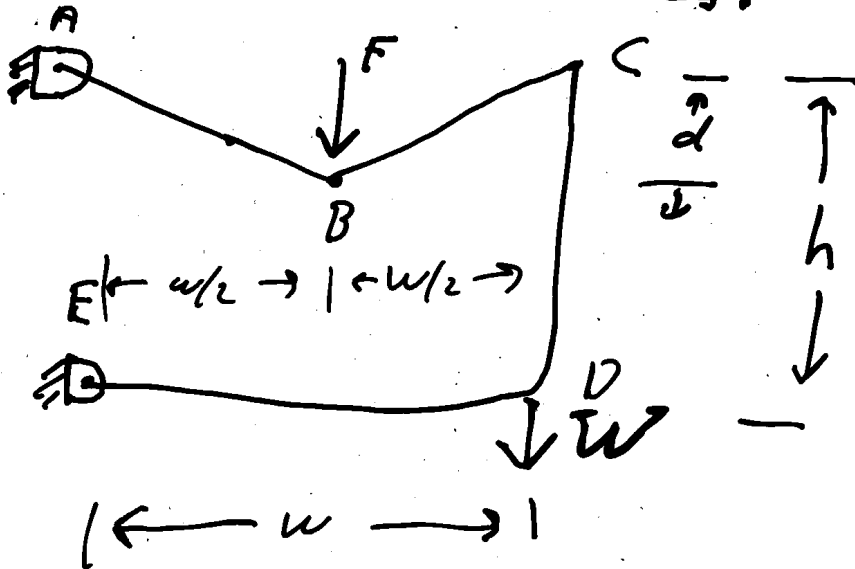
$$\boxed{\frac{F_1}{F_2} = \tan \theta}$$

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Feb. 19, 2003

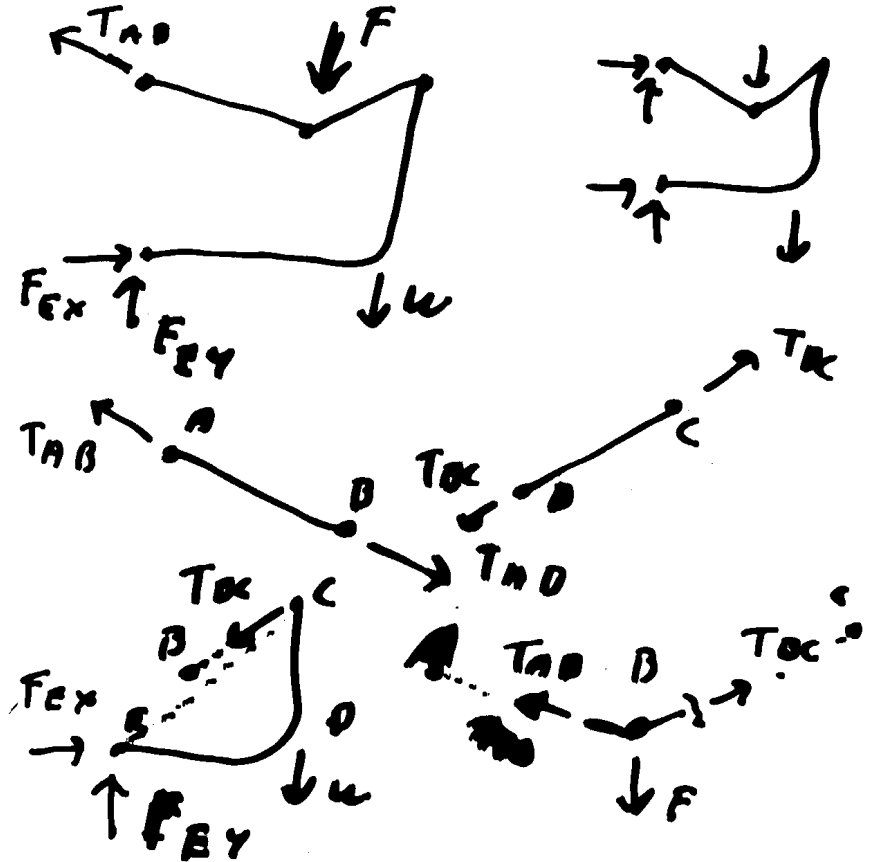
TODAY: Struct & Practice (cont'd)

ex) (from final exam a couple years ago.)



Given w, d, h, W find F, T_{BC} .

FBDs



Body EDC (2/11 19.2)

$$\sum \underline{M}_{/E} = \underline{0}$$

$$-wW \hat{k} + \underline{r}_{C/E} \times (T_{OC} \hat{\lambda}_{CB}) = \underline{0} \quad (1)$$

$$\underline{r}_{C/E} = \underline{r}_{EC} = w\hat{i} + h\hat{j}$$

$$\hat{\lambda}_{CB} = \frac{-w/2 \hat{i} - d\hat{j}}{\sqrt{w^2/4 + d^2}}$$

$$\begin{aligned} \underline{r}_{C/E} \times \hat{\lambda}_{CB} &= (w\hat{i} + h\hat{j}) \times \frac{(-w/2 \hat{i} - d\hat{j})}{\sqrt{w^2/4 + d^2}} \\ &= \frac{(wd + \frac{hw}{2}) \hat{k}}{\sqrt{w^2/4 + d^2}} \end{aligned}$$

(1) \Rightarrow

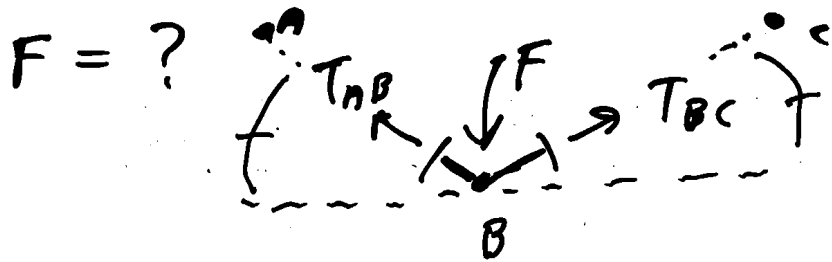
$$\left\{ \begin{aligned} -wW \hat{k} + \frac{(wd + \frac{hw}{2})}{\sqrt{w^2/4 + d^2}} T_{OC} \hat{k} \\ = \underline{0} \end{aligned} \right\}$$

$$\left\{ \right\} \cdot \hat{k} \Rightarrow$$

$$T_{OC} = W \frac{w\sqrt{w^2/4 + d^2}}{(\frac{hw}{2} - wd)}$$

$$T_k = \frac{W \sqrt{w^2/4 + d^2}}{(\frac{h}{2} - d)}$$

$$\begin{aligned} T_{OC} &> 0 \quad \text{if } d < h/2 \\ T_{OC} &< 0 \quad \text{if } d > h/2 \\ T_{OC} &\rightarrow \infty \quad d \rightarrow h/2 \end{aligned}$$



2/11 (PS2)

$$\sum \underline{F} = \underline{0}$$

$$\left\{ T_{AB} \hat{\lambda}_{BA} + T_{BC} \hat{\lambda}_{BC} - F \hat{j} = \underline{0} \right\}$$

take x & y comps $(\{ \} \cdot \hat{i}, \{ \} \cdot \hat{j})$

Symmetry on $\sum F_x = 0$

$$\Rightarrow T_{AB} = T_{BC}$$

$$\sum F_y = 0 \Rightarrow T_{BC} = \dots$$

$$\left[\begin{array}{l} \sum \underline{M}_A = 0 \Rightarrow \underline{F} \text{ w/out } T_{AB} \\ \sum \underline{M}_{\text{any pt on lke } AB \text{ except } B} \text{ ''} \\ \{ \} \cdot \hat{\lambda}_{\perp \text{ to } AB} \text{ ''} \\ (\{ \} \times \hat{\lambda}_{AB}) \cdot \hat{k} \end{array} \right]$$

Feb 21, 2003

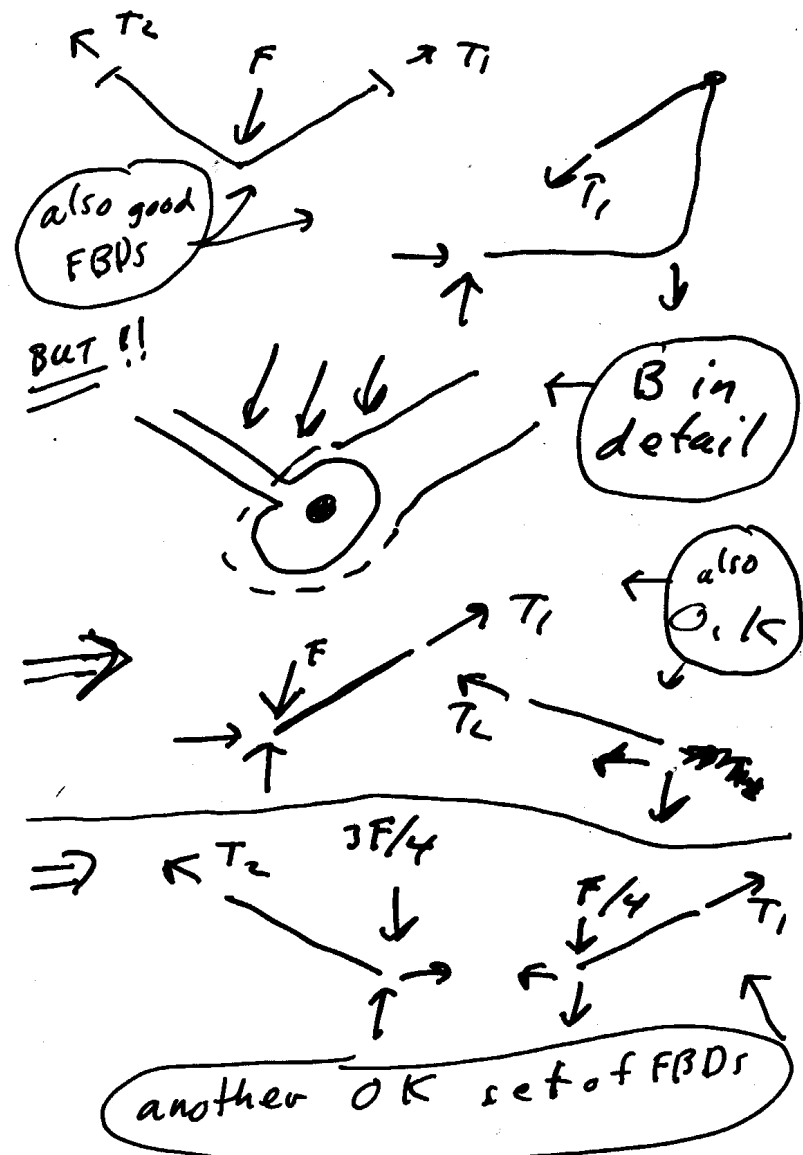
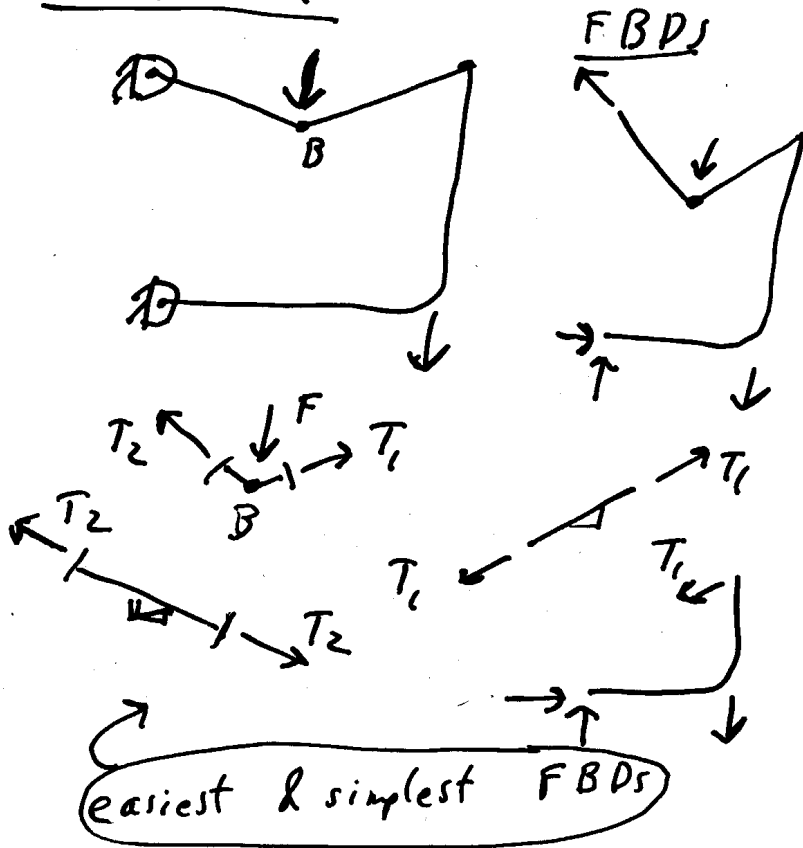
TODAY: Machines (cont'd)

a) last lecture example again.

b) force amplification

c) An old Prelim question

last lecture

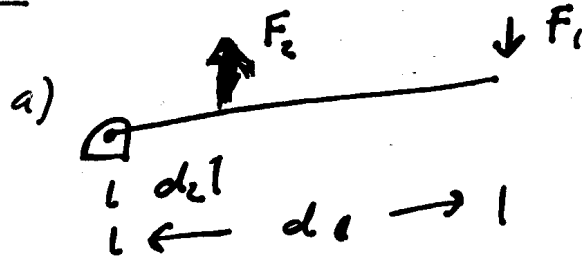


The ultimate values of T_1 & T_2 will not depend on whether F is applied to the pin at B , or bar 1, or both bars.

2/21 page 2

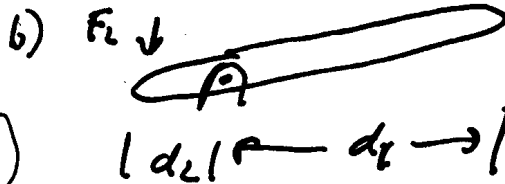
How to get big forces from small forces?

ex/ lever

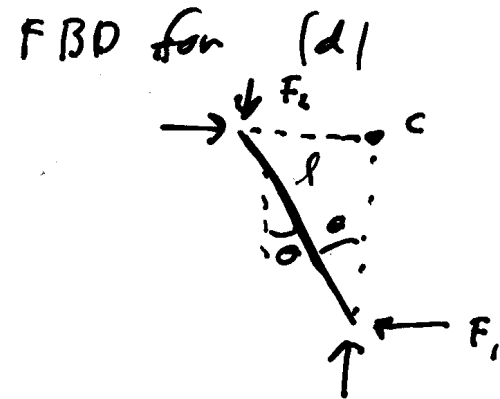
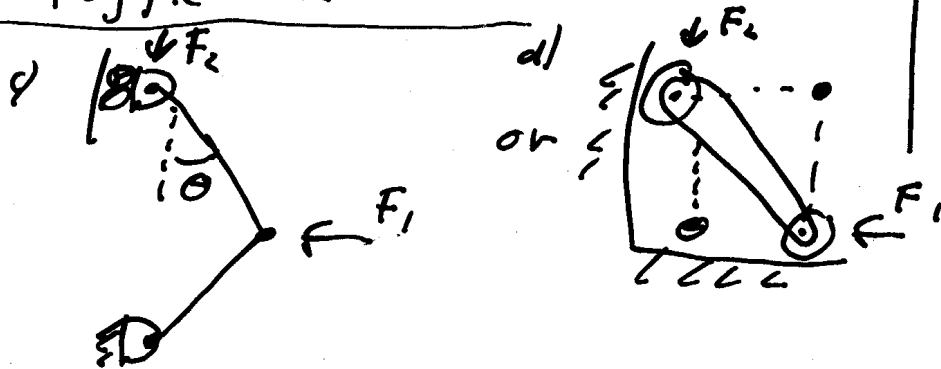


$$F_2 = F_1 \frac{d_1}{d_2}$$

$\frac{d_1}{d_2}$ big \Rightarrow
 F_2 big



ex/ Toggle mechanism



$$\sum \square_{lc} = 0$$

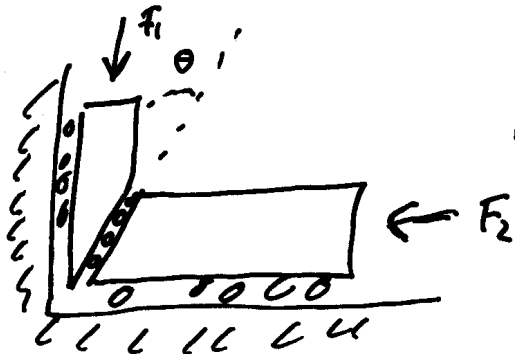
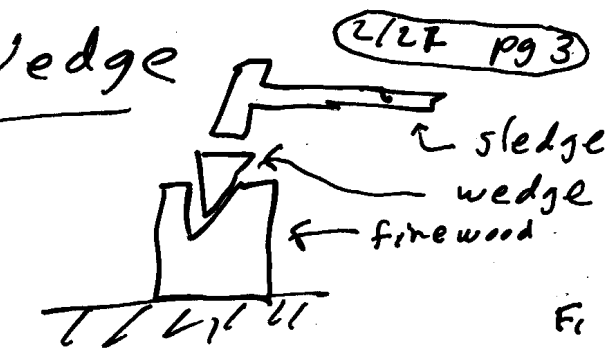
$$-F_1 l \cos \theta + F_2 l \sin \theta = 0$$

$$F_2 = F_1 / \tan \theta$$

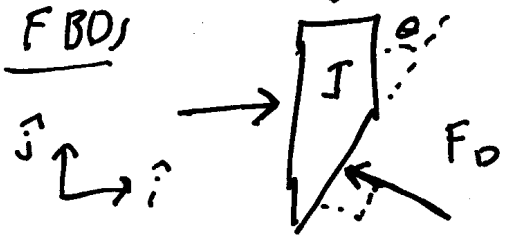
θ small \Rightarrow F_2 big

ex) Wedge

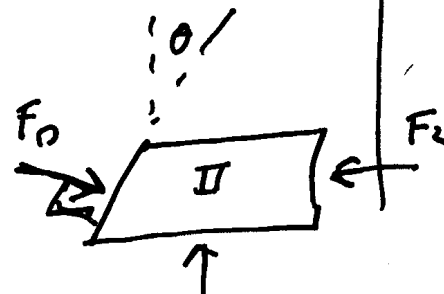
2/27 pg 3



FBD I



FBD I: $\sum F_x = 0$



$\Rightarrow F_1 = F_D \sin \theta$ (1)

FBD II: $\sum F_x = 0$

$F_2 = F_D \cos \theta$ (2)

$F_2 = F_1 / \tan \theta$
 small $\theta \Rightarrow$ big F_2

Morals: For frictionless passive machine

Energy balance

\Rightarrow Net work done on machine = 0

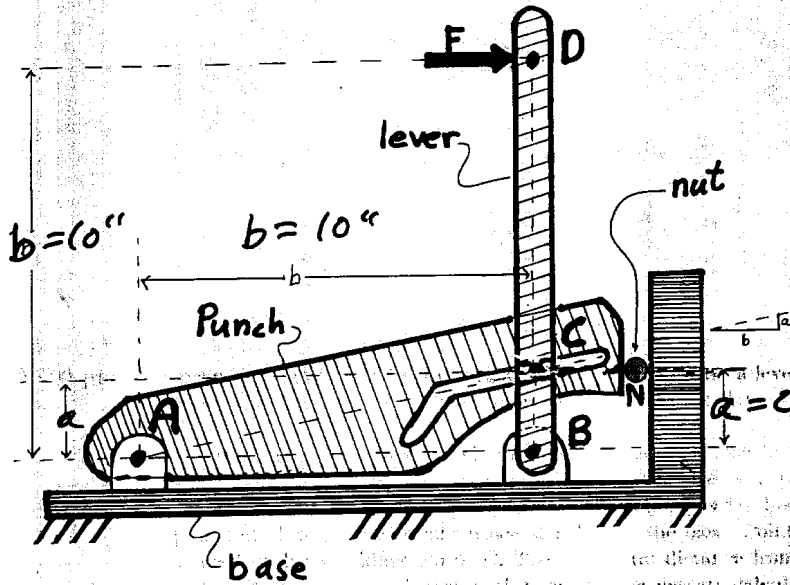
$|\delta_1 F_1| = |\delta_2 F_2|$
 δ_1 & δ_2 are displ. of F_1 & F_2

Force amplifiers are motion attenuators

3) (30 pts) The proposed nutcracker design consists of two moving parts: a lever hinged to the fixed base at B and a punch hinged to the fixed base at A. All joints and slots are assumed to have negligible friction.

Mechanism and geometry clarifications: The vertical lever has a pin at C and a horizontal force F applied at D. The punch has a slot in which the lever pin slides at C. The slot is parallel to the line AC. The spherical nut is cracked by being squeezed between the vertical surface of the punch at N and the vertical surface attached to the base. Point N at the left edge of the nut is level with the sliding pin at C. The horizontal distance from C to N does not enter the solution, but assume it is c if you need it for an intermediate calculation.

Quantities: $F = 10 \text{ lb}$, $a = 2 \text{ in}$, $b = 10 \text{ in}$.



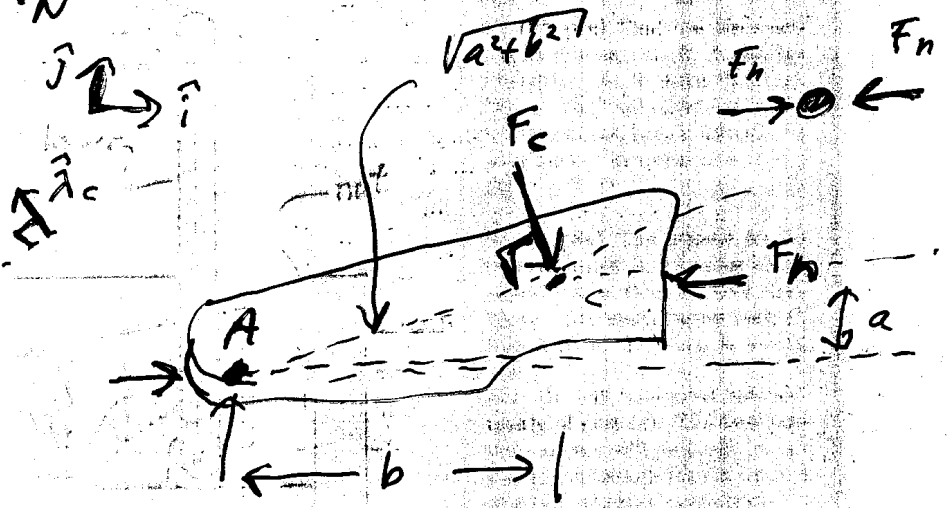
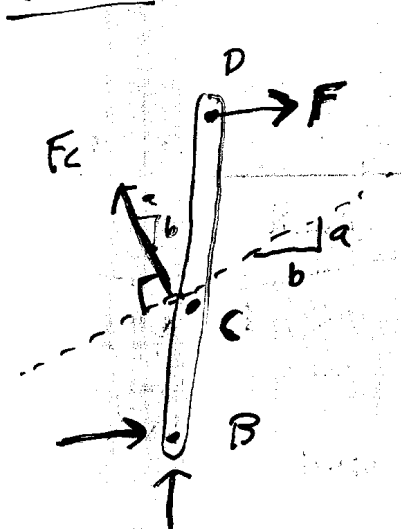
a) (25 pts) Find the force acting on the nut at N. A number is desired (i.e., so many lb force). [Hint: Only substitute in numbers when you have a formula for your answer in terms of a , b and F .]

b) (5 pts) The answer to (a) is conspicuous in its being either much smaller than F , very similar to F , or much bigger than F . Which is it? Explain, in words, why? Part (b) will be graded independently of part (a). The best possible answer will generate an approximate formula for the force at N using next-to-no equations.

$F = 10 \text{ lb}$

$F_N = \text{force on nut} = ?$

FBDs



(cont'd next lecture)

~~Feb 24~~ Feb 24 (pg 7) ex (cont'd)

Lever: $\sum \underline{M}_{/B} = \underline{0}$

$$-Fb\hat{k} + \underline{r}_{C/B} \times (F_c \hat{\lambda}_c) = \underline{0}$$

$$\underline{r}_{C/B} = a\hat{j}$$

$$\hat{\lambda}_c = \frac{-a\hat{i} + b\hat{j}}{\sqrt{a^2 + b^2}}$$

$$-Fb\hat{k} + (a\hat{j}) \times \left(F_c \left(\frac{-a\hat{i} + b\hat{j}}{\sqrt{a^2 + b^2}} \right) \right) = \underline{0}$$

$$\left\{ -Fb\hat{k} + \frac{a^2}{\sqrt{a^2 + b^2}} F_c \hat{k} = \underline{0} \right\} \quad (1)$$
$$\left\{ \right\} \cdot \hat{k} \Rightarrow \boxed{F_c = \frac{b\sqrt{a^2 + b^2}}{a^2} F}$$

Punch $\sum \underline{\Pi}_{/A} = \underline{0}$

$$\underline{r}_{C/A} \times (F_c (-\hat{\lambda}_c)) + \underline{r}_{N/A} \times F_n (-\hat{i}) = \underline{0}$$

$$\left\{ -\sqrt{a^2 + b^2} F_c \hat{k} + a F_n \hat{k} = \underline{0} \right\}$$
$$\left\{ \right\} \cdot \hat{k} \Rightarrow \boxed{F_n = \frac{\sqrt{a^2 + b^2}}{a} F_c} \quad (2)$$

$$(1) \& (2) \Rightarrow F_n = \frac{b(a^2 + b^2)}{a^2} F$$

$$b = 10'', \quad a = 2'', \quad F = 10 \text{ lb}$$

$$F_n = \frac{10(104)}{8} = 10 \text{ lb}$$

$$\boxed{F_n = 130 \cdot 10 \text{ lb} = 1300 \text{ lb}}$$

This lecture ~~problem~~ picks up w/ the example (nut cracker) from last lecture.

2/24/03 (Pg. 2)

CENTER OF MASS, CENTROID (center of gravity)

\underline{r}_G = average position of the mass
↑
position of C.O.M. w.r.t. 0

Analogy: test scores

$$T_A = \frac{T_1 \cdot n_1 + T_2 \cdot n_2 + T_3 \cdot n_3}{n_1 + n_2 + n_3}$$

↑ test score average

$$= \frac{65 \cdot 3 + 75 \cdot 5 + 95 \cdot 2}{10}$$

$$= \dots$$

average = { (value 1) (amount of stuff) / value 1
+ (value 2) (amount of stuff) / value 2
+ ...
} / (total amount of stuff)

$$\underline{r}_G = \frac{\sum \underline{r}_i m_i}{m_{tot}}$$

$$m_{tot} = \sum m_i$$

$$m_{tot} \underline{r}_G = \sum \underline{r}_i m_i$$

$$m_{tot} X_G = X_1 m_1 + X_2 m_2 + X_3 m_3 + \dots + X_n m_n$$

Why be interested in \underline{r}_G ? 2/29
173

1) The gravity forces on a system (for near-earth gravity: $-g\hat{j}$; constant g downwards) are "equivalent" to the total weight acting at \underline{r}_G .

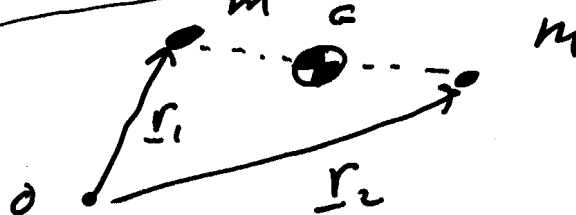


2) Centroid is important for understanding beam cross sections (as you will see in a month or so).

FACT

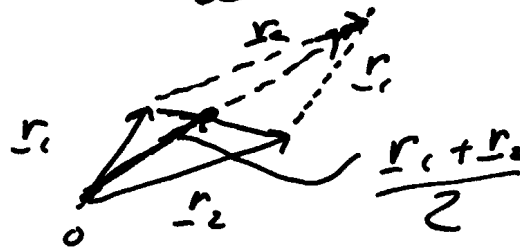
$$m_{tot} \underline{r}_G = \begin{cases} \sum \underline{r}_i m_i & \text{discrete} \\ \int \underline{r} dm & \text{continuous} \end{cases}$$

ex) Two equal pt. masses



$$2m \underline{r}_G = m \underline{r}_1 + m \underline{r}_2$$

$$\underline{r}_G = \frac{\underline{r}_1 + \underline{r}_2}{2}$$



$G = \text{Con}$ is at midpoint.

[Note: "objective". Location is independent of position or orientation of coord. syst. used for calculation.]

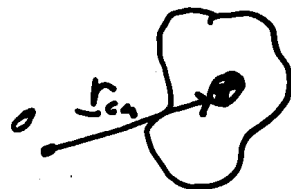
Feb. 26, 2003

PS 1

TODAY: 1) C.O.M. (cont'd) 2) Q & A

Center of mass

$$M_{tot} \mathbf{r}_{cm} = \begin{cases} \sum \mathbf{r}_i m_i \\ \int \mathbf{r} dm \end{cases}$$



$$dm = \begin{cases} \rho dV \\ \rho dA \\ \rho ds \end{cases}$$

ρ = mass per unit

$\begin{cases} \text{volume} \\ \text{area} \\ \text{length} \end{cases}$ as appropriate

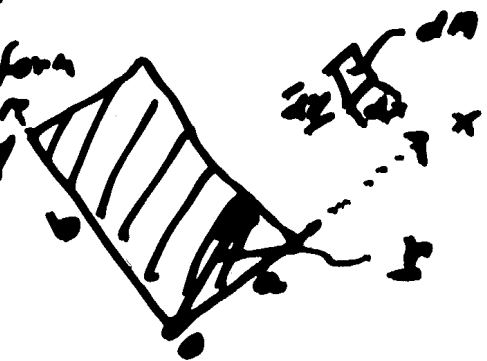
ex) two equal masses (m)



C.O.M. is "objective", location has a well defined place w.r.t. object \Rightarrow

You can use any coord. syst you like to find C.O.M.

ex) uniform rectangular density ρ

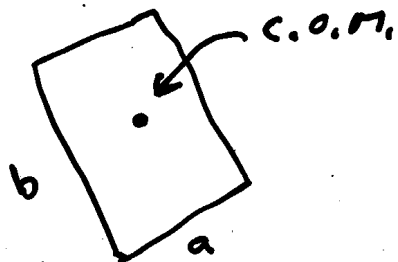


$$\begin{aligned} M_{tot} \mathbf{r}_{cm} &= \int \mathbf{r} dm \\ \rho a b \mathbf{r}_{cm} &= \int_0^a \int_0^b (x\mathbf{i} + y\mathbf{j}) \rho dx dy \\ &= \int_0^b \left(\frac{a^2}{2} \mathbf{i} + ya\mathbf{j} \right) \rho dy \end{aligned}$$

L/26 p92

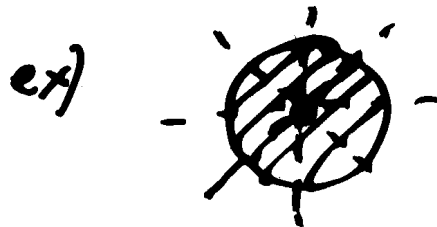
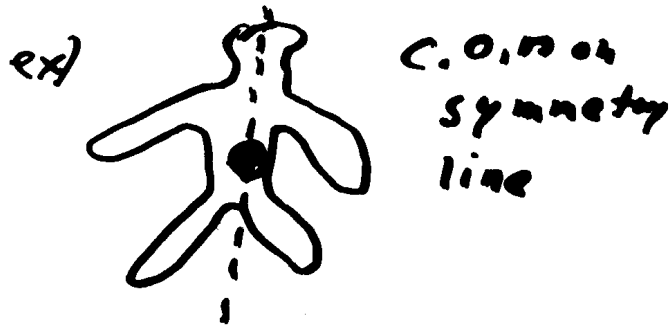
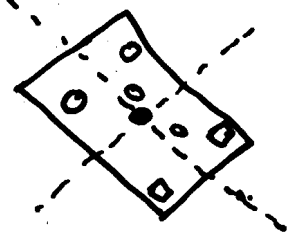
$$= \left(\frac{a^2 b}{2} \hat{i} + \frac{a b^2}{2} \hat{j} \right) \rho$$

$$\Rightarrow \underline{r}_{cm} = \frac{a}{2} \hat{i} + \frac{b}{2} \hat{j}$$

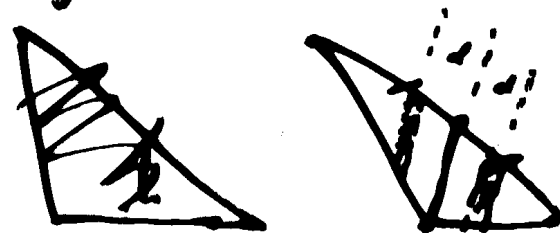


Would have got same location w/ this coord syst. but w/ much more calculation.

Note: C.O.M. respects symmetry of object

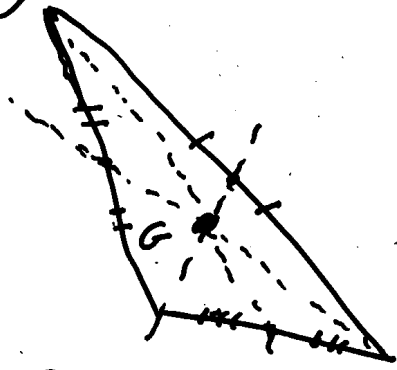


ex) triangle

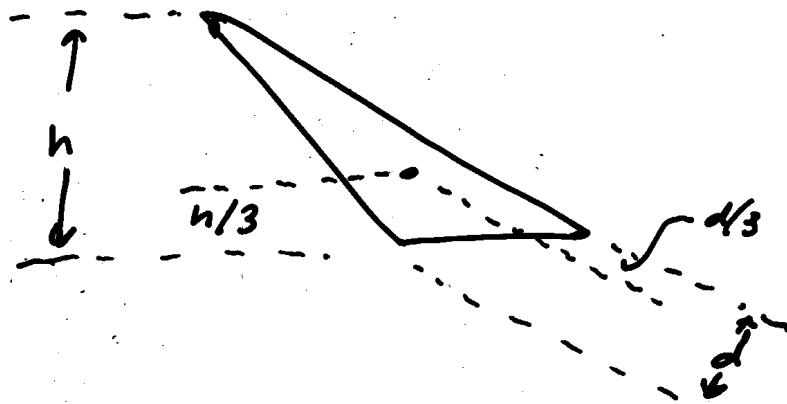


C.O.M. on this line
repeat for all 3 side bisectors - - -

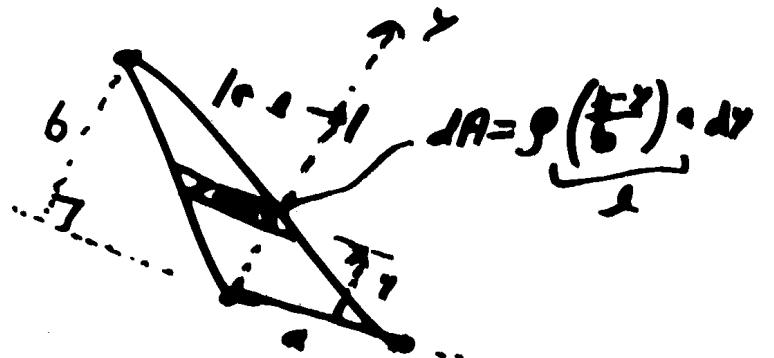
(2/26 page 3)



Geometry fact: G is $\frac{2}{3}$ of the way up from each base



ex) triangle using calculus



$$\{m \bar{y}_m = \int \bar{y} dm\} \cdot \hat{j}$$

$$m \bar{y}_m = \int \bar{y} dm$$

$$= \int_0^b \bar{y} \underbrace{y \frac{b-y}{b} dy}_{dm}$$

$$\left(\int_0^b \bar{y} dm \right) \hat{j} = \int_0^b \left(\frac{y^2}{2} \Big|_0^b a - \frac{y^3}{3} \Big|_0^b \right) \hat{j}$$

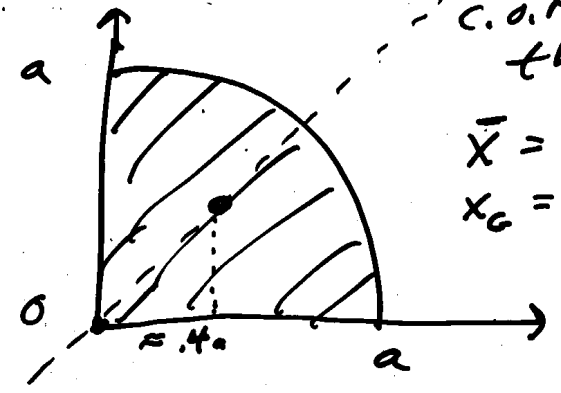
$$= \int_0^b \left(\frac{b^2 a}{2} - \frac{y^3}{3} \right) \hat{j} = \frac{1}{3} a b^2 \hat{j}$$

$$\boxed{\bar{y}_m = b/3}$$

2/26 p99)

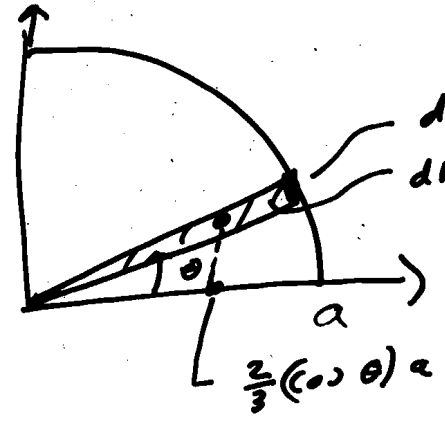
(uniform)

ex) quarter circle



C.M. on this line
 $\bar{X} = \bar{y}$
 $x_c = \frac{4a}{3}$

triangular wedges



$$ds = a d\theta$$

$$dm = \rho \frac{a^2 d\theta}{2}$$

$$M_{tot} x_{cm} = \int x dm$$

$$= \int_0^{\pi/2} \underbrace{\left(\frac{2}{3}(\cos \theta)a\right)}_x \underbrace{\left[\frac{\rho a^2}{2}\right]}_{dm} d\theta$$

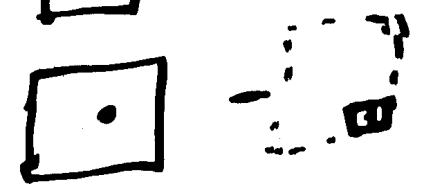
$$= \frac{1}{3} \rho a^3 \int_0^{\pi/2} \cos \theta d\theta$$

$$\frac{\pi a^2}{4} \rho x_{cm} = \frac{1}{3} \rho a^3$$

$$x_{cm} = \frac{4}{3\pi} a \approx 0.4 a$$

$$\underline{r}_{cm} = \frac{4}{3\pi} a (\hat{i} + \hat{j})$$

ex) composite object



$$M_{tot} \underline{r}_{cm} = m_1 \underline{r}_1 + m_2 \underline{r}_2$$

$$(m_1 - m_2) \underline{r}_{cm} = m_1 \underline{r}_1 - m_2 \underline{r}_2$$

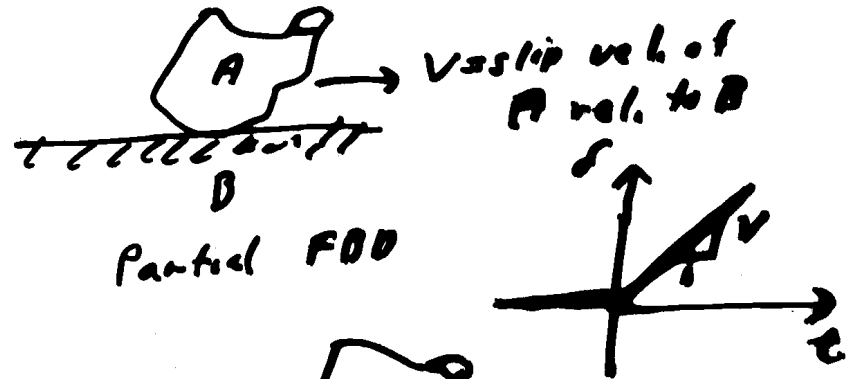
(March 3, 2003 page 1)

TODAY: Friction

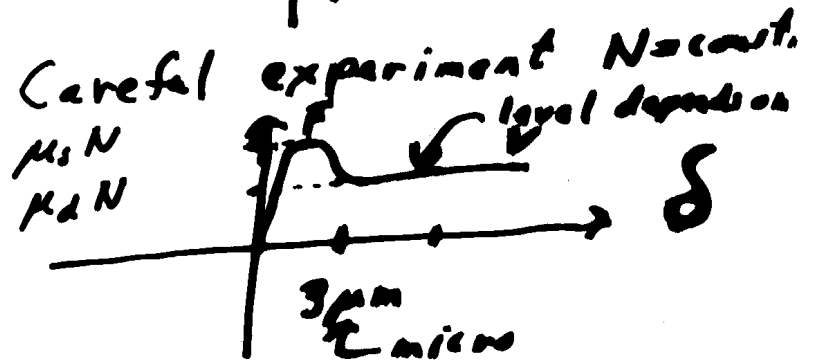
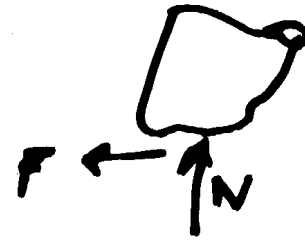
Friction = force which resists or prevents slipping between contacting solids.

Some books: "smooth" = no friction
"rough" = ∞ friction

(actually friction strength is not well correlated to surface roughness unless surface is lubricated = wet w/ water or oil)



Partial FBD



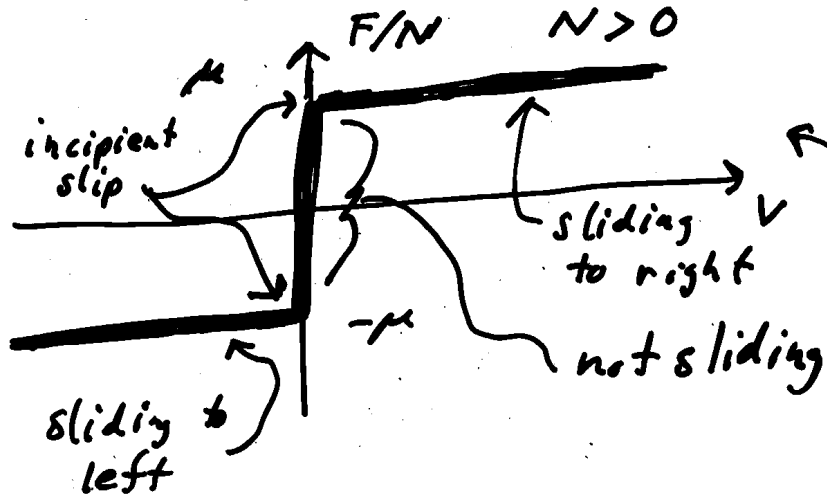
μ_s = static or stationary coeff. of friction


$\mu_d = \mu_k$ = dynamic or kinetic coeff. of friction

This is too complicated \Rightarrow use a simple model \Rightarrow

$$\mu = \mu_s = \mu_k = \mu_d$$

on coeff. of friction



this  curve is the constitutive law for dry friction which we will use

friction law called

5/3 p7.2

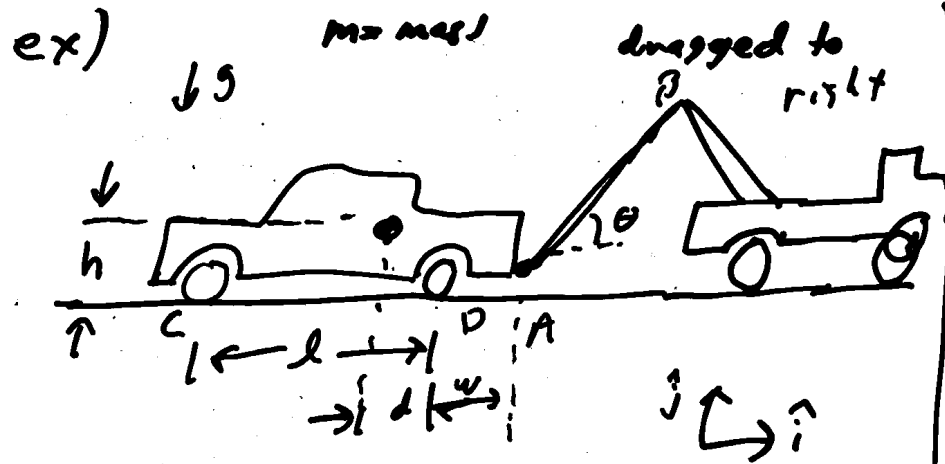
[Coulomb's law /
Amontons's law /
Davinci law] for friction

$$F = \begin{cases} \mu N & \text{for } v > 0 \\ -\mu N & \text{for } v < 0 \end{cases}$$

$$|F| \leq \mu N \quad \text{for } v = 0$$

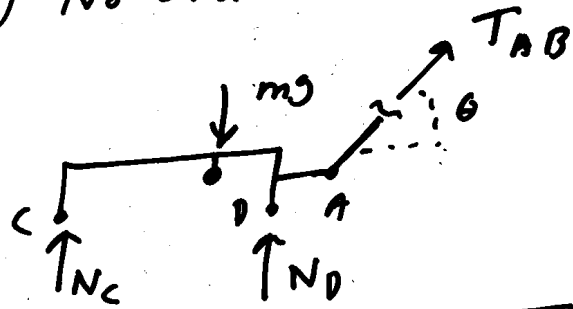
$$-\mu N \leq F \leq \mu N$$

\Rightarrow Try various cases



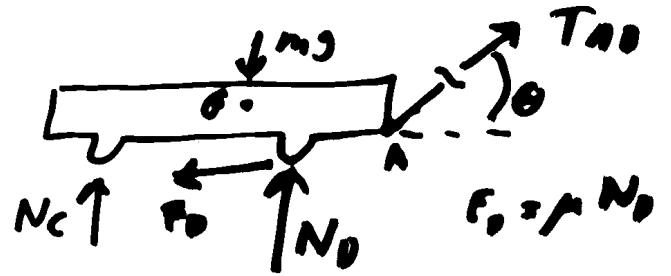
$$T_{AB} = ?$$

sub ex) No brakes on:



$$\sum F_x = 0 \Rightarrow \boxed{T_{AB} = 0}$$

sub ex) Front brakes on (3/3 p.3)



Brake approach:

$$\sum F_x = 0 \Rightarrow -\mu N_B + T_{AB} \cos \theta = 0$$

$$\sum F_y = 0 \Rightarrow N_c + N_B + T_{AB} \sin \theta - mg = 0$$

$$\sum \underline{M}_D = 0 \Rightarrow \int_{C/D} \times N_c \hat{j} + \int_{B/D} \times (mg \hat{j}) + \int_{A/D} \times T_{AB} (\cos \theta \hat{i} + \sin \theta \hat{j}) = 0$$

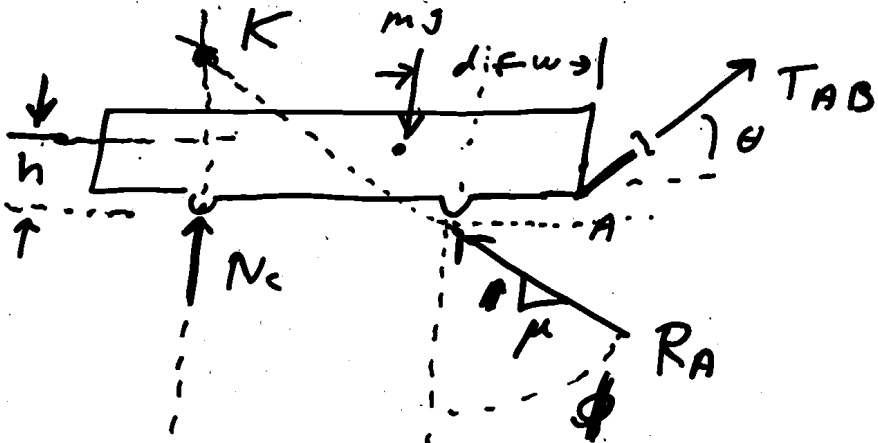
3 eqs

for $N_c, N_B,$

$$T_{AB} \Rightarrow F_b = \mu N_B$$

short cut

Draw FBD again



$\sum \underline{M}/K = 0 \Rightarrow$ one eqn. for
on unknown T_{AB}

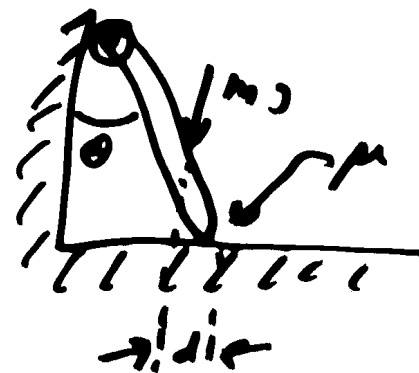
Aside: Alt. description for
friction: friction angle ϕ

$\tan \phi \equiv \mu$

\angle angle between net
contact force &
normal at slip on
incipient slip.

ex)

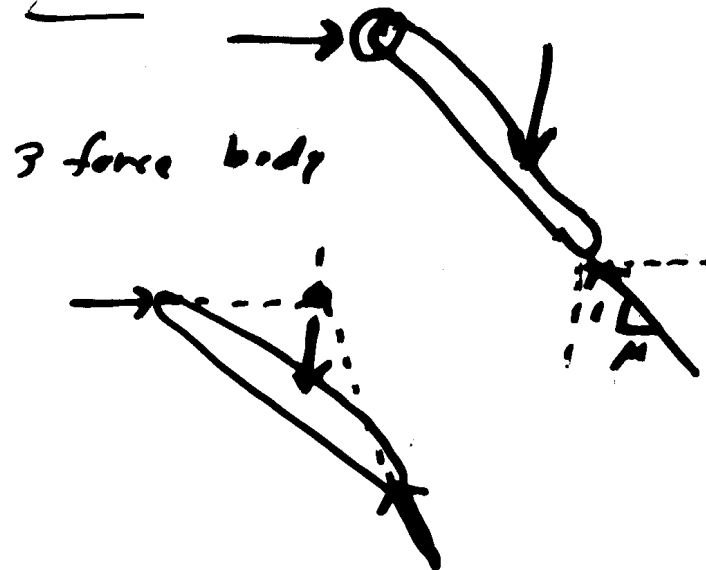
9/3 m4



given μ, θ, m, g how big
can I get for power not
to fall.

Assume incipient slip

FBD:

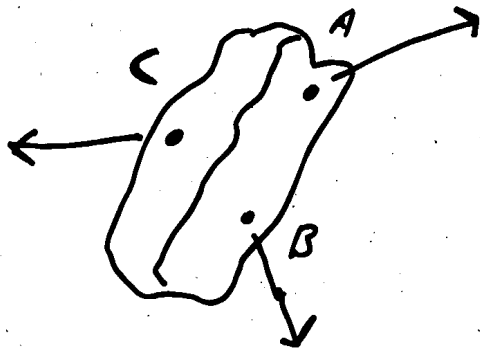


3/5/03 pg. 1/

TODAY; 3 force bodies, friction,
hydrostatics (1st of 2).

3 force body

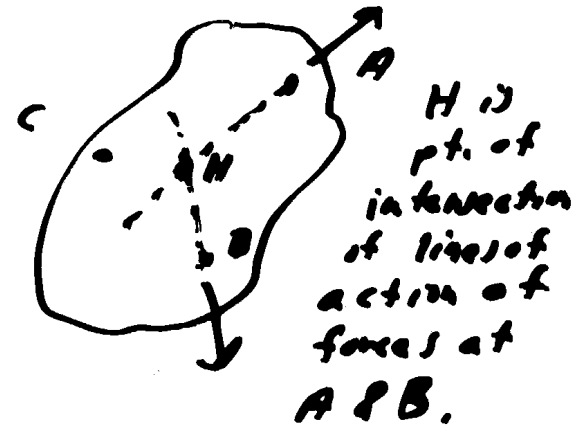
A body (in 2D or 3D) w/
3 forces acting on it.



$\sum M_{AB} = 0$, Force at C only contributes
 \Rightarrow contrib. = 0 \Rightarrow force at C in
plane ABC.

Likewise for $\sum M_{AC} = 0$, $\sum M_{BC} = 0$
 \Rightarrow All forces co-planar in
ABC plane.

\Rightarrow 2D problem in plane ABC.



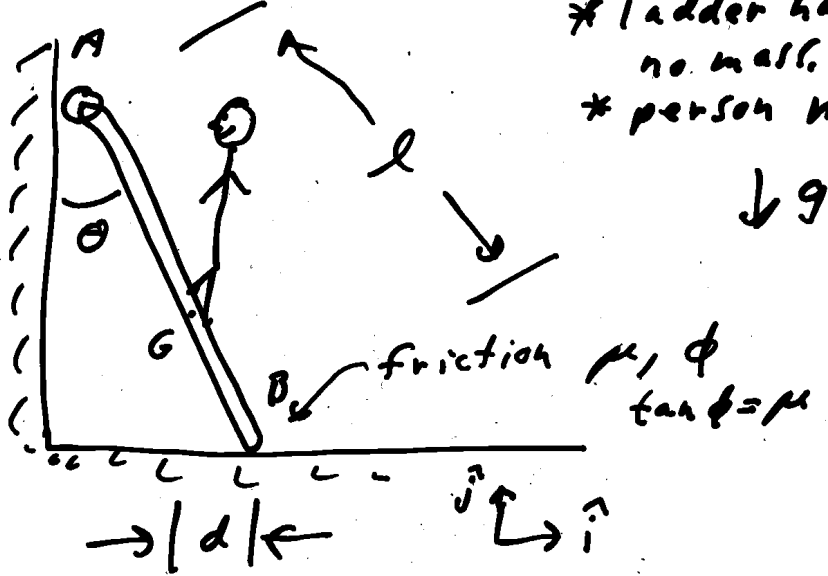
$$\sum M_H = 0$$

\Rightarrow line of action of force at
C intersects H.

\Rightarrow 3 force body all
lines of action intersect
at one pt. in plane ABC.
(that pt can be at ∞ ;
all 3 lines \parallel).

[3/5/04 p. 2]

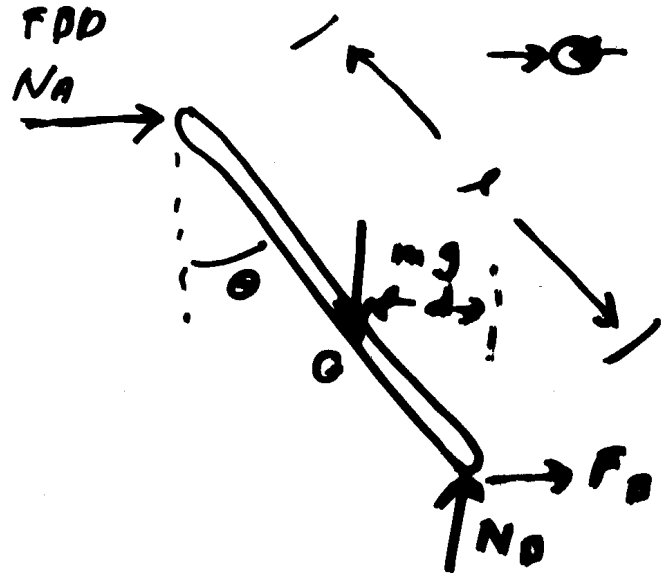
ex) repeat from end of last class



What is biggest d so ladder does not fall?

2 approaches

- 1) Assume no slip at B.
Find reaction at B.
Make sure $F_{fB} \leq \mu N_B$.



Solve for N_A, F_B, N_B as usual ($\sum F = 0, \sum \tau = 0$)
Answer in terms of θ, l, m, g, d .

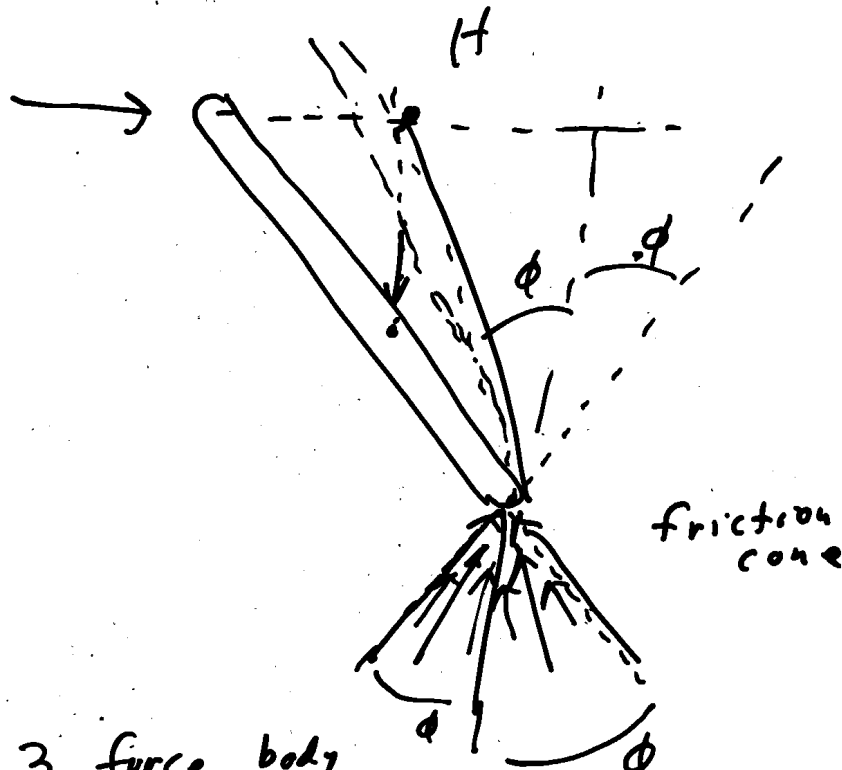
$$\frac{|F_B|}{N_B} \leq \mu$$

gives an inequality for d .
[Will get answer for all d if $\theta < \phi$]

1/5/02

03

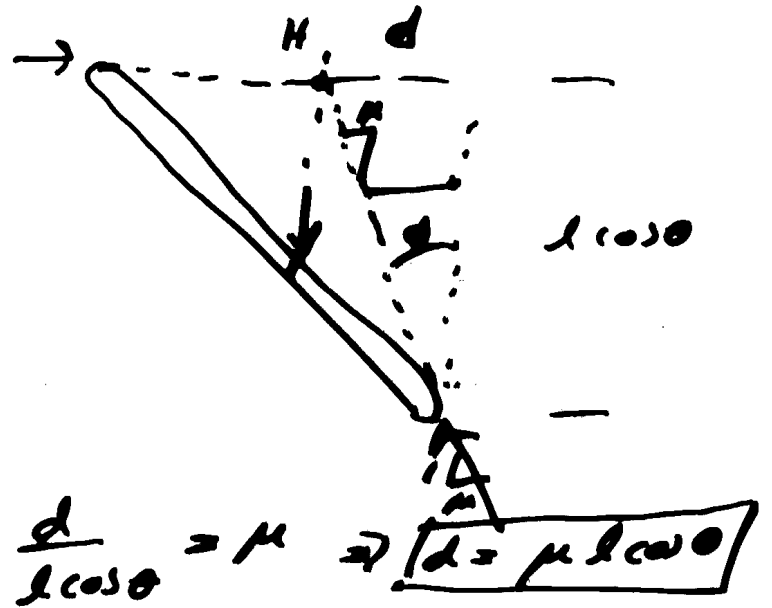
2) More tricky



3) force body

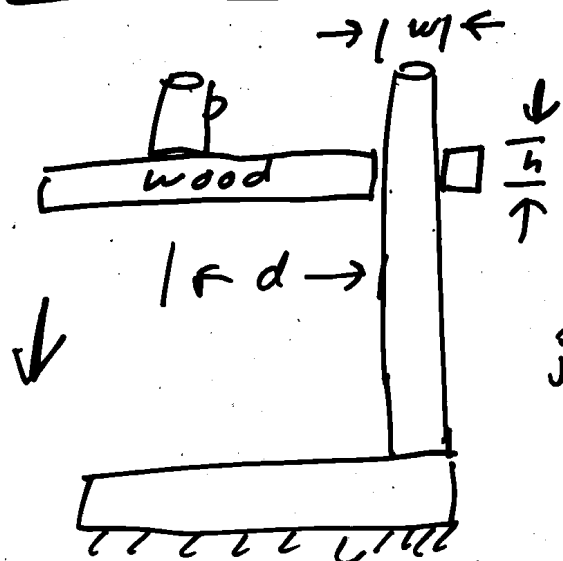
\Rightarrow pt. H in friction cone associated w/ B.

Worst case \Rightarrow H at edge of cone



$$\frac{d}{l \cos \theta} = \mu \Rightarrow \boxed{d = \mu l \cos \theta}$$

[P9, 2003]

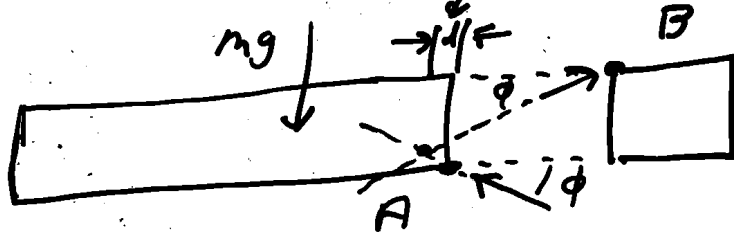


"coffee stand"

min. d
to hold
coffee.

μ given

neglect weight of wood.



How close can d get
& still sat. equilib.

$$\boxed{\tan \phi = \mu}$$

March 3rd, 2003

Lecture given by Professor Alan Zehnder

Topic: Hydrostatics

Special case of distributed loads in which:

- 1) Force is perpendicular to surface it acts on
- 2) Pressure increases linearly with depth

$$P = \rho g h$$

\uparrow \uparrow \uparrow
 density of fluid acceleration of gravity depth

Let's just check units.

$$P = \frac{\text{kg}}{\text{m}^3} \cdot \frac{\text{m}}{\text{s}^2} \cdot \text{m}$$

$$= \frac{\text{kg} \cdot \text{m}}{\text{s}^2} \cdot \frac{1}{\text{m}^2} = \frac{\text{N}}{\text{m}^2} \quad \checkmark$$

For reference,

$$1 \text{ atm} = 10^5 \frac{\text{N}}{\text{m}^2}$$

At the bottom of Cayuga Lake

$$P \approx 10^3 \text{ (}\rho\text{)} \cdot 10 \text{ (}g\text{)} \cdot 100 \text{ (}h\text{)} = 10^6 \text{ Pa} = 10 \text{ atm}$$

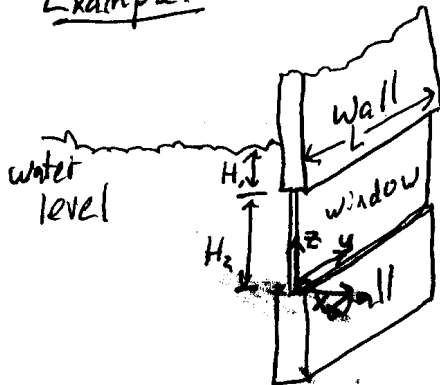
Start from Fundamentals

- Force & moment due to pressure on small area
- Integrate to get total F & M

Build Shortcuts

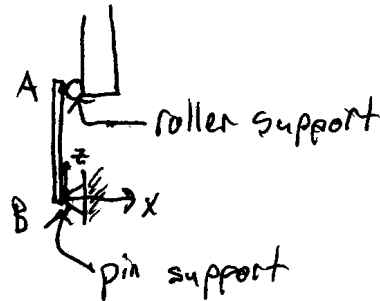
- Replace distributed force by a single force acting at centroid of diagram of force distribution.

Example:

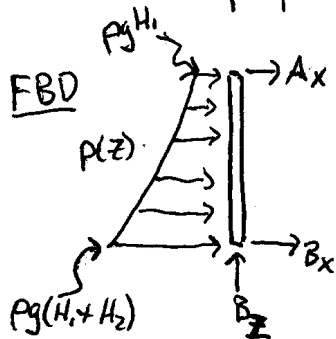


- Find force & moment due to water pushing on window
- Find reaction forces of wall pushing on windows

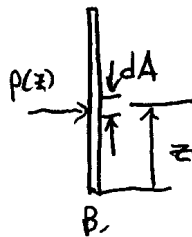
- Idealize problem as 2D
- Idealize as pinned on bottom supports, roller on top



Note: In this coordinate system:
 $P = p(z) = \rho g (H_1 + H_2 - z)$



Force due to $p(z)$ acting on dA



$$dR = p(z) dA \hat{i}$$

$$= p(z) \hat{i} dx dy$$

• Moment (about B) due to $p(z)$ acting on dA

$$dM_B = -p(z) z \, dy \, dz$$

• Integrate to get \underline{R} + M_B

$$\begin{aligned} \underline{R} &= \int_A d\underline{R} = \int_0^L \int_0^{H_2} \rho g (H_1 + H_2 - z) \, dz \, dy \, \hat{i} \\ &= \rho g L \left(H_1 H_2 + \frac{H_2^2}{2} \right) \hat{i} \end{aligned}$$

↑
total force of
water pushing
on window

$$\begin{aligned} M_B &= \int_A dM_B = \int_0^L \int_0^{H_2} -\rho g (H_1 + H_2 - z) z \, dz \, dy \\ &= -\rho g L \frac{H_2^2}{2} \left(H_1 + \frac{H_2}{3} \right) \end{aligned}$$

• Determine Reaction Forces.

$$\sum \underline{F} = \underline{0} = \rho g L \left(H_1 H_2 + \frac{H_2^2}{2} \right) \hat{i} + A_x \hat{i} + B_x \hat{i} + B_z \hat{k}$$

$$\sum M_B = 0 = -\rho g L \frac{H_2^2}{2} \left(H_1 + \frac{H_2}{3} \right) - A_x \cdot H_2$$

Solving, we find that:

$$B_z = 0$$

$$A_x = \frac{-\rho g L \frac{H_2^2}{2} \left(H_1 + \frac{H_2}{3} \right)}{H_2} = -\rho g L \frac{H_2}{2} \left(H_1 + \frac{H_2}{3} \right)$$

$$B_x = -\rho g L \left(H_1 H_2 + \frac{H_2^2}{2} \right) - A_x = -\rho g L H_2 \left(\frac{H_1}{2} + \frac{H_2}{3} \right)$$

Let's try some numbers

Say, $H_1 = 0.5 \text{ m}$

$H_2 = 1 \text{ m}$

$L = 5 \text{ m}$

Approximate $g = 10 \text{ m/s}^2$

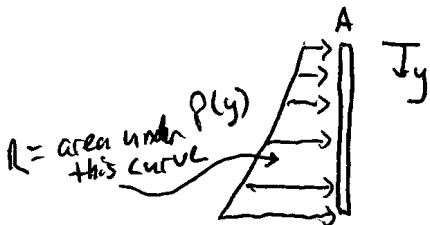
$\Rightarrow \rho g = 10^4$

$A_x = -10^4 / 5 \times (\frac{1}{2})(\frac{1}{2} + \frac{1}{3}) = -\frac{25}{12} \times 10^4 = -2.08 \times 10^4 \text{ N} \approx -4000 \text{ lbs}$

$B_x = -2.92 \times 10^4 \text{ N}$

Check if $R + A_x + B_x = 0$ ✓

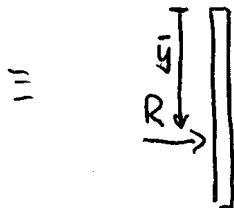
Replace distributed force by single force at centroid of pressure diagram



Force due to pressure

$R = \int_A p(y) dA = L \int p(y) dy$
 ↑ depth into board

$= L \cdot (\text{area under curve of } p(y) \text{ diagram})$



Note: $\bar{y} \equiv$ centroid of area under $p(y)$ curve

Moment around A

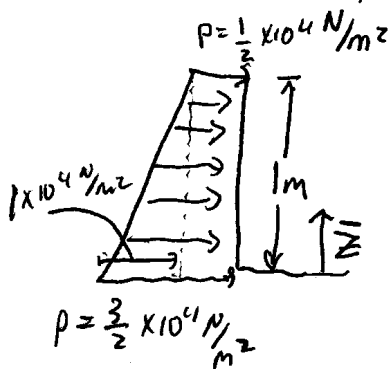
$M_A = \int_A p(y) y dA = L \int y p(y) dy$

$M_A = L \cdot \bar{y} \cdot R$

Centroid of area under $p(y)$ curve

$$\bar{y} = \frac{\int y p(y) dy}{\int p(y) dy}$$

Previous Example:

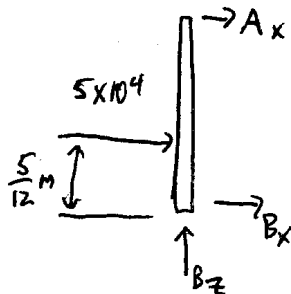


Note: $\rho g \approx 10^4$
 $L = 5$

$$R = 1 \text{ m} \left(\frac{1}{2} \times 10^4 + \frac{1}{2} (1 \times 10^4) \right) 5$$

$$= 5 \times 10^4 \text{ N} = 50,000 \text{ N}$$

$$\bar{z} = \frac{\text{(rectangular area)} \left[\frac{1}{2} (1 \times \frac{1}{2} \times 10^4) \right] + \frac{1}{3} \left[1 \times \frac{1}{2} (1 \times 10^4) \right]}{\text{Total Area}} \times 5 = \frac{5 \left(\frac{1}{4} + \frac{1}{6} \right) \times 10^4}{5 \times 10^4} = \frac{5}{12} \text{ m}$$



13/10/03 ①

This week:

MON: Hydrostatics (cont'd)

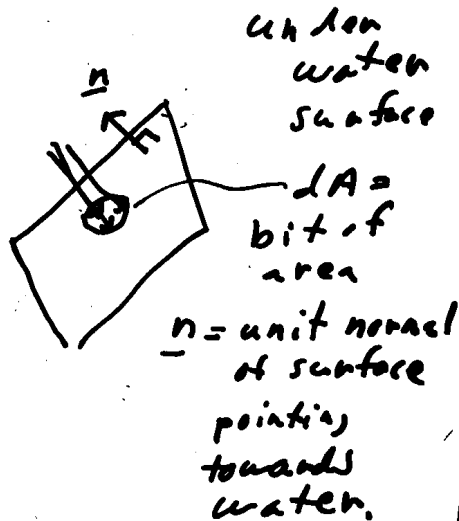
WED: Quiz

FRI: Statils Capstone

Hydrostatics

$$dF = -p dA \underline{n}$$

$p =$ pressure



OF INTEREST:

$$\underline{F}_{TOT} = \int d\underline{F} = \int_S p(-\underline{n}) dA$$

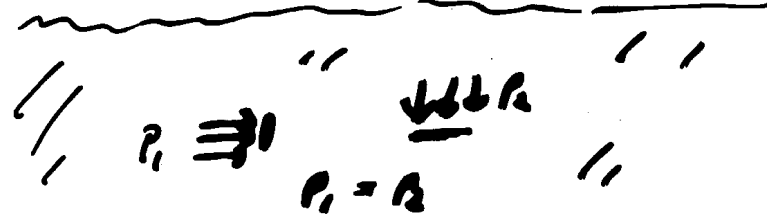
$$\underline{M}_{TOT} = \int \underline{r}_{ic} \times d\underline{F}$$

$$= \int_S \underline{r}_{ic} \times (-p\underline{n}) dS$$

dS is like dA
bit of surface

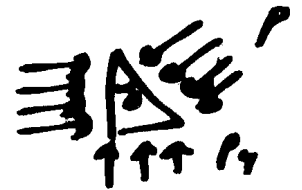
Key facts:

no shear \Rightarrow p is the same for all \underline{n} at a given pt. in space



Why?

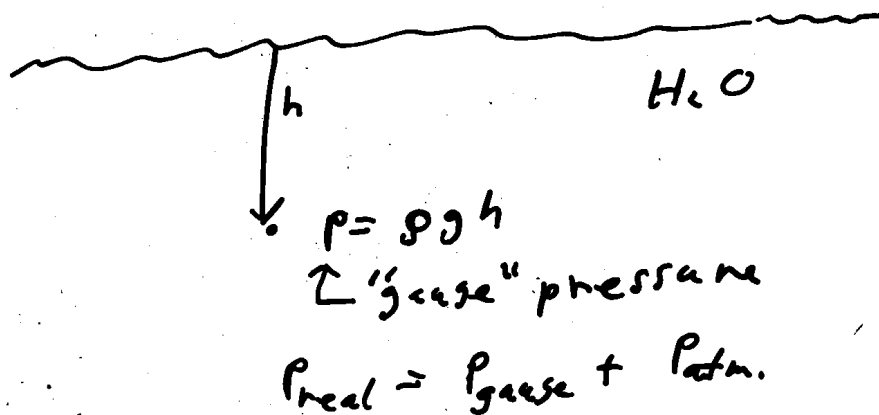
FBD of Δ



$$\begin{aligned} \sum F_x = 0 &\Rightarrow p_2 = p_1 \\ \sum F_y = 0 &= p \end{aligned}$$

3/10/03 (2)

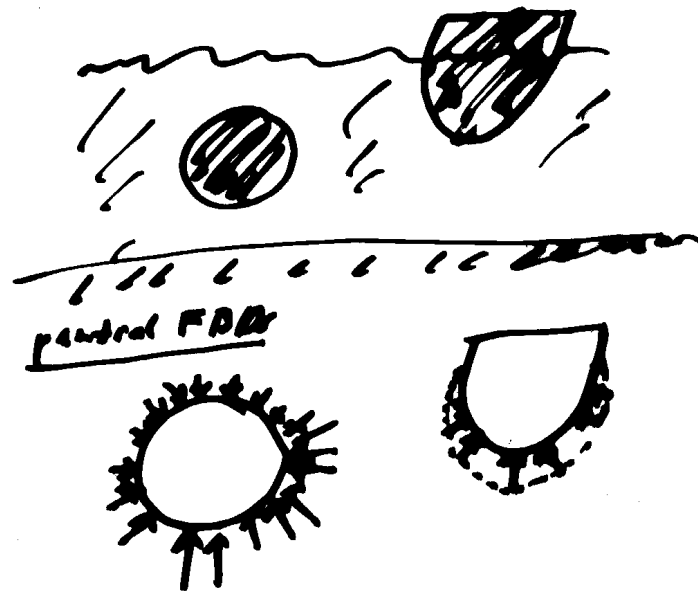
if $\rho =$ density of fluid $=$ const.
[good model for water
almost all the time
[bad model for air over
length scales of kms.



Beginner advice: ignore atmospheric pressure.

Archimedes Principle

"Aha!" Force from H_2O pressure on submerged object.
I.



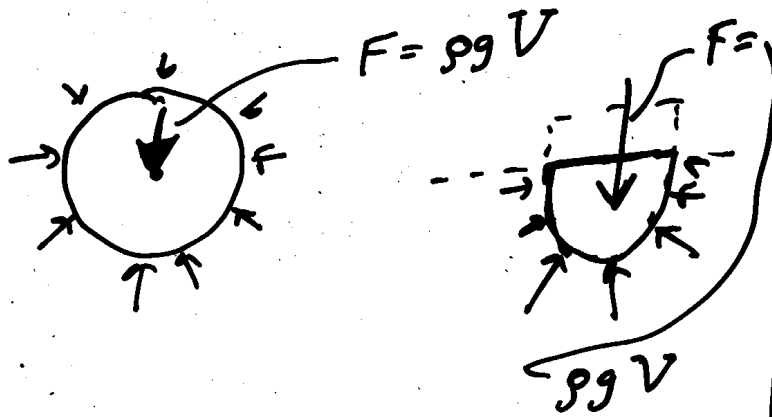
Pressure at each pt. in space same as it would have been if objects replaced w/ water

13/10/03

(3)

Force on object from H_2O
= Force that would have acted on water w/ same shape.

FBD of water



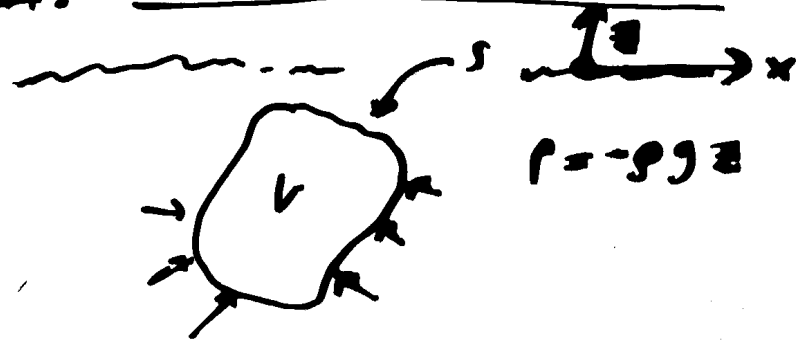
Because water is in equilibrium:

$$\sum \underline{F} = \underline{0}, \quad \sum \underline{m}_i = \underline{0}$$

\Rightarrow net effect of pressure

= a force at centroid
 \vec{L} stat. Equiv. point is up
to w/ magnitude
= weight of
displaced fluid.

II. Modern Calculus approach



$$\underline{F} = \int_S d\underline{F} = \int_S -\rho g z \underline{n} dS$$

Div. Thm. $\int_V \nabla \cdot \underline{F} dV = \int_S \underline{F} \cdot \underline{n} dS$

Simpler version of div. thm. : Gauss's Thm.

$$\int_V \frac{\partial F_1(x, y, z)}{\partial x} dV = \int_S F_1 n_x dS'$$

$$\int_V \frac{\partial F_2}{\partial y} dV = \int_S F_2 n_y dS'$$

$$\int_V \frac{\partial F_3}{\partial z} dV = \int_S F_3 n_z dS'$$

(Gauss Thm. is like one "component" of div. thm.)

For hydrostatics $F_1 = F_2 = F_3$
 = pressure

$$\int_S -P (n_x \hat{i} + n_y \hat{j} + n_z \hat{k}) dS'$$

$$\int_S [+pgz = F = F_1 = F_2 = F_3]$$

$$= \int_V 0 + 0 + +pg \hat{k} dV$$

$$= \boxed{+pgV \hat{k} = F_{tot}}$$

3/14/03

pg. 1

TODAY: (Statics review)

All of Statics

O. You can draw FBD of any system or subsystem.
FBD = picture of syst. w/ all external forces showing (& no internal forces or "inertial" forces).

Laws of Statics (for syst. in static equilib):

I A. $\sum \underline{F}_i = 0$
 \hookrightarrow all ext.

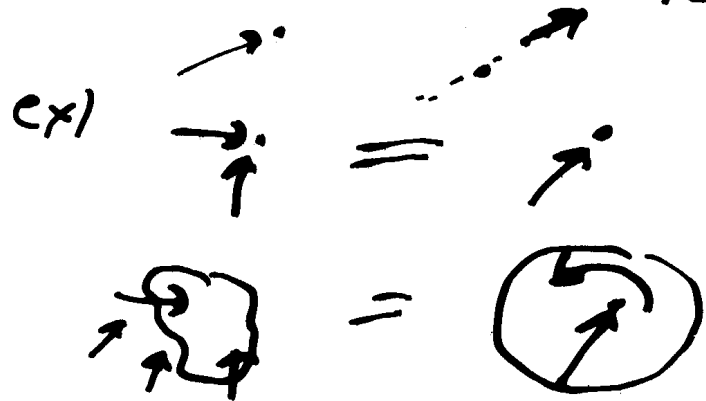
B. $\sum \underline{M}_{i/c} = 0$
 \hookrightarrow any pt.
 (all pts)


$\underline{M}_{i/c} = \underline{r}_{i/c} \times \underline{F}_i$

Observations:

Can, on one FBD, replace a collection of forces w/ "equivalent" system

"equivalent" \equiv same $\sum \underline{F}_i$
 same $\sum \underline{M}_{i/c}$



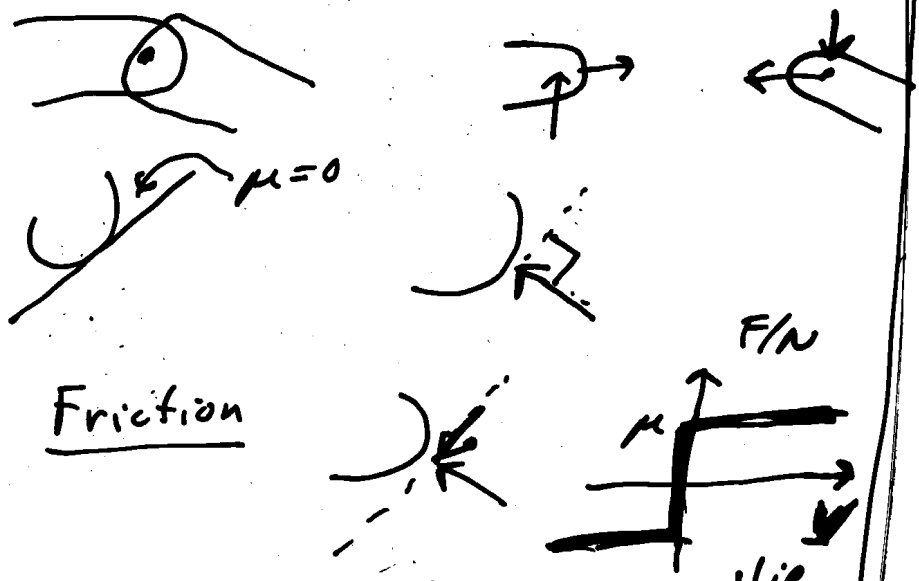
couple \equiv  a representation of a force syst. equiv. to a couple (no net force) by a vector showing net moment.

3/14/03 (2)
(cont'd)

0. Principle of action & reaction.

if A causes \underline{F} on B then
B " $-\underline{F}$ " A w/
same line of action.

Models of interactions and of things.



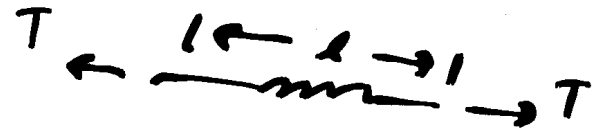
Friction

If a motion is caused or prevented a force or couple shows on FBD.

String



fluid

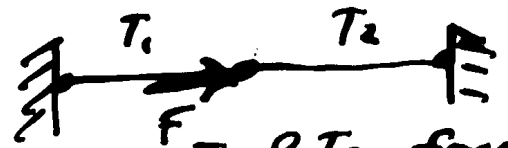


$$T = k(l - l_0)$$

STATICALLY DETERMINATE PROBLEMS.

Problems you can solve using laws of statics.

ex) Not statically determinate



can't find T_1 & T_2 from statics

3/14 ps. 3

Need models of behavior
(force/deformation relations)
to solve statically indeterminate
problems.

What problems are statically
def. ? No single precise
answer.

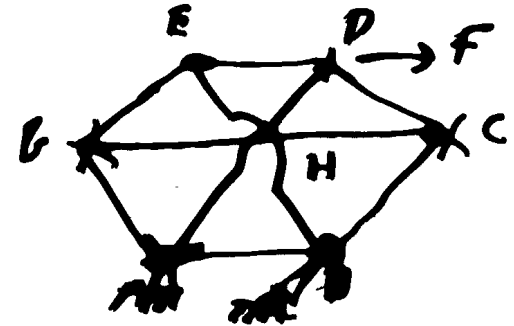
$$\underbrace{\begin{bmatrix} \text{Known} \\ \text{coeff.} \end{bmatrix}}_{\text{known}} \begin{bmatrix} \text{unknown} \\ \end{bmatrix} = \begin{bmatrix} \text{known} \\ \end{bmatrix}$$

eqs. are solvable

Rule of thumb:

$$\# \text{ eqs.} = \# \text{ unknowns}$$

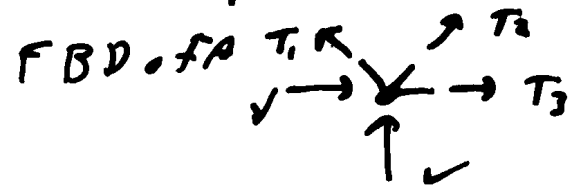
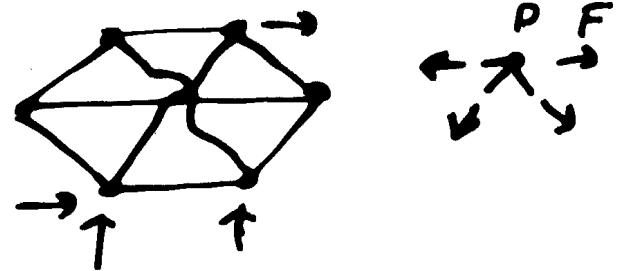
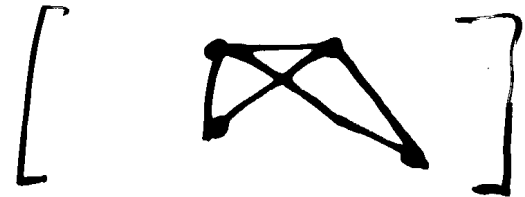
ex) difficult to stress



regular hexagon

no connection at H

given F find all bar tensions



3/14 11.4

All joint eqs

$$2 \times 6 \text{ eqs} = 12 \text{ eqs}$$

12 unknowns

9 tensions
3 react. forces] ~~12~~ 12 unknowns

Need to set up matrix
to solve.

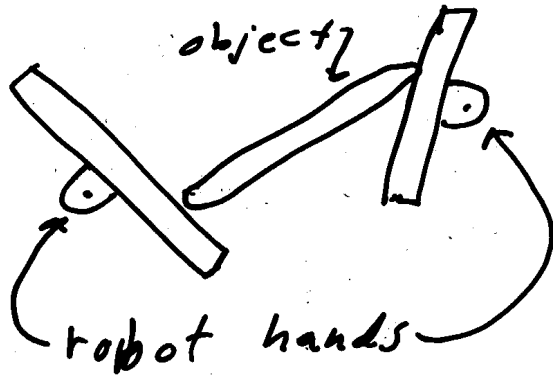
Yet the matrix is
singular (not
solvable)

13124103

①

TODAY 1. Friction hints
2. Stress & strain.

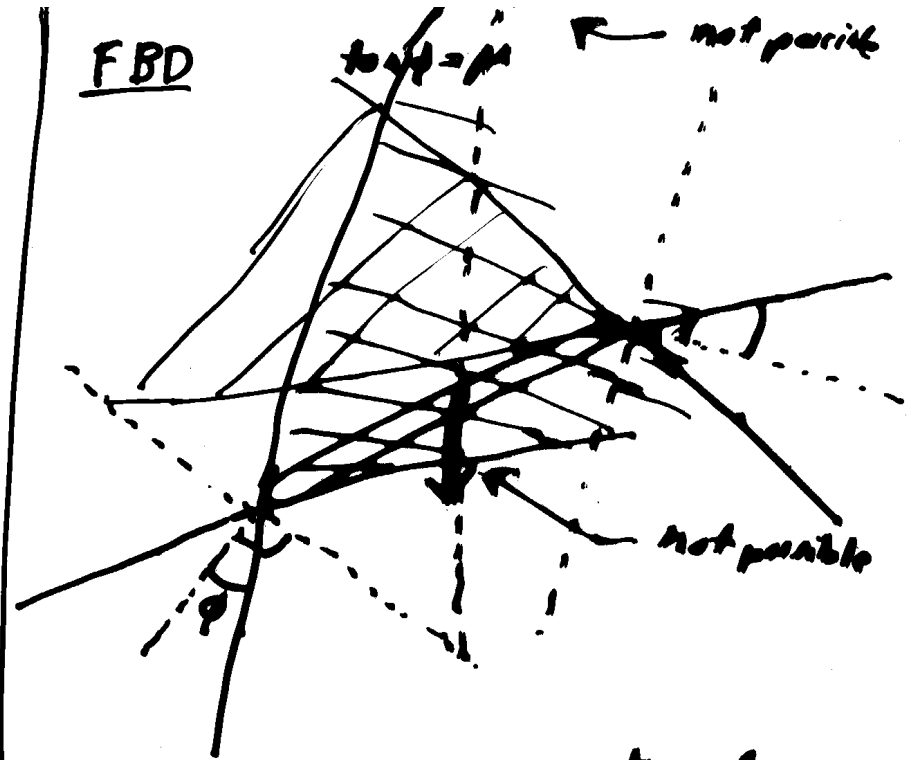
Friction problems.



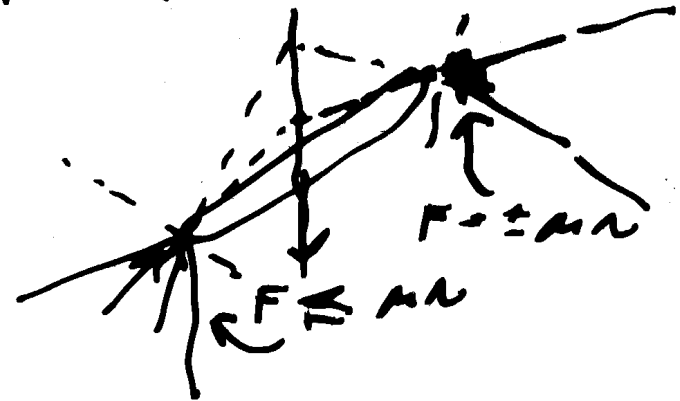
Given a slow hand motion
how does object move.
(Assume statics.)

$$\tan(\phi) \equiv \mu$$

FBD



one of forces is on boundary
of friction cone
Possible situations:

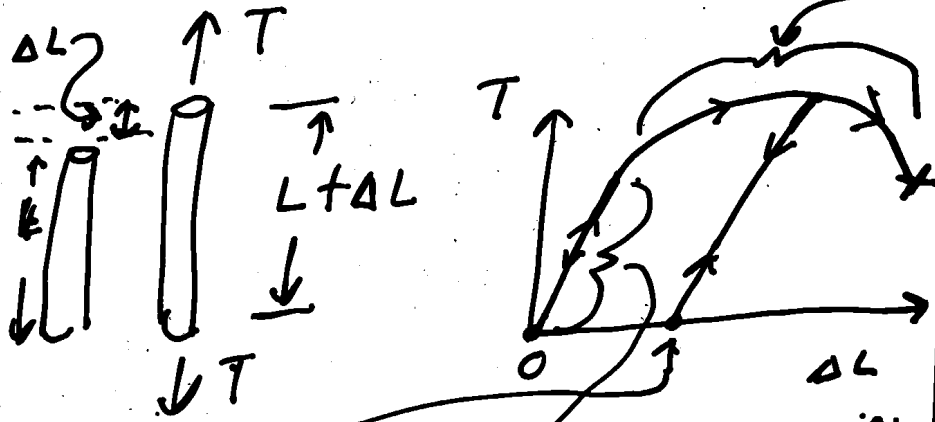


3/24 ②

STRESS & STRAIN (New chaved world.)

[Steel is Jello (Agar Agar)]

Recall Lab 2 : tension test "inelastic"



permanent offset = "damage"
 "proportional elastic"

Assume elastic (linear) for most structural analysis.

Linear regime

$$\Delta L = \frac{TL}{AE}$$

E = elastic modulus
 = Young's modulus

$$\frac{\Delta L}{L} = \frac{1}{E} \frac{T}{A}$$

$$E = \frac{1}{\epsilon} \sigma$$

ϵ = strain = measure of lengthening def. = "tension strain"
 σ = stress = measure of force per unit area.

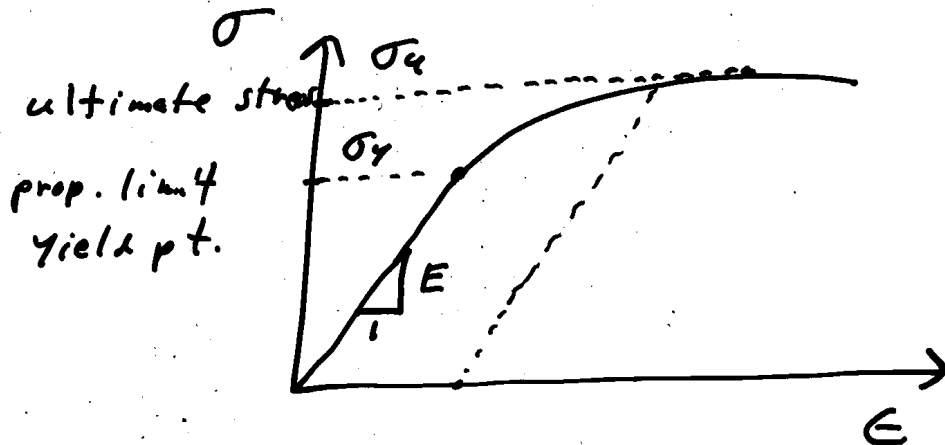
3/24/03

ϵ = "epsilon"

σ = "sigma"

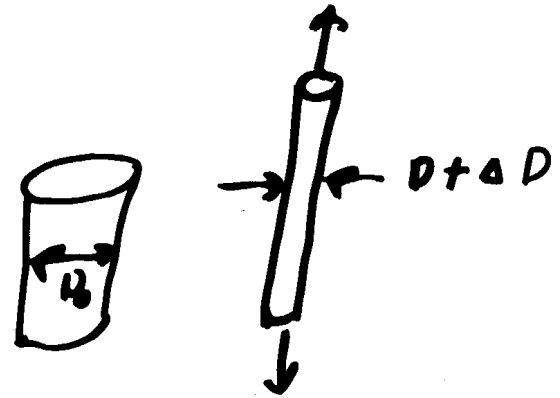
stress is not
a synonym for
strain in mechanics.

Redraw curve!



typically an engineer wants
 $\sigma < \sigma_y$ always.

Poisson contraction



$$\epsilon_t = \text{transverse strain} \\ \equiv \frac{\Delta D}{D}$$

$$\epsilon_t = -\nu \epsilon$$

↳ "Nu"

= Poisson's ratio

$$\nu = -\epsilon_t / \epsilon$$

3/24/03



Some numbers:

$$E = \frac{\sigma}{\epsilon} \text{ in linear prop. regime}$$

$$= \left[\begin{array}{l} 200 \times 10^9 \text{ N/m}^2 \\ 30 \times 10^6 \text{ lbf/in}^2 \end{array} \right] \text{ steel}$$

$$\left[\begin{array}{l} 70 \times 10^9 \text{ N/m}^2 \\ 10 \times 10^6 \text{ lbf/in}^2 \end{array} \right] \text{ Al}$$

$$\sigma_y = \left[\begin{array}{l} 200 \times 10^6 \text{ N/m}^2 \\ 30 \times 10^3 \text{ lbf/in}^2 \end{array} \right] \text{ weak steel}$$

$$\left[\begin{array}{l} 200 \times 10^7 \text{ N/m}^2 \\ 30 \times 10^4 \text{ lbf/in}^2 \end{array} \right] \text{ strongest steel}$$

Note: ϵ at yield is

$$\text{typically } \underbrace{.1}_{.001} \leq \epsilon_y \leq \underbrace{1}_{.01}$$

σ_u usually not much bigger than σ_y
(usually 10-20% but up to 3x)

$$0 \leq \nu \leq .5$$

↳ typically

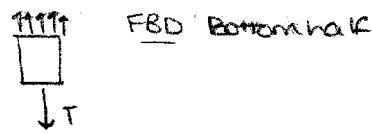
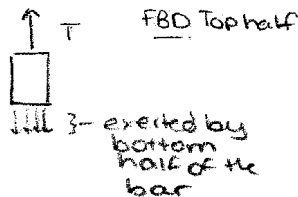
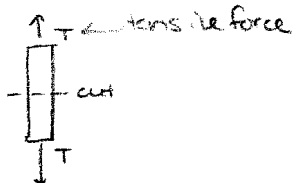
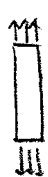
$$-.1 \leq \nu \leq .5$$

Lecture given by: Phoebus Rosakis
 Notes taken by: Bina Lokchander

3-26-03
 200 Lec

Stress - force per unit area

↳ tensile stress



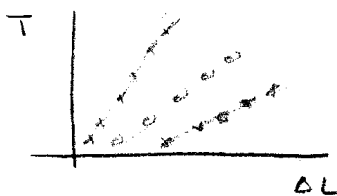
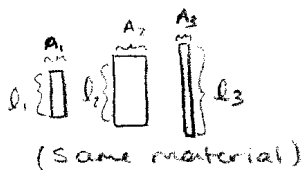
(T)
 Stress - force distributed over area (A)

$$\sigma = T/A \quad \text{Normal stress}$$

Force ÷ Area

units: $N/m^2 = \text{Pascal}$
 psi

$\sigma > 0$ tension
 $\sigma < 0$ compression

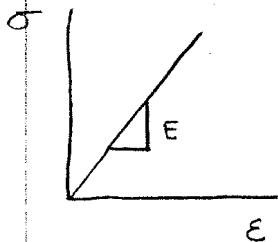


how to get relationship that is independent of strain?

strain $\rightarrow \epsilon = \frac{\Delta l}{l}$

stress $\rightarrow \sigma = \frac{T}{A}$

if plot these against each other, the data points of will collapse together.



Hooke's Law - linear relationship

btw normal stress + normal strain

$$\sigma = E\epsilon$$

* But tangential forces also exist.

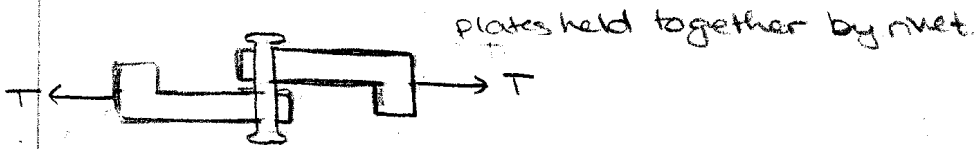
- when forces ^(V) are tangential, over an area (A)



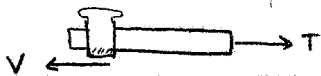
Shear stress

$$\tau = \frac{V}{A}$$

Ex:

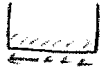


Top half



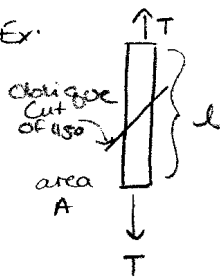
for equilibrium V must = T

distributed tangential force - which can vary along surface.



$$\tau = \frac{V}{A}$$

Ex:



because of this cut + surface it has both tangential + normal stress.

∴ Resolve T into components of shear + normal to surface.



turns out oblique area (A') is greater = $A(\sqrt{2})$

$$T_n = \frac{T}{\sqrt{2}} \quad T_t = \frac{T}{\sqrt{2}} = T_s$$

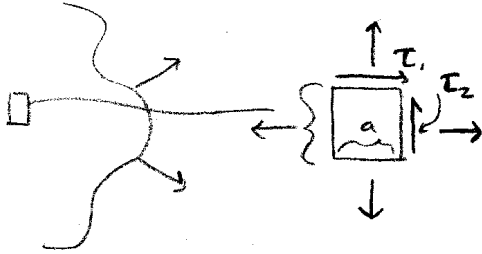
On this cut

$$\text{normal stress } \sigma = \frac{T_n}{A'} = \frac{T/\sqrt{2}}{A\sqrt{2}} = \frac{T}{2A}$$

$$\text{shear stress } \tau = \frac{T_t}{A'} = \frac{T}{2A}$$

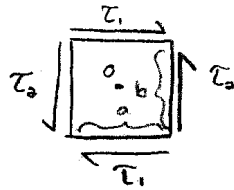
State of stress on a material surface depends on forces and orientation of surface

General State of Stress (2D)



shear stress
what is relation btw τ_1 + τ_2 ?

if No Normal stress

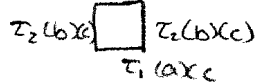


thickness is 'c'

relationship btw τ_1 + τ_2 ?
Moment Balance about o

FBD

$\tau_1(a)(c)$ forces = stresses * areas.

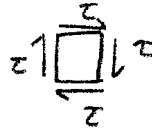


$$\sum M_{10} = -2\tau_1(a)(c)\left(\frac{b}{2}\right) + 2\left(\frac{a}{2}\right)(\tau_2)(b)(c) = 0$$

$$(-\tau_1 + \tau_2)abc = 0$$

$$\tau_1 = \tau_2$$

* shear stress acting on \perp faces must be same



not a scalar or vector, stress turns out to be a matrix. [294 - shear transformation etc]

response to shear stress is a shape change



angle change is known as shear strain $\rightarrow \gamma$

Hooke's Law for Shear

$$\tau = G\gamma$$

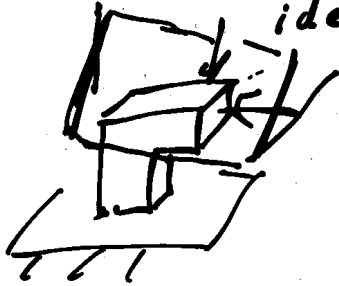
↑
Shear Modulus

(ν) (E) (G) } not indep of each other. relation exists
Poisson's Ratio, Young's Mod, Shear Mod

3/28/01

TODAY: More stress & strain

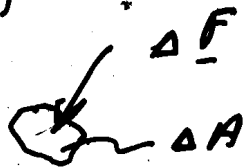
Stress: Force per unit area on a surface exposed by a FBD cut. Usually cut plane inside some solid (same idea works for fluids)



\underline{n} = unit vector + to surface pointing out

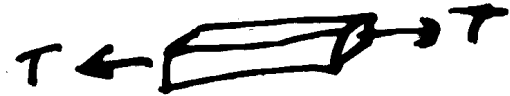
ΔA = a little bit of area

= $(\Delta l)^2$



The word "stress" something used to describe:

a scalar: $\sigma = \frac{T}{A} = \frac{\text{tension}}{\text{stress}}$



in this context σ is a scalar like "tension" is a scalar.

the stress vector:

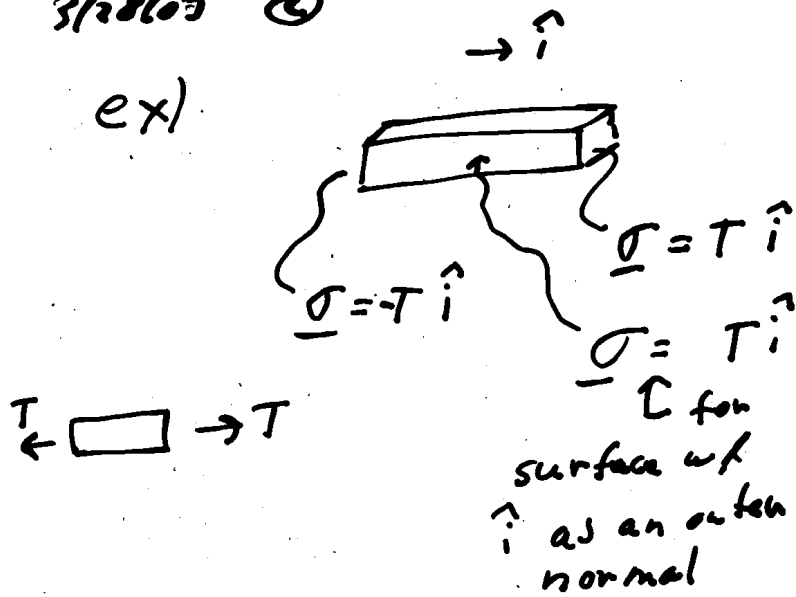
$$\underline{\sigma} = \frac{\Delta \underline{F}}{\Delta A}$$

for small ΔA
" $\Delta A \rightarrow 0$ "

$\underline{\sigma}$ will depend on where in solid & $\underline{\sigma}$

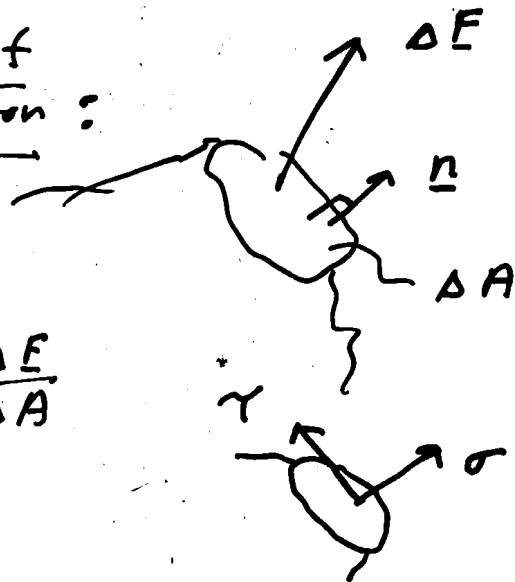
3/28/03 (3)

ex)



a component of stress vector:

$$\underline{\sigma} = \frac{\Delta F}{\Delta A}$$



σ = normal stress
 = tension stress
 = comp of $\underline{\sigma}$ in \underline{n} dir.
 = $\underline{\sigma} \cdot \underline{n}$
 τ = shear stress
 = comp of $\underline{\sigma}$ tangent to surface

$$\left[\begin{array}{l} |\underline{\sigma} - \sigma \underline{n}| \quad 3D \\ \underline{\sigma} \cdot \underline{e}_t \quad 2D \\ \quad \quad \underline{t} \text{ unit tangent} \end{array} \right.$$

The stress matrix: 3×3

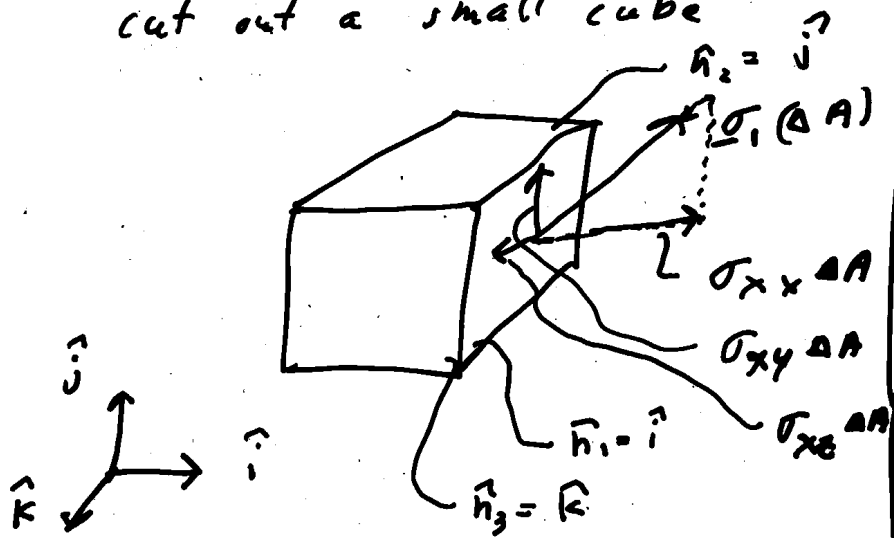
comps of $\underline{\sigma}$ for

$$\underline{n} = \underline{e}_1, \underline{e}_2, \underline{e}_3$$

3/28/03

③

cut out a small cube



$\underline{\sigma}$ on $n_i = \hat{i}$ face

$$\underline{\sigma} = \sigma_{xx} \hat{i} + \sigma_{xy} \hat{j} + \sigma_{xz} \hat{k}$$

Like wise for other 2 surfaces

$$[\underline{\sigma}] \text{ matrix} = \begin{bmatrix} \sigma_{xx} & \sigma_{yx} & \sigma_{zx} \\ \sigma_{xy} & \sigma_{yy} & \sigma_{zy} \\ \sigma_{xz} & \sigma_{yz} & \sigma_{zz} \end{bmatrix}$$

↑ Comps of

$\underline{\sigma}$ vector for $n = \hat{i}$

Stress tensor:

Explanation 1: Huh?

Expl. 2: Forget about it
you'll learn in
courses like
TAM 610, 663 etc.

Expl. 3: Analogy

$\underline{\sigma}$ tensor is to
[$\underline{\sigma}$] like

3/20/03 (4)

\underline{V} is to its list of components.

expl. 4; $\underline{\sigma}$ is a function

$$\mathcal{F} : \mathcal{F}(\underline{n}) = \underline{\sigma}$$

that has \underline{n} as input and $\underline{\sigma}$ as output. [A linear function. "A linear operator"]

Aside on "continuum mechanics"

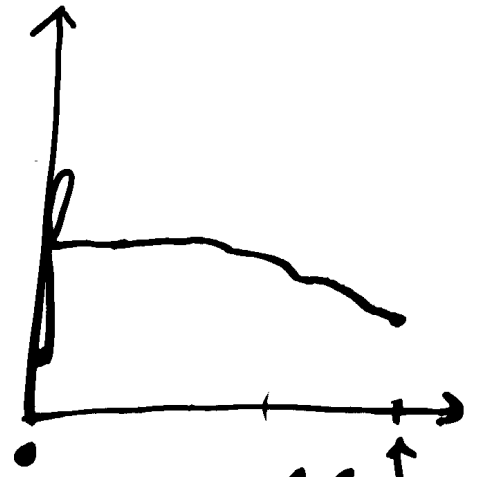
$$\underline{\sigma} = \frac{\Delta F}{(\Delta l)^2}$$

small Δl
small $\Delta A = \Delta l^2$

At some point in space inside a solid. (perpendicular)

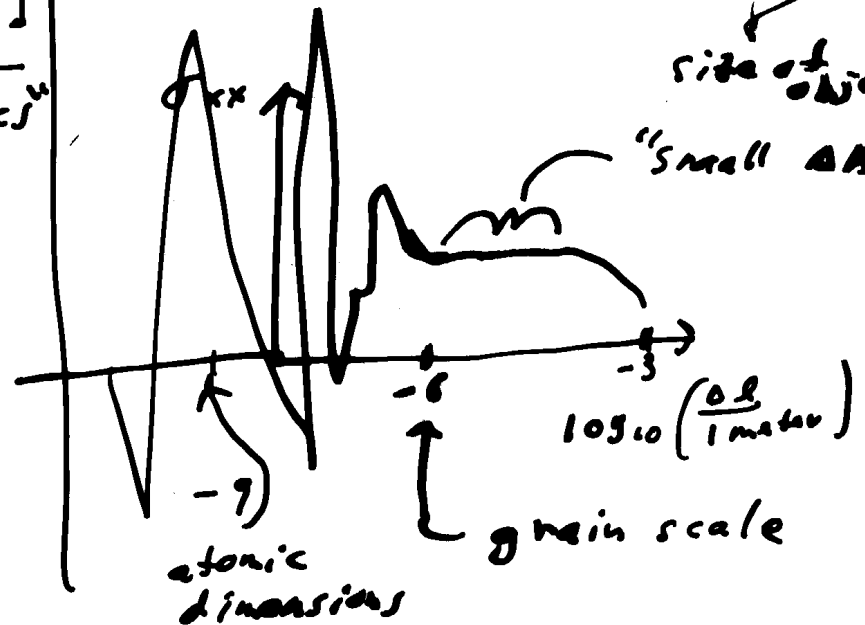
$$\sigma_{xx} = \frac{\Delta F}{\Delta A} \cdot \hat{i}$$

(fix $\underline{n} = \hat{i}$)



size of object

"small ΔA "



atomic dimensions

grain scale

3/28/03 ⑤

Back to reality

Recall: Big & small numbers

March 31, 03 ① A. Poisson's ratio

TODAY: B. Collections of baby
in tension

A. Poisson's ratio

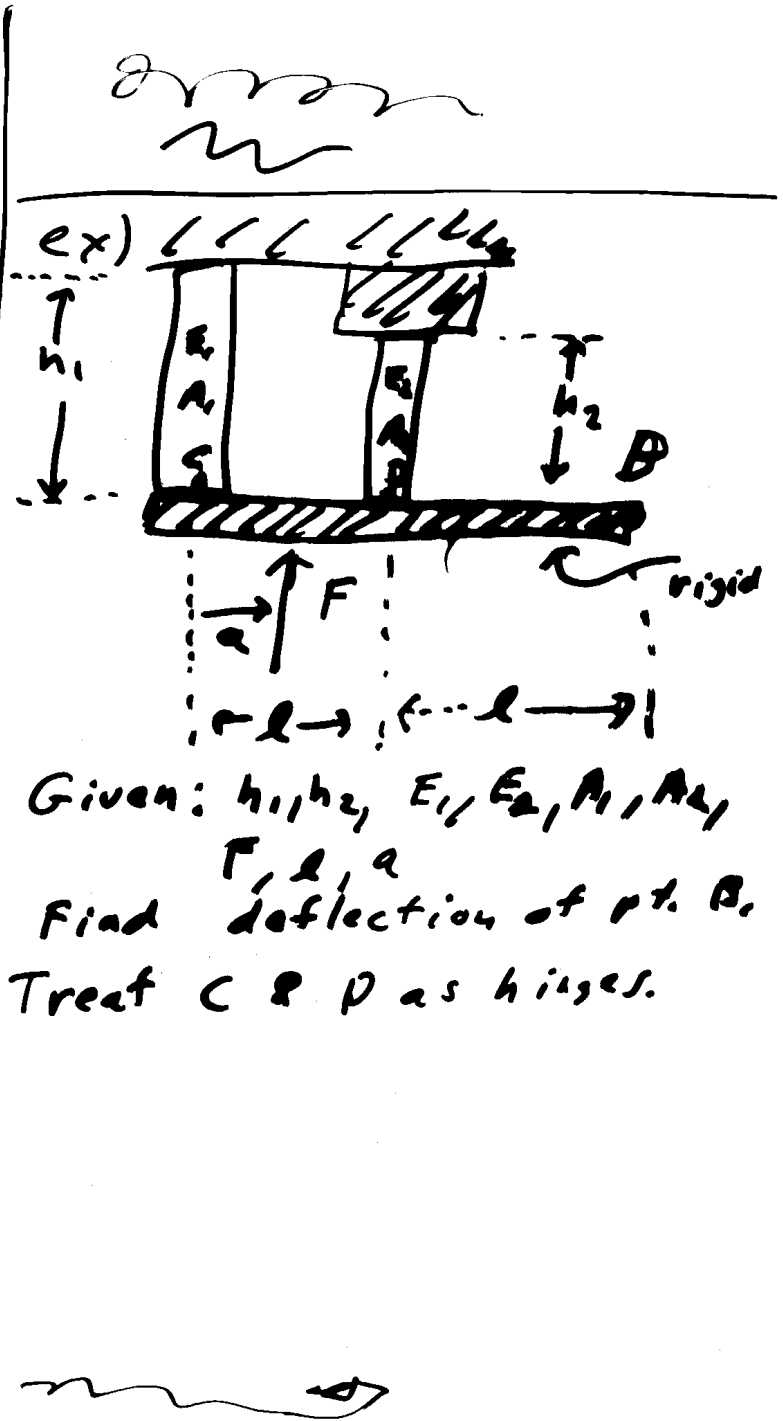
ex) Metals $\nu \approx .2 \text{ or } .3$

Rubber, $\nu \approx .5$

Jello .49

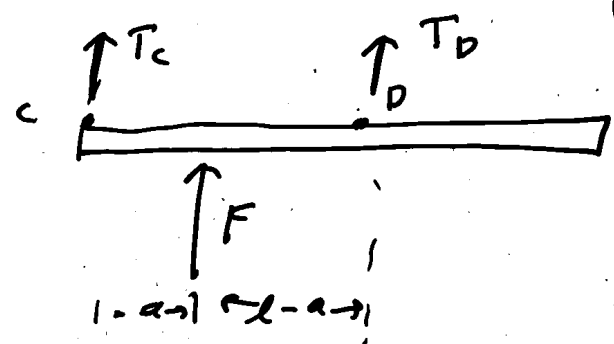
(.5 = ν corresponds to
const. volume)

Recall $\nu = - \frac{\epsilon_t}{\epsilon_l}$



Given: $h_1, h_2, E_1, E_2, A_1, A_2, F, l, a$
 Find deflection of pt. B.
 Treat C & D as hinges.

3/31/03 (1)
 FBD of CPB (S) ICAB



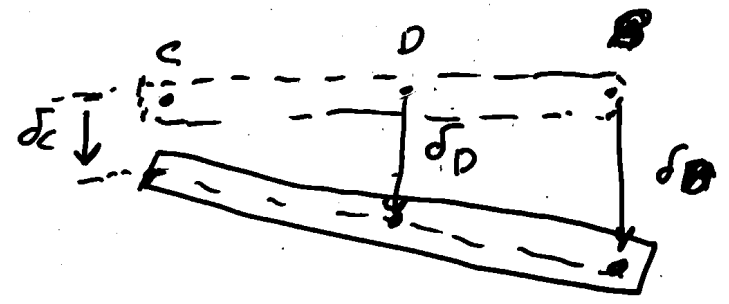
Sign Convention
 Tension is positive

$$\sum M = 0, \sum F_i = 0 \Rightarrow \left. \begin{aligned} T_D &= \frac{-a}{l} F \\ T_C &= \frac{-(l-a)}{l} F \end{aligned} \right\} (1)$$

Kinematics, Geometry

Assume small slopes \Rightarrow neglect horiz. displacements

Increase in length is positive



slope = slope

$$\frac{\delta_D - \delta_C}{l} = \frac{\delta_D - \delta_B}{l}$$

~~2/2/03~~

$$\delta_A = 2\delta_D - \delta_C \quad (2)$$

Mech. Properties, Constit Laws

bar C $\delta = \frac{Tl}{AE}$
 $E_1 = \frac{\sigma_D}{\epsilon}$
 $l = h_1, h_2$

$$\frac{\delta_D}{h_2} = \frac{1}{E_1} \frac{-a}{l} \frac{F}{A_2}$$

$$\delta_D = \frac{-a h_2}{l A_2 E_1} F$$

$$\delta_D = \frac{-a h_2}{l A_2 E_2} F$$

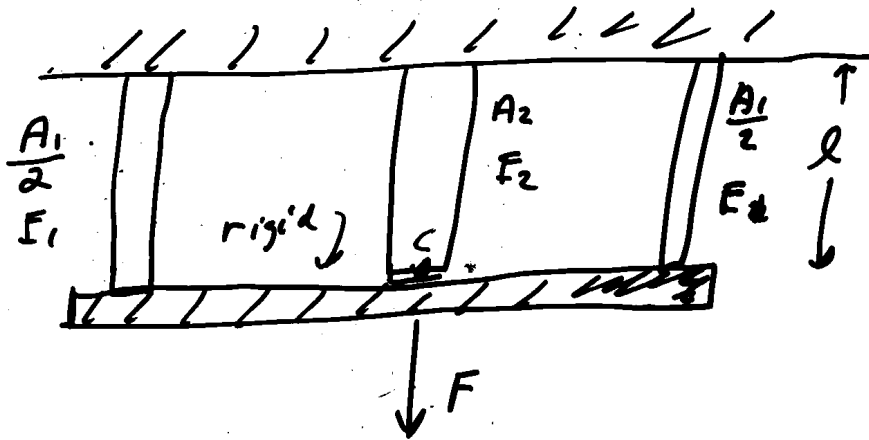
3/31/03 (3)

$$\delta_c = \frac{-(l-a)hc}{2A_1 E_1} F$$

(3) apply to (2)

\Rightarrow

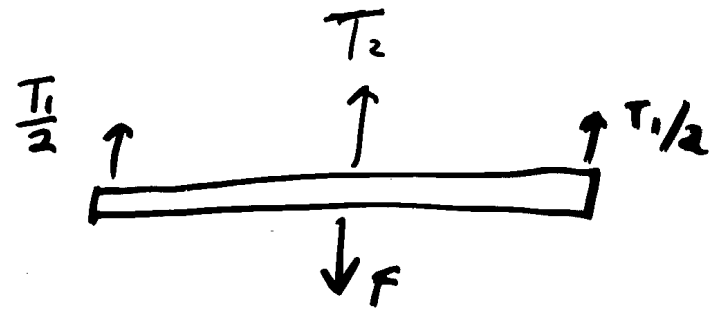
$\delta_A =$ mess
involving things we know



Given F, A_1, E_1, A_2, E_2, l

Find δ_c .
Symmetry \Rightarrow no rotation

Mechanics



$$\sum F_i = 0, \sum M = 0 \Rightarrow$$

$$T_1 + T_2 = F \quad (1)$$

can't find T_1 & T_2 from statics alone \Rightarrow

statically indeterminate

3/31/00 (1)

Kinematics / Geometry

$$\delta_1 = \delta_2 \quad (2)$$

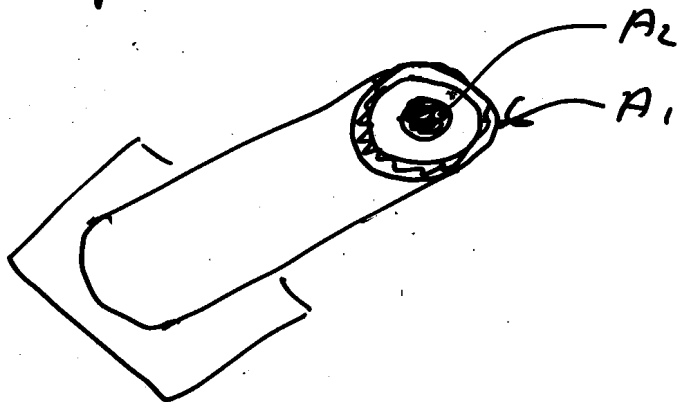
Material Properties

$$T_1 = \frac{E_1 A_1 \delta_1}{l}$$

$$T_2 = \frac{E_2 A_2 \delta_2}{l}$$

(1), (2), (3) to get answer.

Graphical derivation



$$F = T_1 + T_2$$

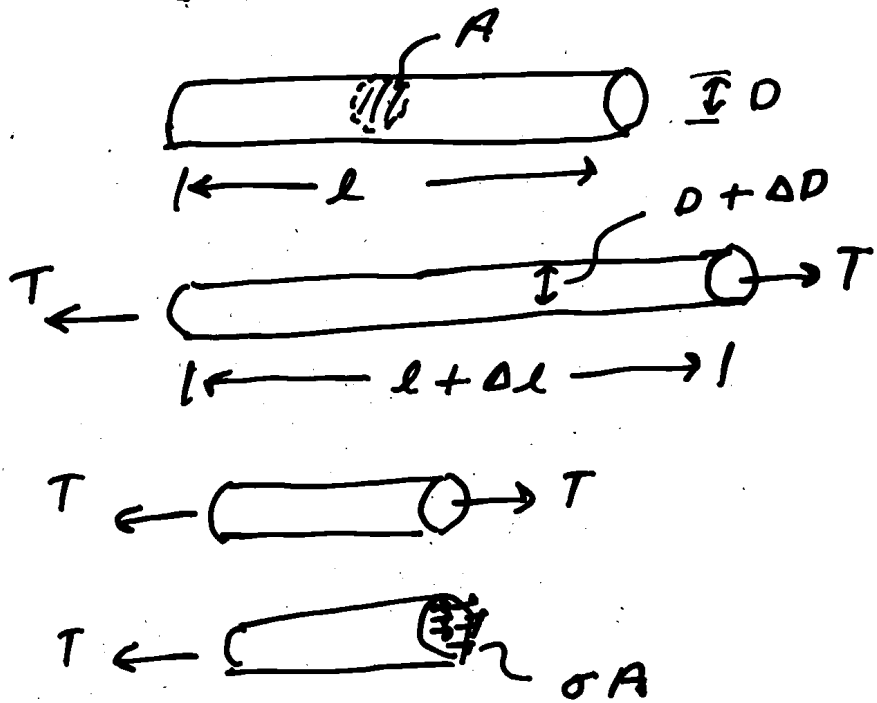
$\uparrow \frac{E_1 A_1 \delta}{l} \qquad \uparrow \frac{E_2 A_2 \delta}{l}$

$$F = \left[\frac{E_1 A_1}{l} + \frac{E_2 A_2}{l} \right] \delta$$

$$\delta = \frac{F}{[\dots]}$$

[4/02/03] ①

TODAY: More tension



T = tension in bar (pos. if pulling on)
 A = cross section area ends
 σ = tension stress
 $\equiv T/A$ (same sign convention as for T)

$\epsilon = \epsilon_s = \text{elongation stretch}$
 $\equiv \frac{\Delta l}{l}$ increase
 $\Delta l = \text{change in length}$

Linear Elastic constitutive law

$\sigma = E \epsilon$ "Young"

$T = \frac{EA}{l} \Delta l$ " $T = k \delta$ "

$\left(\epsilon = \frac{1}{E} \sigma, \Delta l = \frac{l}{EA} T \right)$

Think of rod as a spring w/ spring constant

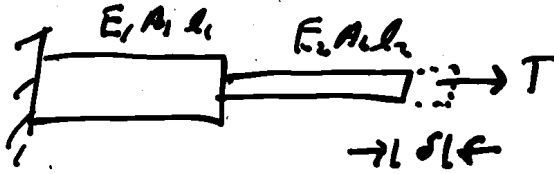
$k = \frac{EA}{l}$

$\frac{\Delta D}{D} = -\nu \frac{\Delta l}{l}$ Poisson's ratio

$\nu = -\frac{\epsilon_t}{\epsilon_l}$

4/2/09 (2)

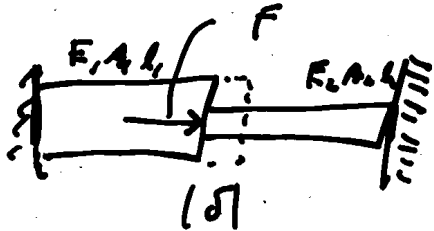
ex 1)



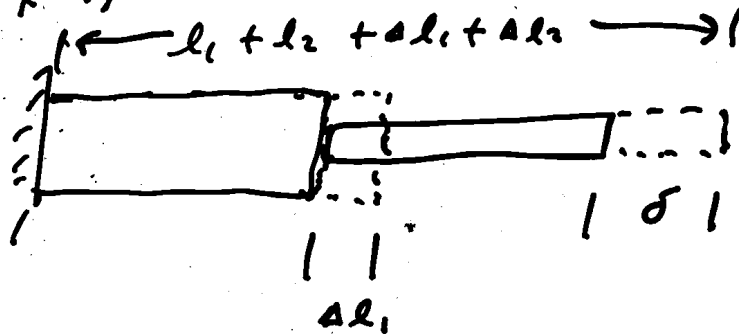
compare

gives δ find T

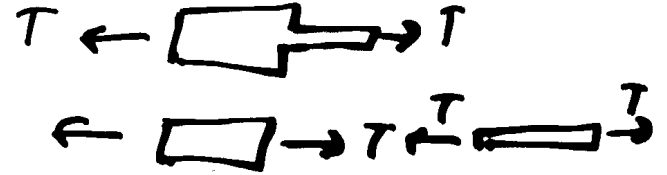
ex 2)



ex 1)



FBDs



Geometry

$$\delta = \delta_1 + \delta_2$$

$\delta_1 = \frac{T l_1}{A_1 E_1}$
 $\delta_2 = \frac{T l_2}{A_2 E_2}$

$$\delta = \left[\frac{l_1}{A_1 E_1} + \frac{l_2}{A_2 E_2} \right] T$$

$$T = \frac{\delta}{\left[\frac{l_1}{A_1 E_1} + \frac{l_2}{A_2 E_2} \right]}$$

$$K = \frac{1}{\frac{l_1}{A_1 E_1} + \frac{l_2}{A_2 E_2}}$$

4/2/03

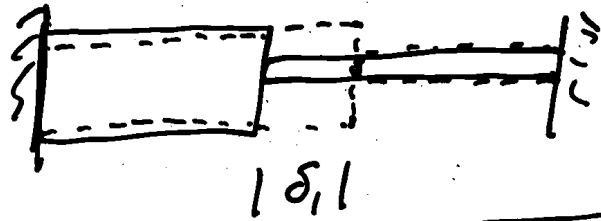
③

$$\Rightarrow \delta_1 = \frac{T l_1}{A_1 E_1}$$

$$= \frac{l_1 / (A_1 E_1)}{[\dots]} \delta$$

$$\delta_2 = \dots$$

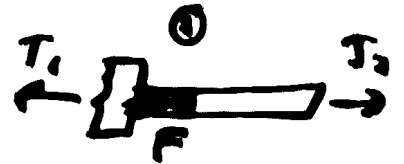
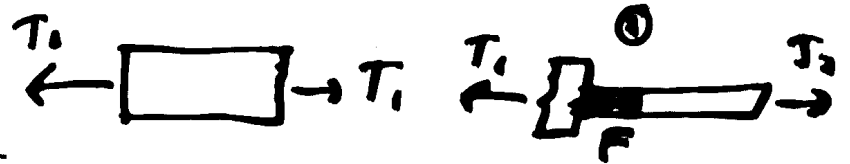
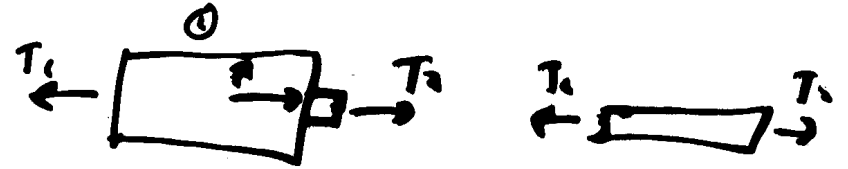
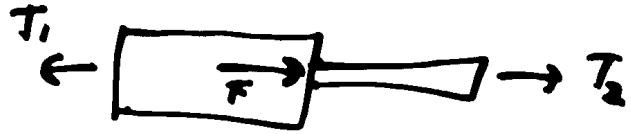
ex 2)



Geometry: $\delta_1 = -\delta_2$

Assume: $T_1 = T_2 = 0$ when $F = 0$

“No prestress”



FBDs ① or ②

$$\Rightarrow F + T_2 - T_1 = 0$$

$$\begin{matrix} \uparrow & \uparrow \\ \delta_2 \frac{E_2 A_2}{l_2} & \delta_1 \frac{E_1 A_1}{l_1} \end{matrix}$$

$$F + \delta_2 \frac{E_2 A_2}{l_2} + \delta_2 \frac{E_1 A_1}{l_1} = 0$$

$$\delta_2 = - \frac{F}{\left[\frac{E_1 A_1}{l_1} + \frac{E_2 A_2}{l_2} \right]}$$

4/7/03 (Ruina not Hai)

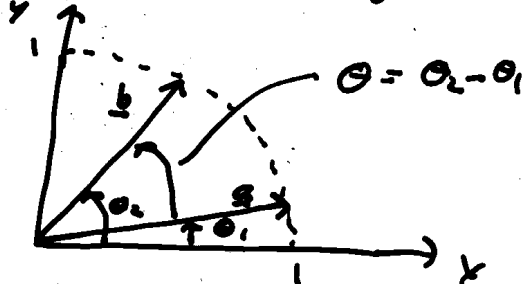
- TODAY
- 1) Trig. add. formulas
 - 2) Stress on inclined sections

TRIG. ASIDE

Consider 2 unit vectors \underline{a} & \underline{b}

$$\underline{a} = \cos \theta_1 \hat{i} + \sin \theta_1 \hat{j}$$

$$\underline{b} = \cos \theta_2 \hat{i} + \sin \theta_2 \hat{j}$$



1)

$$\underline{a} \cdot \underline{b} = \underline{a} \cdot \underline{b}$$

$$\underbrace{|\underline{a}|}_{1} \underbrace{|\underline{b}|}_{1} \cos \theta = \cos \theta_1 \cos \theta_2 + \sin \theta_1 \sin \theta_2$$

$\theta = \theta_2 - \theta_1$

$$\cos(\theta_2 - \theta_1) = \cos \theta_1 \cos \theta_2 + \sin \theta_1 \sin \theta_2$$

$$\cos(\theta_2 + \theta_1) = \cos \theta_1 \cos \theta_2 - \sin \theta_1 \sin \theta_2$$

"addition formula for cosine"

2)

$$\underline{a} \times \underline{b} = \underline{a} \times \underline{b}$$

$$\left\{ \underbrace{|\underline{a}|}_{1} \underbrace{|\underline{b}|}_{1} \sin \theta \hat{k} = (\cos \theta_1 \sin \theta_2 - \sin \theta_1 \cos \theta_2) \hat{k} \right\}$$

$$\{ \} \cdot \hat{k} =$$

$$\sin(\theta_2 - \theta_1) = \cos \theta_1 \sin \theta_2 - \sin \theta_1 \cos \theta_2$$

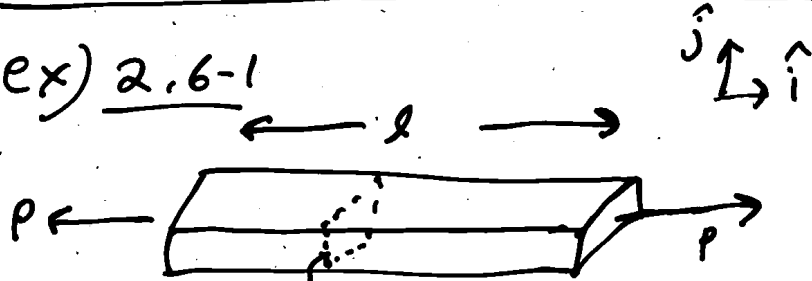
$$\sin(\theta_2 + \theta_1) = \cos \theta_1 \sin \theta_2 + \sin \theta_1 \cos \theta_2$$

"addition formula for sine"

14/7/01 (2)

Stresses on inclined sections

ex) 2.6-1



$$A_0 = (2 \text{ in}) \cdot (2 \text{ in}) = 4 \text{ in}^2$$

Material strength

$$\sigma_{ult} = 16000 \frac{\text{psi}}{\text{lb/in}^2}$$

$$\tau_{ult} = 9000 \text{ psi}$$

MAX value of P to prevent failure?

Look at tension stress first:

$$\sigma = \frac{P}{A} \leq \sigma_{ult}$$

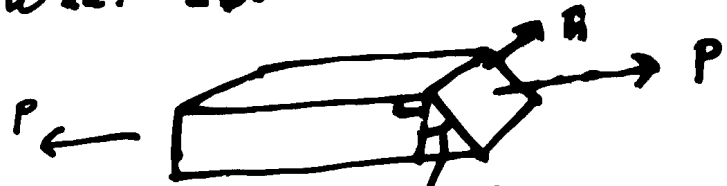
$$P \leq \sigma_{ult} A$$

⇒ To avoid failure from $\sigma_{ult} \Rightarrow P \leq \sigma_{ult} A$

$$\Rightarrow P \leq 16000 \left(\frac{\text{lb}}{\text{in}^2}\right) (4 \text{ in}^2)$$

$$P \leq 64 \times 10^3 \text{ lbf.}$$

Hey! no stress on surface w/ normal in ↑ direction. What about crooked surfaces?



$$\hat{n} = \cos \theta \hat{i} + \sin \theta \hat{j} \neq \hat{i}$$

Area of surface:



$$\frac{A_0}{A} = \cos \theta$$

$$A = \frac{A_0}{\cos \theta}$$

$$\left[\begin{array}{l} A_0 \\ \infty \end{array} \right.$$

$$\begin{array}{l} \theta = 0 \\ \theta = \pi/2 \end{array}$$

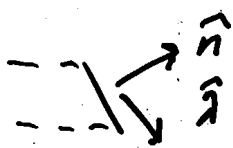
4/7/03

③

stress vector $\vec{\sigma} = \frac{P \hat{i}}{A}$

$\frac{\text{force}}{\text{area}}$ on cut surface = $\frac{P \cos \theta}{A_0} \hat{i}$

Want to look at normal & shear comps.

$\vec{\sigma} = \sigma_n \hat{n} + \tau \hat{a}$

 \hat{a} a unit vector tangent to surface

(See text for FBD approach)

$\vec{\sigma} = \vec{\sigma}$
 $\vec{\sigma} \cdot \hat{n} = \vec{\sigma} \cdot \hat{n}$
 $(\sigma_n \hat{n} + \tau \hat{a}) \cdot \hat{n} = \left(\frac{P \cos \theta}{A_0} \hat{i} \right) \cdot \hat{n}$

$\sigma_n = \frac{P \cos \theta}{A_0} \underbrace{(\cos \theta)}_{\hat{i} \cdot \hat{n}}$

$\sigma_n = \frac{P}{A_0} \cos^2 \theta$

note σ_n is max for $\theta = 0 \Rightarrow$ answer used before for tension stress

$\vec{\sigma} \cdot \hat{a} = \vec{\sigma} \cdot \hat{a}$
 $(\sigma_n \hat{n} + \tau \hat{a}) \cdot \hat{a} = \left(\frac{P \cos \theta}{A_0} \hat{i} \right) \cdot \hat{a}$

$\tau = \frac{P \cos \theta}{A_0} \frac{\hat{i} \cdot \hat{a}}{\sin \theta}$

$\tau = \frac{P}{A_0} \cos \theta \sin \theta$

4/7/03



Can we out tris. formulas

$$\sin \theta \cos \theta = \frac{1}{2} \underbrace{(\sin \theta \cos \theta + \sin \theta \cos \theta)}_{\sin(\theta + \theta)}$$

$$= \frac{\sin 2\theta}{2}$$

$$\tau = \frac{P}{A_0} \cos \theta \sin \theta = \frac{P}{A_0} \frac{\sin 2\theta}{2}$$

$$\sigma = \dots = \left(\frac{1}{2} + \frac{\cos 2\theta}{2} \right) \frac{P}{A_0}$$

↑ see text

Back to problem at hand.

Shear stress is biggest when $\frac{\sin(2\theta)}{2}$ biggest

$$\Rightarrow \theta = \pi/4 \quad (\sin(2\theta) = 1)$$

Shear biggest on surfaces at 45° from tension dir.

$$\tau_{max} = \frac{P}{2A_0}$$

$$\tau_{max} \leq \tau_{ult}$$

$$\frac{P}{2A_0} \leq 7000 \text{ PSI}$$

$$P \leq (7000 \frac{\text{lb}_f}{\text{in}^2}) 8 \text{ in}^2$$

$$\leq 72000 \frac{\text{lb}_f}{\text{in}^2}$$

2 answers

$$1) P \leq 64000 \frac{\text{lb}_f}{\text{in}^2}$$

$$2) P \leq 72000 \frac{\text{lb}_f}{\text{in}^2}$$

take smaller

$$\Rightarrow P \leq 64000 \frac{\text{lb}_f}{\text{in}^2}$$

$$\cos^2 \theta = \cos^2 \frac{\theta}{2} + \cos^2 \frac{\theta}{2}$$

$$+ \sin^2 \frac{\theta}{2} = \frac{\sin^2 \theta}{2}$$

$$= \frac{(\cos^2 \theta + \sin^2 \theta)}{2}$$

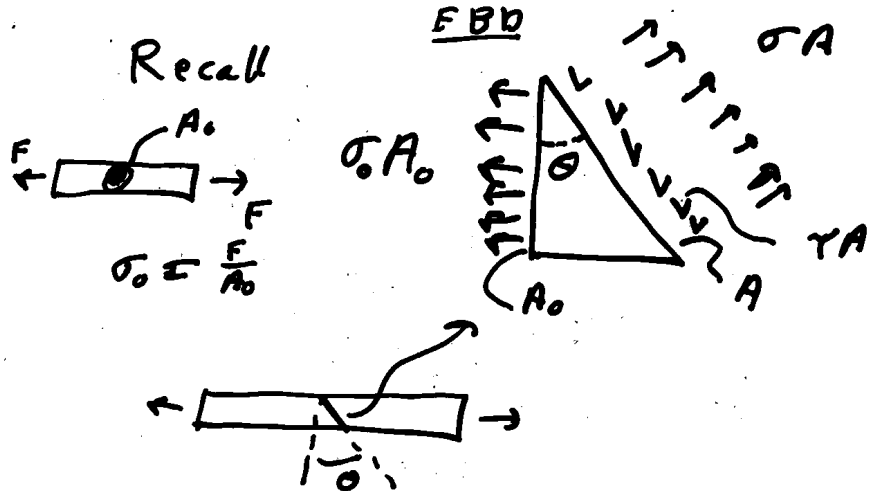
$$+ \frac{\cos^2 \theta - \sin^2 \theta}{2}$$

4/9/03

①

TODAY: a) Tension b) Torsion

TENSION (CONT'D)



$$\sum F_{\sigma \text{ dir}} = 0 \Rightarrow -\sigma_0 A_0 \cos \theta + \sigma A = 0$$

$\uparrow A \cos \theta = A_0$

$$\Rightarrow \boxed{\sigma = \sigma_0 \cos^2 \theta}$$

$$\sigma = \frac{1}{2} (1 + \cos 2\theta) \sigma_0$$

$$\sum F_{\tau \text{ dir}} = 0 \Rightarrow -\sigma_0 A_0 \sin \theta + \tau A = 0$$

$$\Rightarrow \boxed{\tau = \sigma_0 \sin \theta \cos \theta}$$

$$= \sigma_0 \sin 2\theta$$

②

Why interesting?

A. Might want to know surface w/ biggest σ ($\theta=0$) or biggest τ ($\theta=\pi/4$)

B. For a weld or glue joint σ & / or τ_{max} may be specified for a given θ .

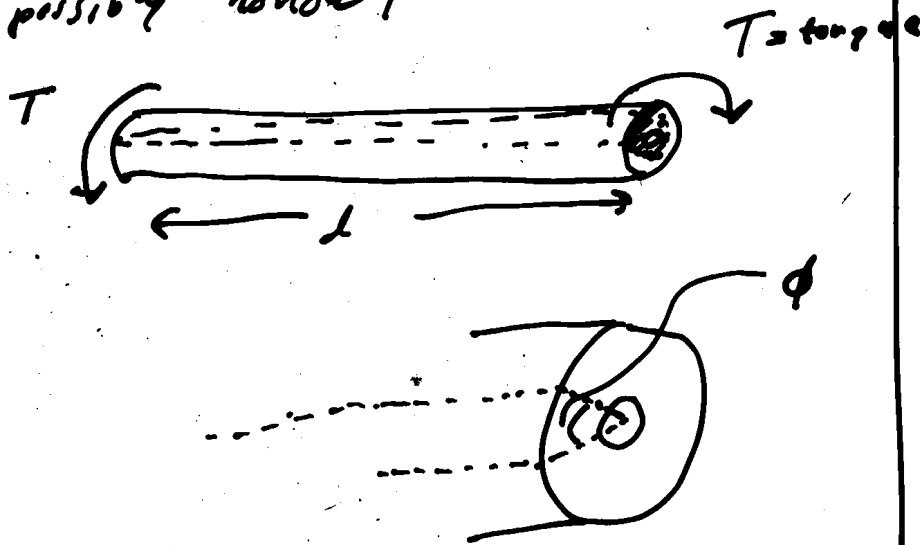


9/09/03 (2)

TORSION

WARNING!! For round bars only!! If you care about not-round bars it is incorrect to use the formulas below by just recalculating J !!!

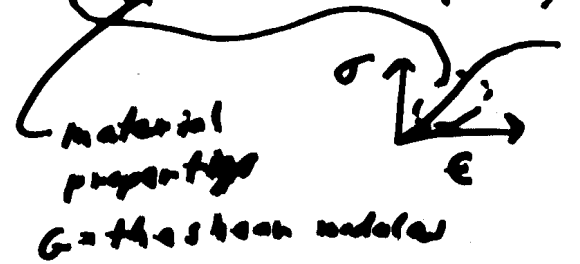
TORSION OF ROUND BARS (possibly hollow)



ϕ = net rotation of one end rel. to other

(3)

$$\phi = \frac{Tl}{G(\text{some measure of resistance})}$$



$$T = K_{torsion} \phi$$

$$K_{torsion} = \frac{G \cdot (J)}{l}$$

What about stress?

$$\gamma = \frac{T \cdot (\text{where it is})}{(\text{torque})}$$

4/11/03 (1)

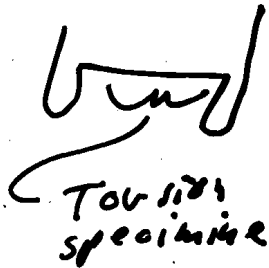
TODAY: TORSION

TORSION

ex/ paperclip

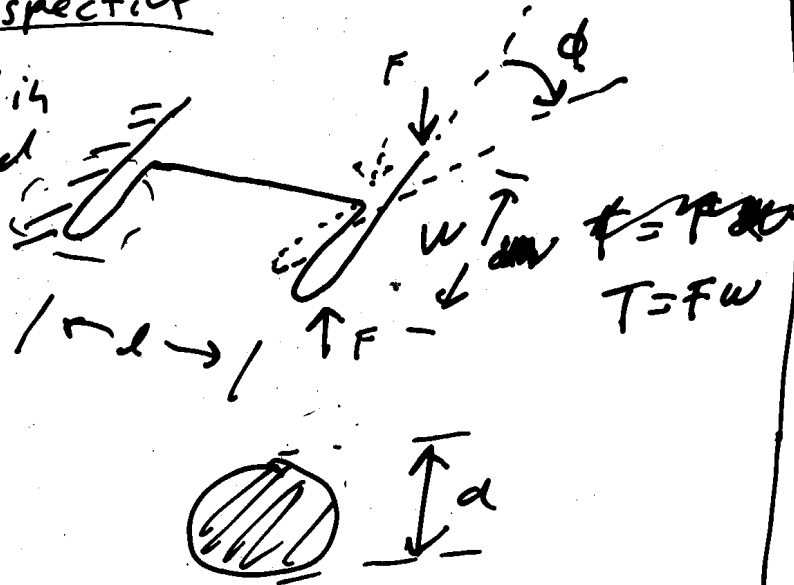


⇒
↑
plastic
bending



perspective

hold it
one hand



question:

Given $l, d, \text{material}$,
w Find F to twist a
given ϕ .

Recall from last lecture

$$T = \frac{(\text{proportional to}) \phi}{l}$$

↑
common sense reasoning

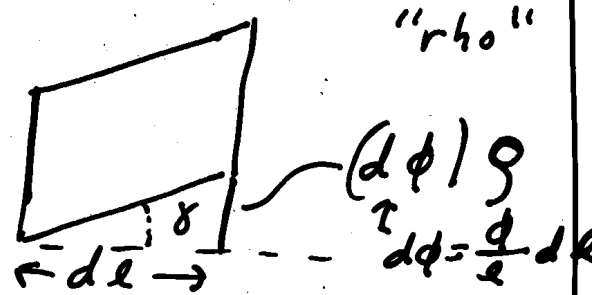
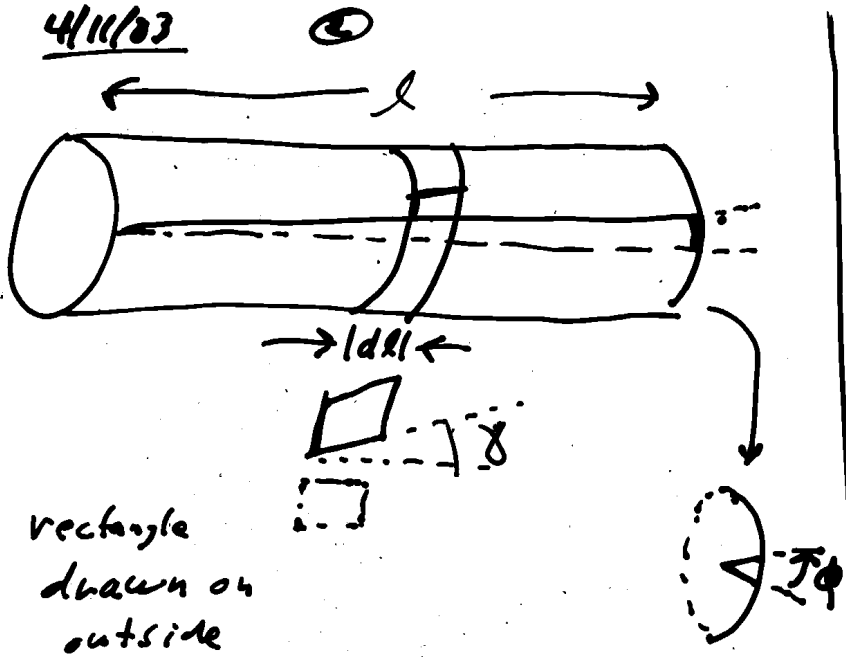
Torsion Theory

(Geometry, Mat. prop., Mechanics)

Geometry

Assume deformation
is like a bunch of stacked
pennies each rotated slightly
compared to neighbor.

4/11/03



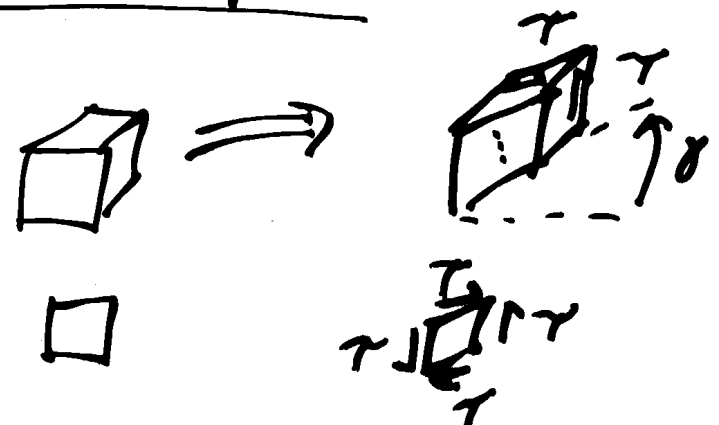
ρ = radius from centerline

$$\gamma = \frac{(d\phi)\rho}{dl} = \frac{(\phi/l)\rho}{dl}$$

↑ "gamma"

$$\gamma = \frac{\rho\phi}{l}$$

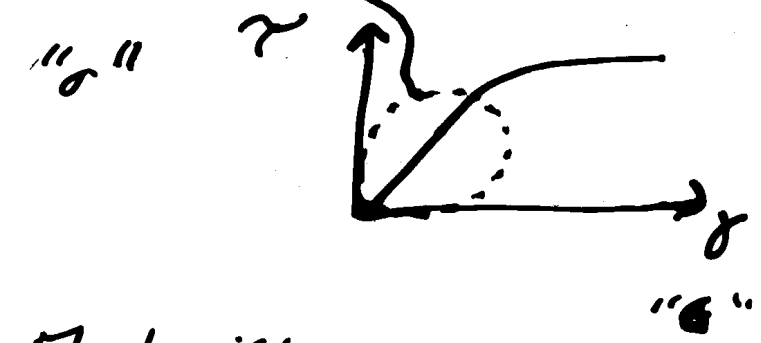
Material Properties



Linear elastic:

$$\tau = G \gamma$$

↑ Shear modulus

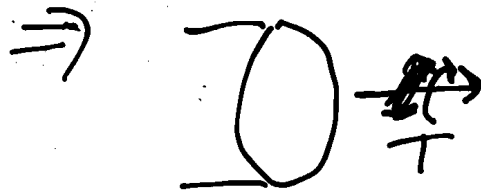
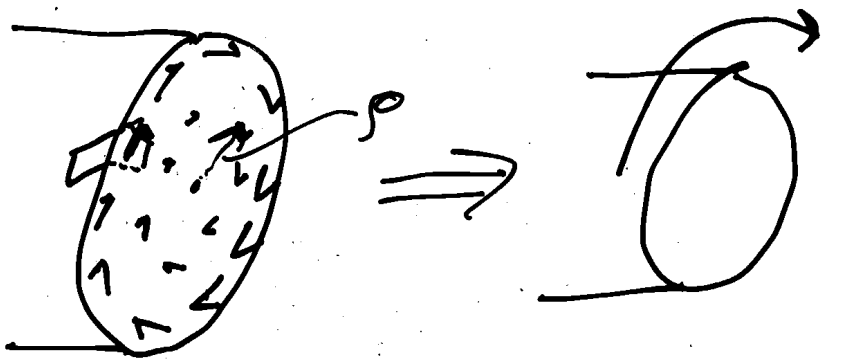


Mechanics

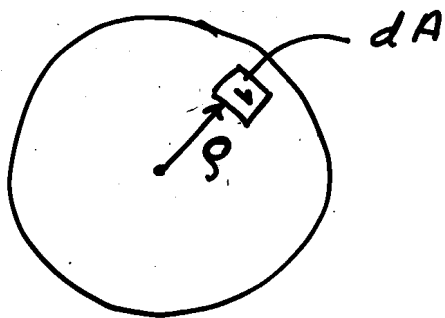
recall "Tension = σA "
 we want something like this for torsion.

41143 (3)

Look at an end cap (internal cut surface)



T for torque (not tension)



$$T = \int_{\text{cross section}} \rho \tau dA$$

(3)

All linear elastic torsion formulas come from (1), (2), (3).

Algebra & Calculus:

$$T = \int \rho \tau dA$$

$$\tau \uparrow G \delta$$

$$\delta \uparrow \frac{\phi \rho}{l}$$

$$= \int G \rho^2 \frac{\phi}{l} dA$$

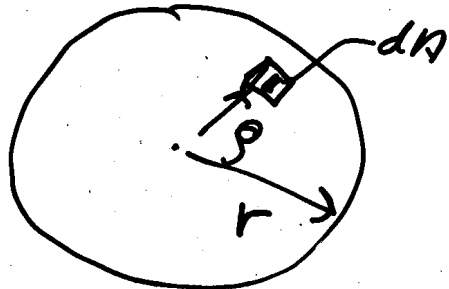
$$= \frac{G \phi}{l} \int \rho^2 dA$$

$$T = \frac{G \phi}{l} J$$

(A)

$J = \int \rho^2 dA$
= polar area moment of inertia

Theory only holds for round bars (penny stacking geometry is otherwise wrong.)



J for round bar

$$\int \rho^2 dA = \int_0^r \int_0^{2\pi} \rho^2 \underbrace{\rho d\rho d\theta}_{dA}$$

$$= 2\pi \int_0^r \rho^3 d\rho = \frac{\pi r^4}{2}$$

$$J = \frac{\pi r^4}{2}$$

4!!

twice the radius =>
16 times the torsion stiffness

Kit

What about failure

$$\gamma = \frac{\rho \phi}{L} = \frac{\rho (\tau L / G J)}{L}$$

$$= \frac{\rho \tau}{G J}$$

$$\tau = G \gamma \Rightarrow \boxed{\tau = \frac{T \rho}{J}} \text{ (B)}$$

Back to paper clip

$$T = \frac{\phi G J}{L}$$

$$W F = \frac{\phi G J}{L}$$

$$W \approx .5", L \approx 1"$$

$$G \approx 12 \times 10^6 \text{ lbf/in}^2$$

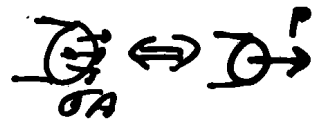
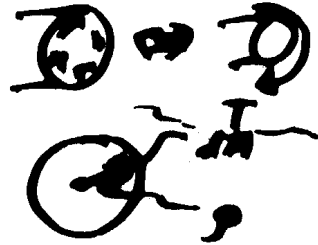
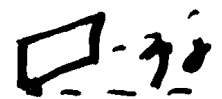
$$J = \frac{\pi}{2} \left(\frac{1}{8}\right)^4 \text{ in}^4$$

$$\phi = \pi/18$$

⇒ F

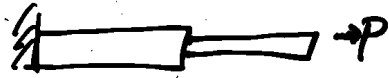
4/14/03 ①

Torsion (w/ Tension Review)

	Tension	Torsion
Mechanics	$P = \sigma A$  $[P = \int \sigma da]$	$T = \int \rho r da$ 
Constit. Law	$\sigma = E \epsilon$ $[\epsilon = \frac{\sigma}{E} + \alpha \Delta T]$	$\gamma = G \theta$ 
Geometry of Def.	$\epsilon = \frac{\delta}{L}$	$\gamma = \frac{\theta r}{L}$ (stacked pennies)
Summary of commonly used formulas	$\sigma = \frac{P}{A}, \delta = \frac{PL}{AE}$	$\gamma = \frac{\theta r}{L}, \phi = \frac{TL}{JG}$ $J = \int \rho^2 da = \frac{1}{2} \pi r^4$

4/14/03

bars in series



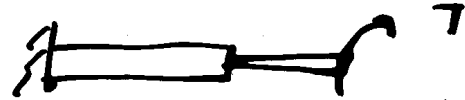
$$P_1 = P_2, \quad \delta = \delta_1 + \delta_2$$

$$\Rightarrow K_{eff} = \frac{1}{\frac{1}{K_1} + \frac{1}{K_2}}$$

$$K_1 = A_1 E_1 / l_1$$

$$K_2 =$$

$$C = C_1 + C_2$$



$$T_1 = T_2, \quad \phi = \phi_1 + \phi_2$$

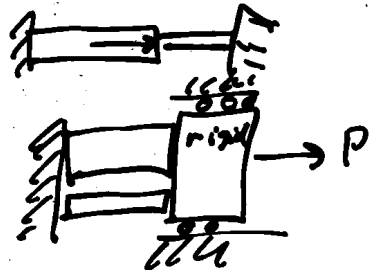
$$K_{eff} = \frac{1}{\frac{1}{K_1} + \frac{1}{K_2}}$$

$$K_1 = \frac{J_1 G}{l_1}, \quad K_2 =$$

$$C = C_1 + C_2$$

$$C_1 = \gamma K_1$$

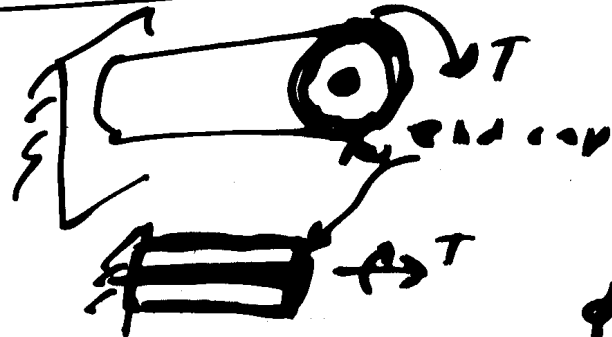
bars in parallel



$$P = P_1 + P_2, \quad \delta_1 = \delta_2$$

$$K_{eff} = K_1 + K_2$$

$$C_{eff} = \frac{1}{\frac{1}{C_1} + \frac{1}{C_2}}$$



$$T = T_1 + T_2, \quad \phi_1 = \phi_2$$

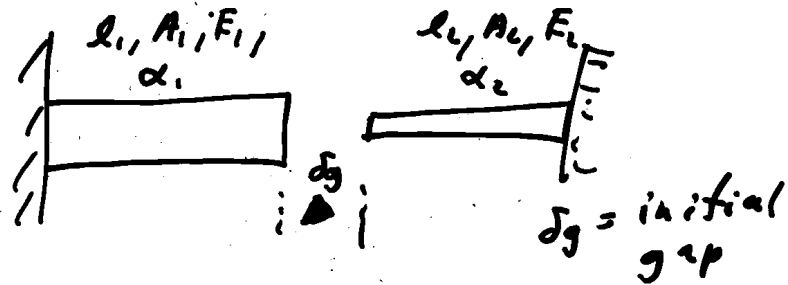
$$K_{eff} = K_1 + K_2$$

$$C_{eff} = \frac{1}{\frac{1}{C_1} + \frac{1}{C_2}}$$

4/14/03

(3)

Statically ind. problems require mixing up diff. parts of the theory (hint: be especially careful w/ signs).

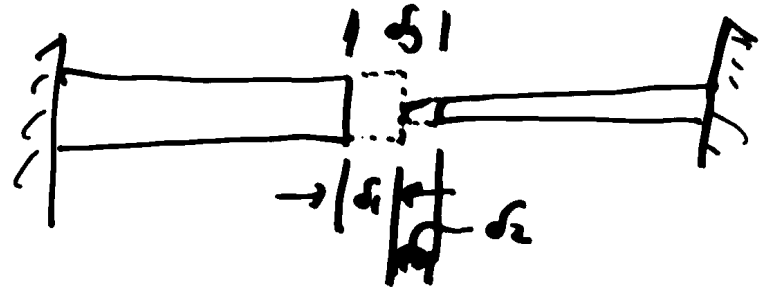


Given all geometry & properties

Given ΔT , find σ_i .

Geometry

temp. rise



δ_1, δ_2 are increase in length

[we expect that if A_1, E_1, α_1 big & ΔT big then $\delta_2 < 0$]

(1) $\delta_1 + \delta_2 = \delta_g$ [if ΔT big enough]

(2) $\delta_2 = \left[\frac{\sigma_2}{E_2} + (\Delta T) \alpha_2 \right] l_2$

$\delta_1 = \left[\frac{\sigma_1}{E_1} + (\Delta T) \alpha_1 \right] l_1$

(3) $P \leftarrow \square \rightarrow P \quad P \leftarrow \square \rightarrow P$

$P = P$

$\sigma_1 A_1 = \sigma_2 A_2$

4/14/03 (1)

Apply (3) to (2), (2) to (1)

⇒ 1 eqn. for 1 unknown σ_1

(1) $\delta_1 + \delta_2 = \delta_g$

⇒ $\left[\frac{\sigma_1}{E_1} + \Delta T \alpha_1 \right] l_1 + \left[\frac{\sigma_1 A_1}{A_2 E_2} + \Delta T \alpha_2 \right] l_2 = \delta_g$

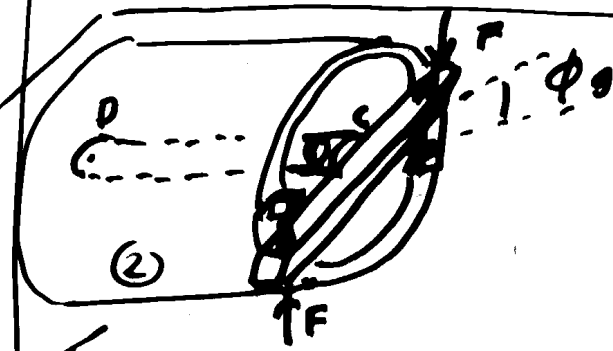
1 eqn. in 1 unknown σ_1

$\sigma_1 = \dots$

if answer is ~~not~~ $\sigma_1 > 0$
replace w/ $\sigma_1 = 0$

if answer is $\sigma_1 < 0$
leave alone $\sigma_1 =$ $\sigma_1 =$

TORSION indet. prob.



Rigid bar AB is welded to end of shaft DC. When gap δ_g is covered ($\phi_{DC} = \phi_g$) then contact made w/ outer tube

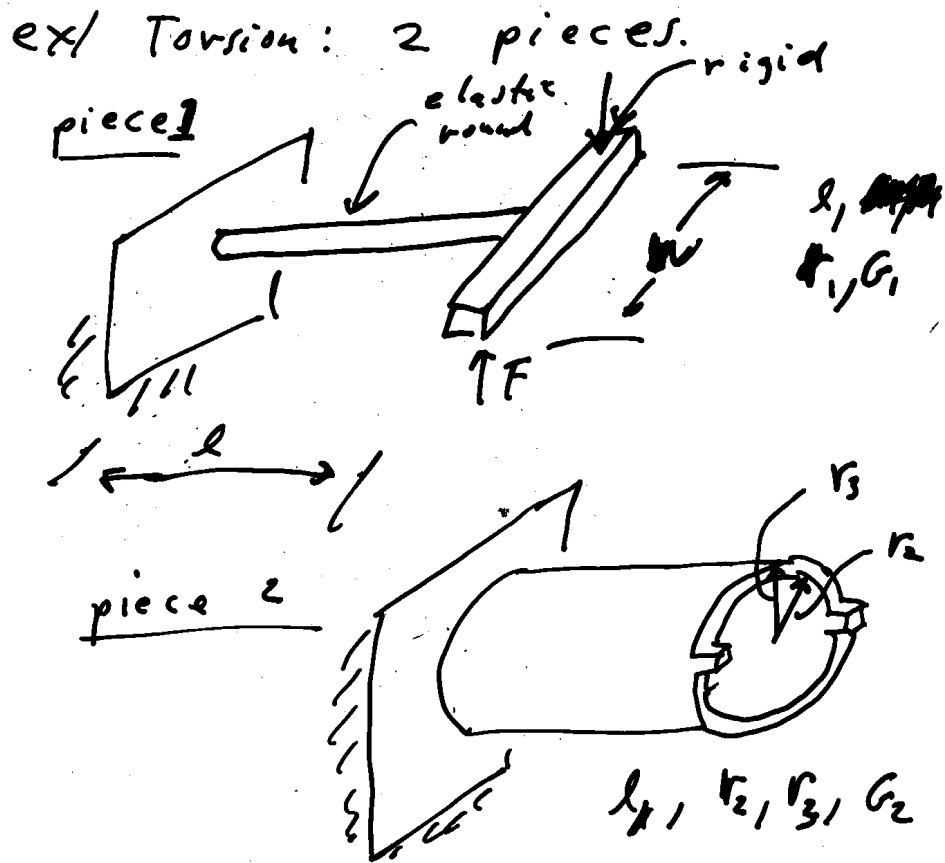


Given F & all geom & prop. Find τ in tube (2) outer wall.

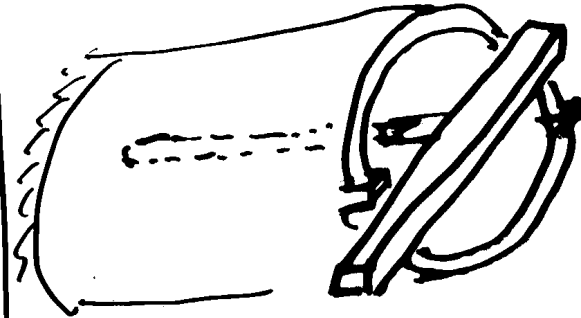
4/16/03 ①

TODAY

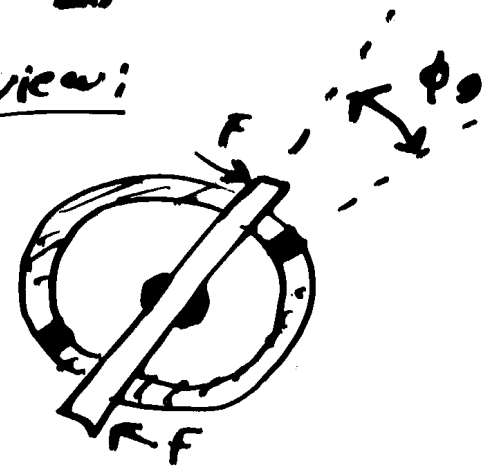
- 1) Torsion ex cont'd
- 2) More about tension & torsion
- 3) long narrow thin
- 4) intro to bending.



put together



end view:



Given all geometry & properties & F find max stress in bar 2.
 Assume F big enough to close gap:

$$\frac{(Fw)l}{\sigma, G} > \phi_0$$

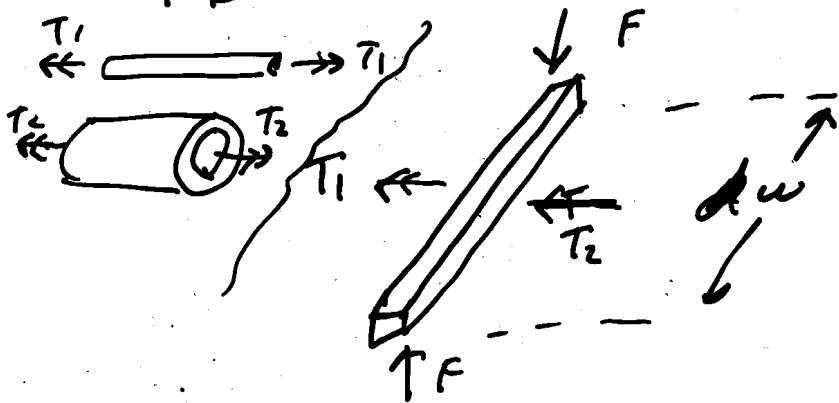
4/16/03 ②

Geometry

$$\boxed{\phi_2 = \phi_1 - \phi_g} \quad (1)$$

Mechanics

FBD of rigid bar & shafts



\$T_2\$ comes from 2 shafts

$$\boxed{-T_1 + -T_2 = F \delta u} \quad (2)$$

Note: We neglect funny effects at ends

Mat prop.

$$\boxed{T_1 = \frac{J_1 \phi_1}{L_1 G_1}, \quad T_2 = \frac{J_2 \phi_2}{L_2 G_2}} \quad (3)$$

$$L_1 = L_2$$

$$J_1 = \pi r_1^4 / 2$$

$$J_2 = \frac{\pi}{2} (r_3^4 - r_2^4)$$

$$\approx \frac{\pi}{2} ((r_3 - r_2) 2\pi r_3) r_3^2$$

$$\approx t 2\pi r^3$$

$$\propto r_3 - r_2$$

$$r = r_2 = r_3$$

① → ③ can solve for \$T_2\$

$$\boxed{\tau_{max 2} = \left| \frac{T_2 r_{max}}{J_2} \right|}$$

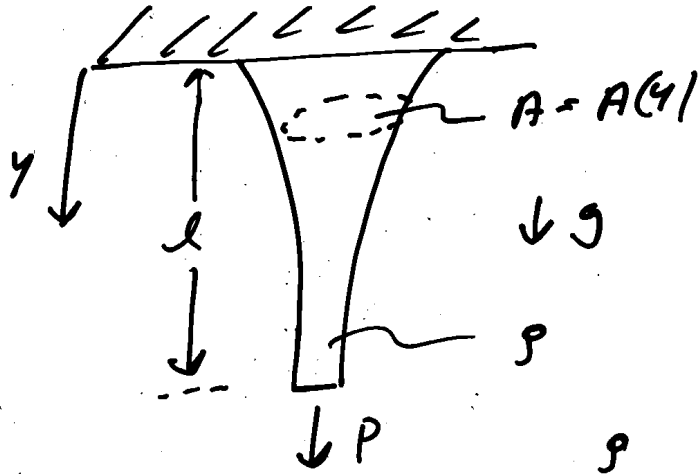
Answer

4/16/03

(3)

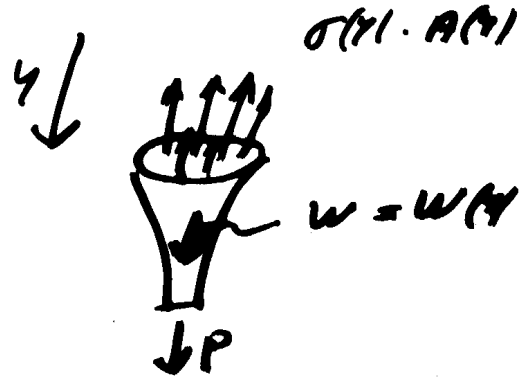
Tension P , Torsion T can vary along length, as can properties & geometry

ex)



given E , $A(y)$, l , P , what is change in length

FBD



$$W(y) = \left[\rho \int_y^l A(y') dy' \right] g$$

$$\sigma(y) = \frac{P + W(y)}{A(y)}$$

$$e(y) = \sigma(y) / E$$

$$\delta = \int \delta s = \int_0^l \frac{\delta s}{E(y) dy}$$

4/16/03 (4)

Likewise for torsion

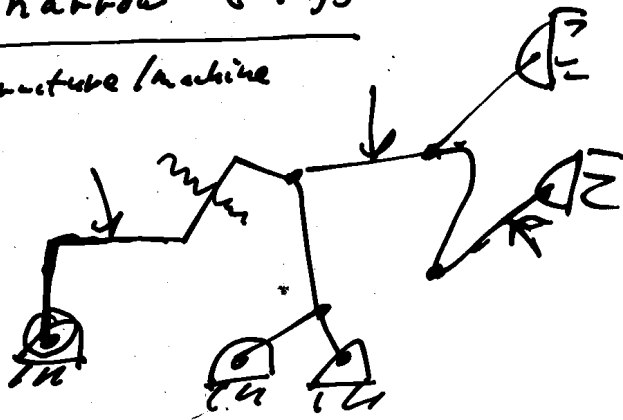
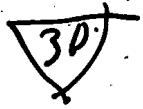
$$\left[\phi_{tot} = \int d\phi = \int_0^L \frac{T}{JG} dl' \right]$$

HW hint:

- 1) "rate of twist" meant is
ave. rate of twist = ϕ_{tot}/L
- 2) exercise your integration skills.

Long narrow things

some structure/machine



force & moment at
cut can be resolved
into

P	tension
V	shear
T	torque
M	bending Moment

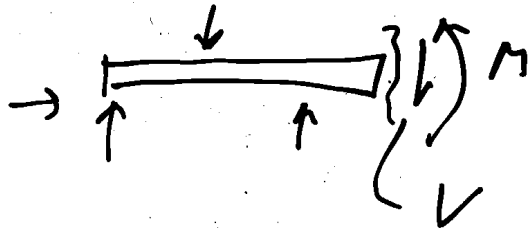
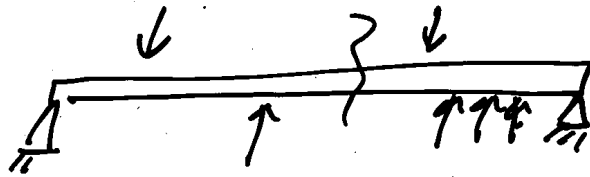
We are dealing w/ these
1 at a time.

4/16/03

⑤

Now, new topic, bending
Most important of all

ex)



Use statics to find V &

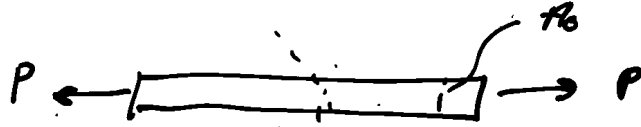
M at cut.

↳ important.

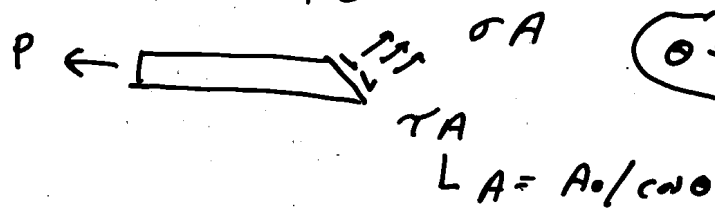
4/21/03 | ①

TODAY: 1) Q & A, 2) V, M diagrams

Q. Shear stress

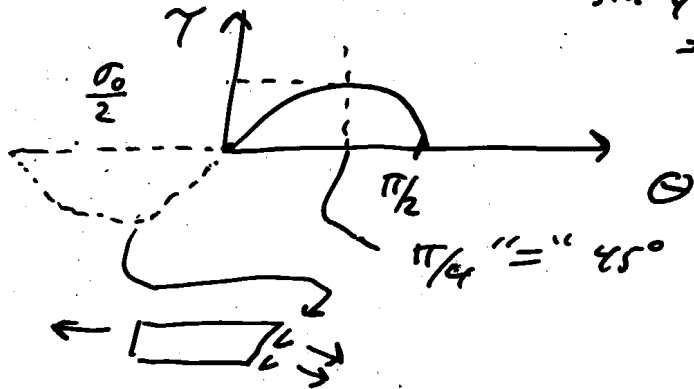


$$\sigma_0 = \frac{P}{A_0}$$



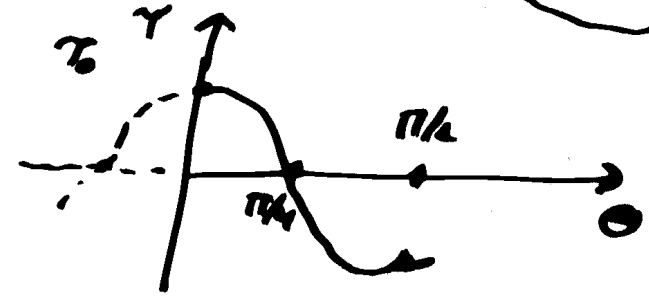
$$\tau = \sigma_0 \cos \theta \sin \theta$$

$$\sin \frac{\pi}{4} = \cos \frac{\pi}{4} = \frac{\sqrt{2}}{2}$$

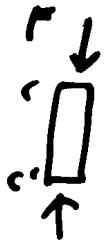


$$\tau_0 = \frac{T r}{J}$$

$$\tau = \tau_0 \cos 2\theta$$



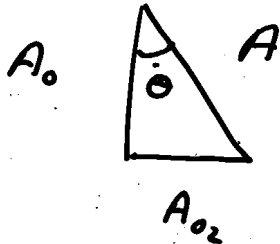
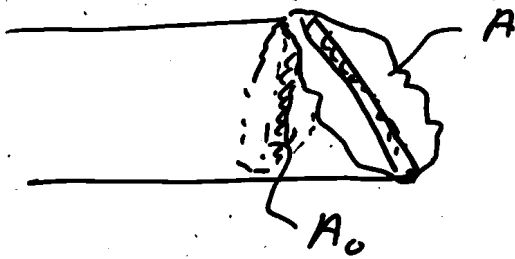
Q. Thermal Expansion



$$\delta = \alpha (\Delta T) l + \frac{P l}{A E}$$

④ Mat Props.

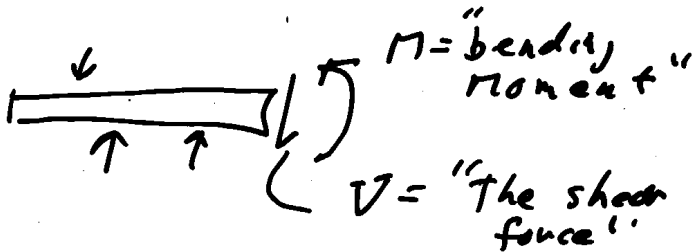
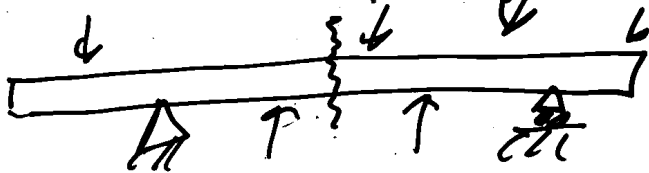
14/2/07 (1)



$$A_0/A = \cos \theta$$

$$A_{02}/A = \sin \theta$$

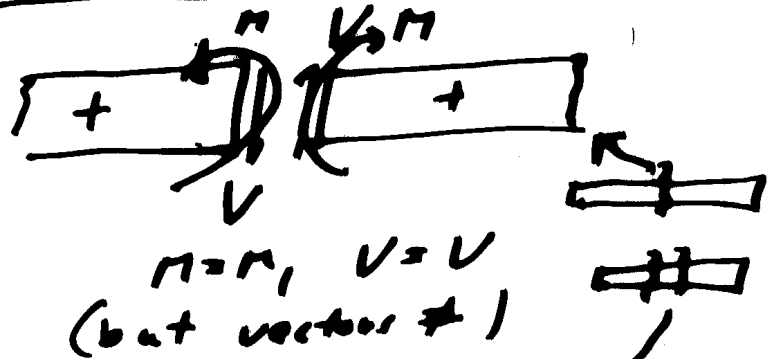
V, M diagrams



"Sign Convention" issues:

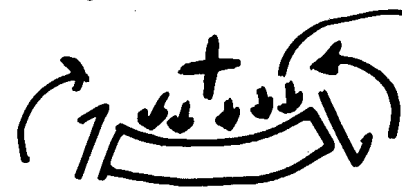
Need to have pre-defined defs. of up \uparrow & clockwise \curvearrowright . (A drawing defines this.)

Sign Convention



OR

smiling beam



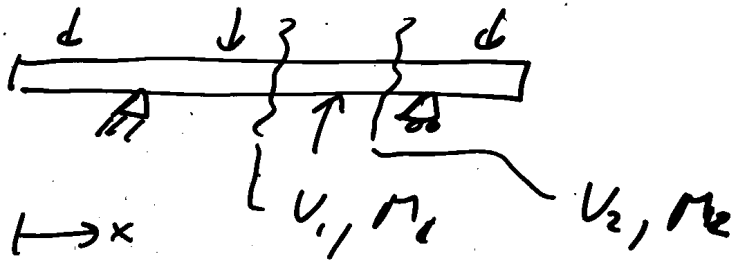
distributed load \downarrow + is down

Moment is + if causes smiling
Shear is + if causes clockwise rotation: (down on right, up on left.)

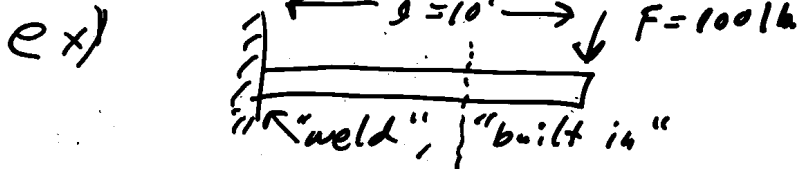
[4/23/03]

③

A common issue: How do V & M depend on pos. in beam?

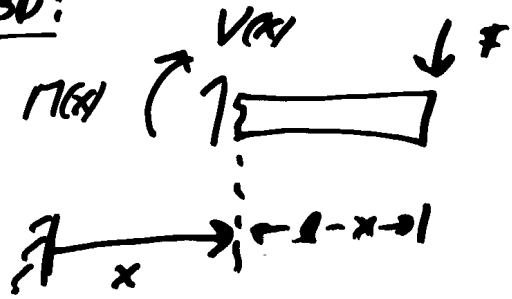


$V(x) = ?$, $M(x) = ?$



called "clamped-free" beam
or "cantilever" beam

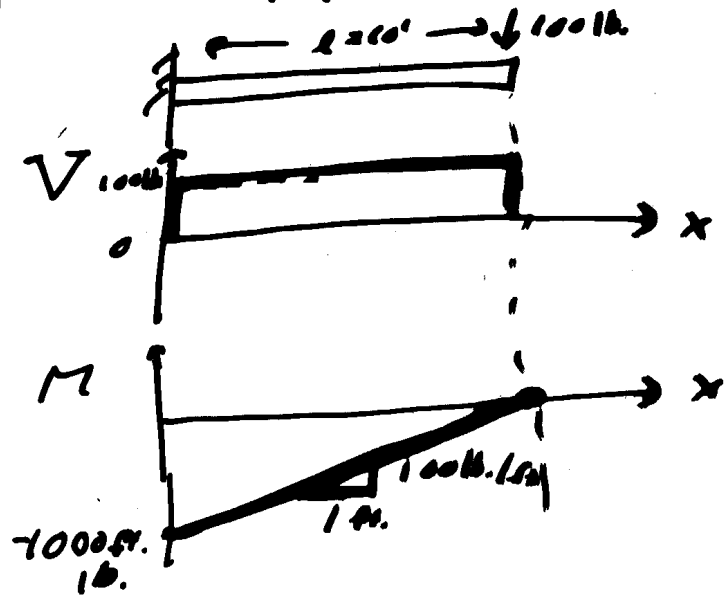
FBD:



$\Sigma F_y = 0 \Rightarrow V(x) = F = 100 \text{ lb}$
 $L \text{ const.}$

$\Sigma M_x = 0 \Rightarrow -M(x) - F(l-x) = 0$

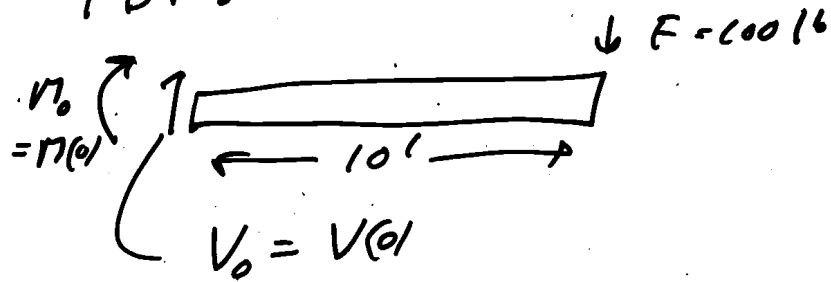
$M(x) = -100 \text{ lb}(l-x)$



4/21/03

Alternative Soln.

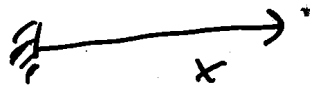
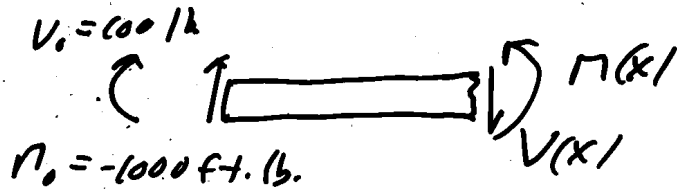
FBD of whole beam



$$\sum \tau_{10} = 0, \sum F_y = 0 \Rightarrow V_0 = 100 \text{ lb}$$

$$M_0 = -1000 \text{ ft.lb.}$$

Now make cut at x_1 , but look at left piece



$$\sum F_y = 0 \Rightarrow V(x_1) = 100 \text{ lb}$$

\leftarrow const \rightarrow

$$\sum \tau_{10} = 0 \Rightarrow$$

$$- (-1000 \text{ ft.lb}) - V(x_1) \cdot x$$

$$+ M(x_1) = 0$$

$$M(x_1) = -1000 \text{ ft.lb}$$

$$+ (100 \text{ lb}) \cdot x$$

same as before

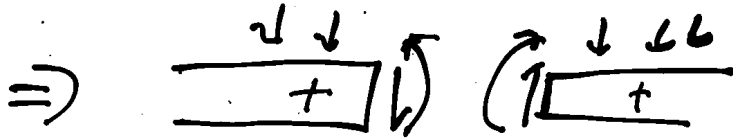
Novel: Same answer for left FBD as right FBD.

4/2/03

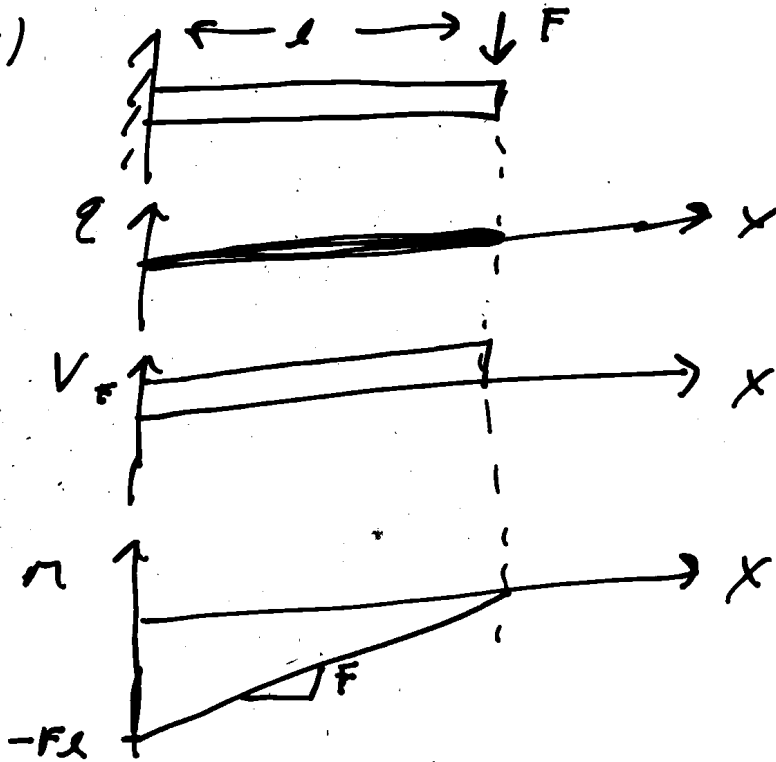
① RP section 47

TODAY: V, M diagrams cut's

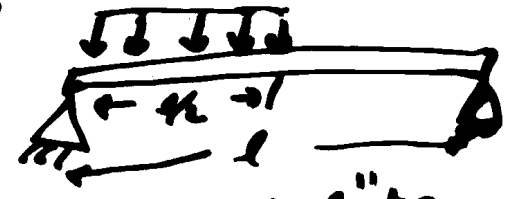
Recall



ex)

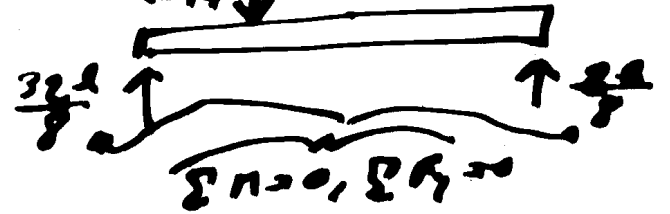


ex) 2



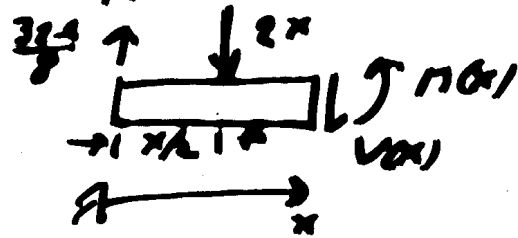
"Simply supported" beam

FBD of whole beam
cutting $2 \times \frac{l}{2}$



cut at x

first $x < l/2$



[4/27/09] (2)

$$\sum F_x = 0 \Rightarrow 0 = 0$$

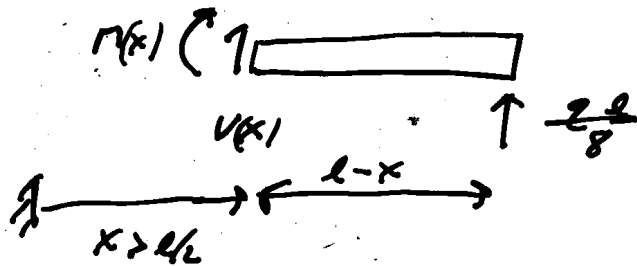
$$\sum F_y = 0 \Rightarrow \frac{32l}{8} - 2 \cdot x - V(x) = 0$$

$$V(x) = \frac{32l}{8} - 2 \cdot x \quad x \leq l/2$$

$$+\sum M_{i,x} = 0 \Rightarrow M(x) - \left(\frac{32l}{8}\right) \cdot x + (2 \cdot x) \frac{x}{2} = 0$$

$$M(x) = \frac{32l}{8} x - 2x^2/2 \quad x \leq l/2$$

Next look at $x > l/2$

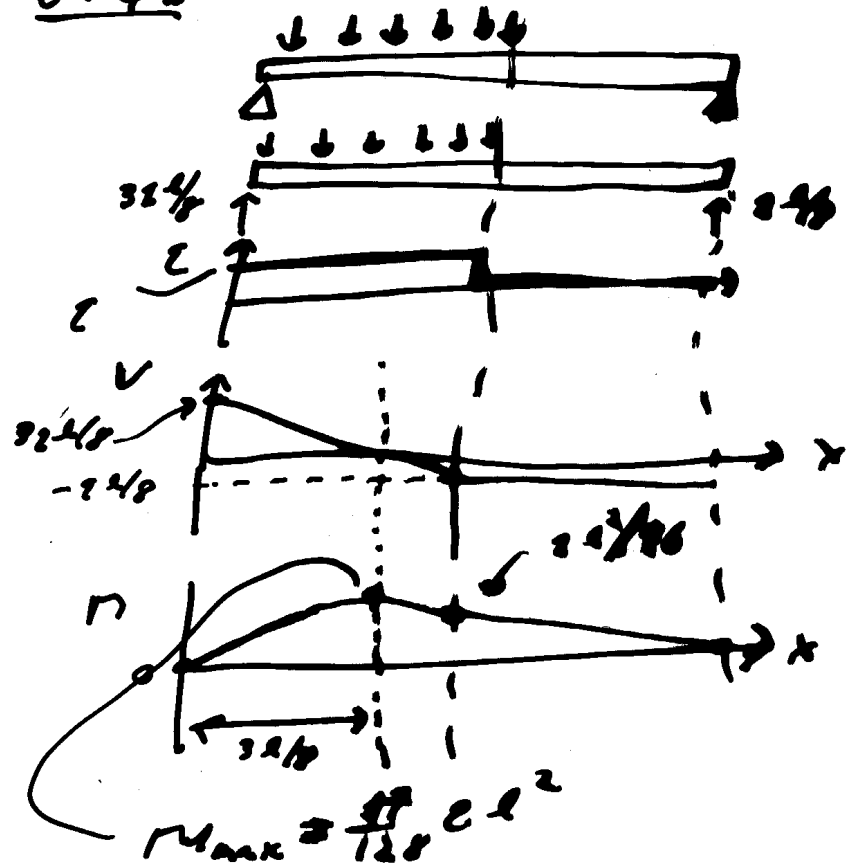


$$\sum F_y = 0 \Rightarrow V(x) = -24l \quad x \geq l/2$$

$$+\sum M_{i,x} = 0 \Rightarrow -M(x) + \frac{2l}{8}(l-x) = 0$$

$$M(x) = \frac{2l}{8}(l-x) \quad x \geq l/2$$

Graphs



[4/23/03] (3)

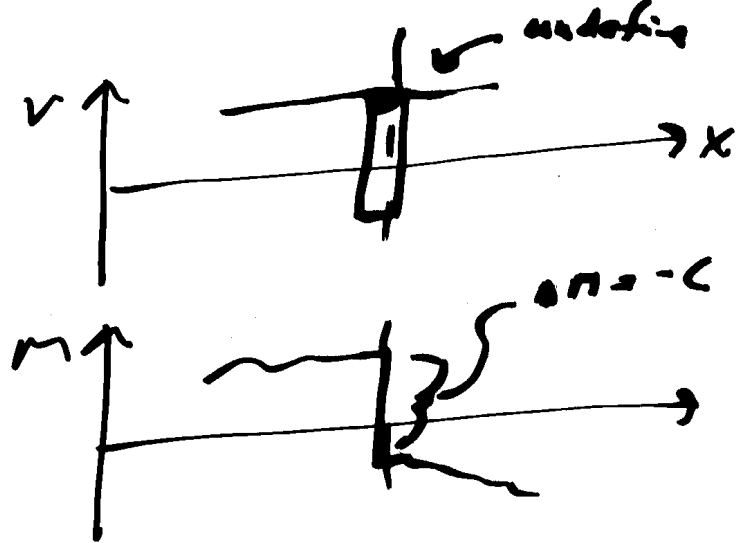
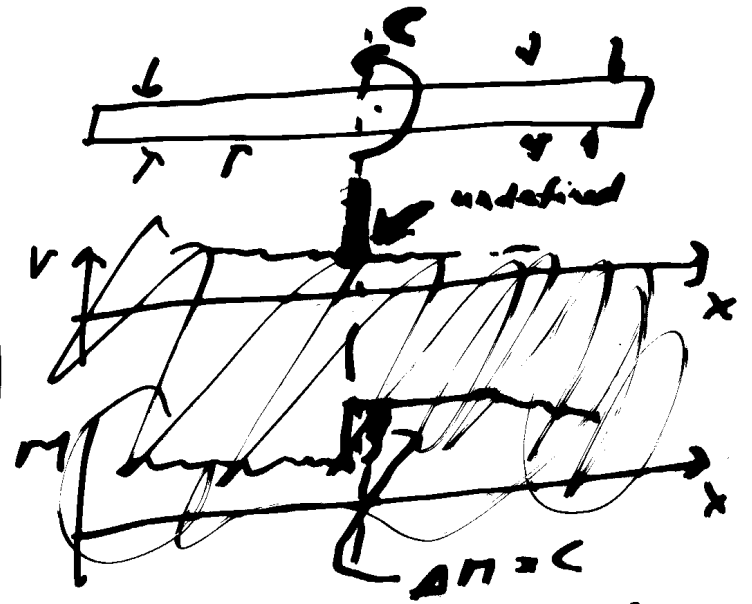
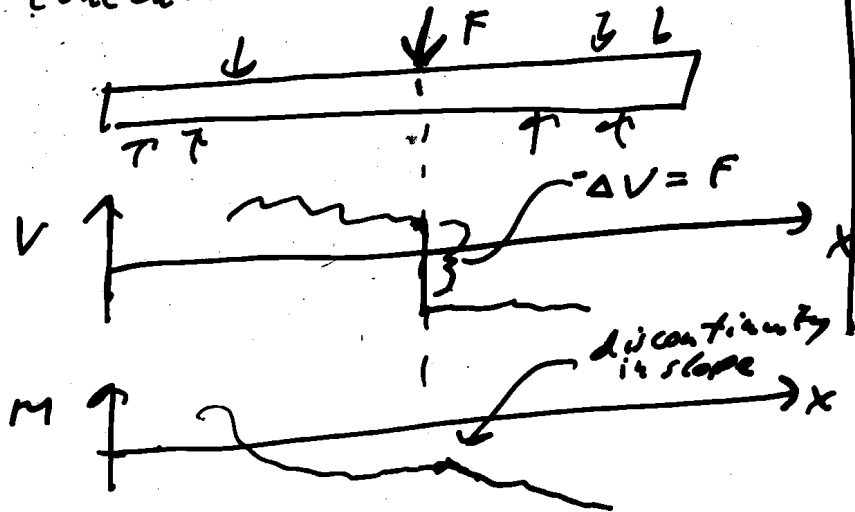
Fact: Always $\frac{dM}{dx} = V(x)$

$\frac{dV(x)}{dx} = -e(x)$

$\Rightarrow V(x) = V_0 + \int_0^x -e(x') dx'$

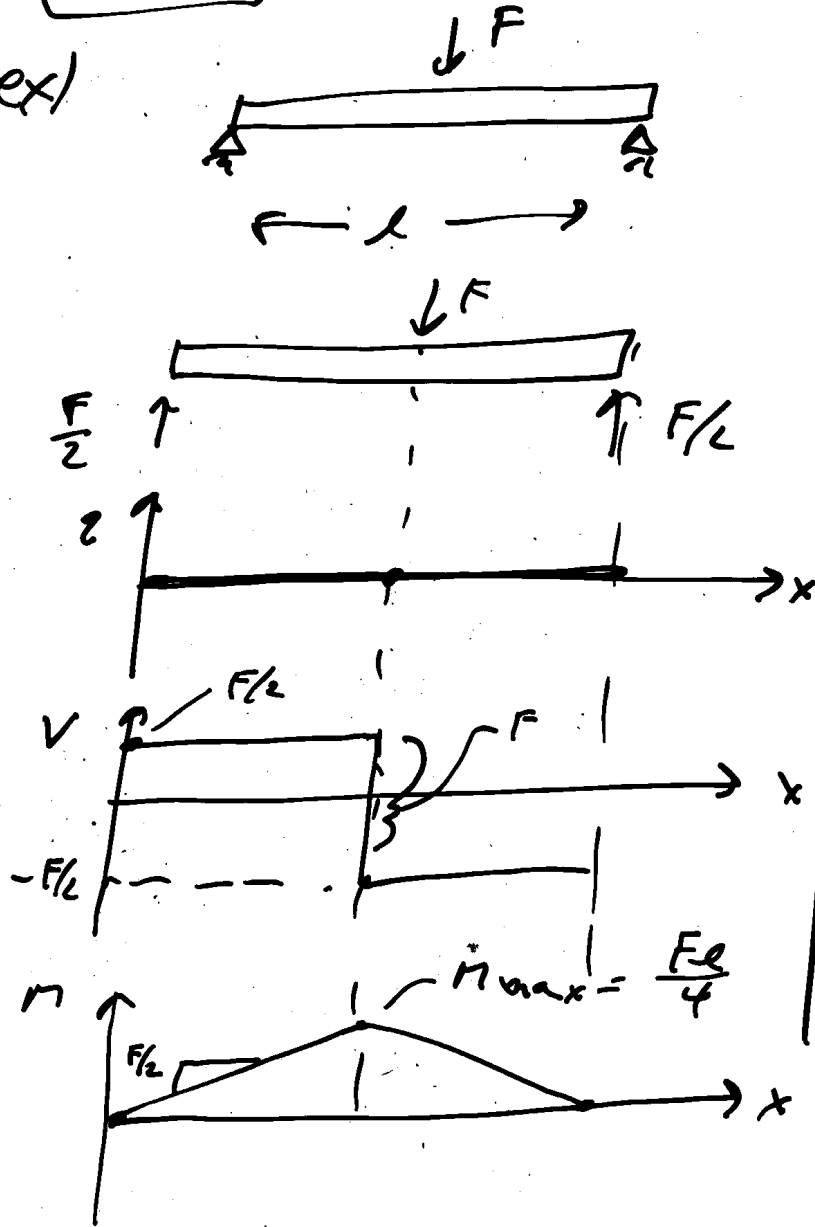
$M(x) = M_0 + \int_0^x V(x') dx'$

Need to worry about concentrated loads & couples



[4/23/23] 4

ex)



~~FP~~
Note: $\frac{dV}{dx} = -c$

$$\frac{dM}{dx} = V$$

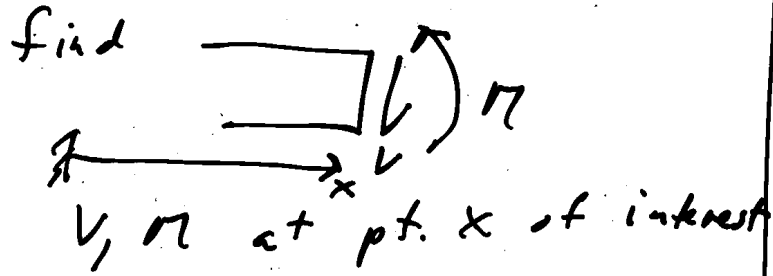
is a shortcut.

Can always use FBD instead.

14/15/03

TODAY: σ, τ in beams

Procedure: Use statics to

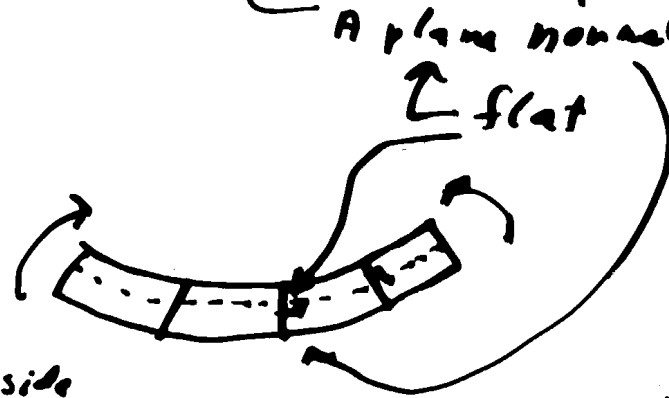


Goal: to find stresses.

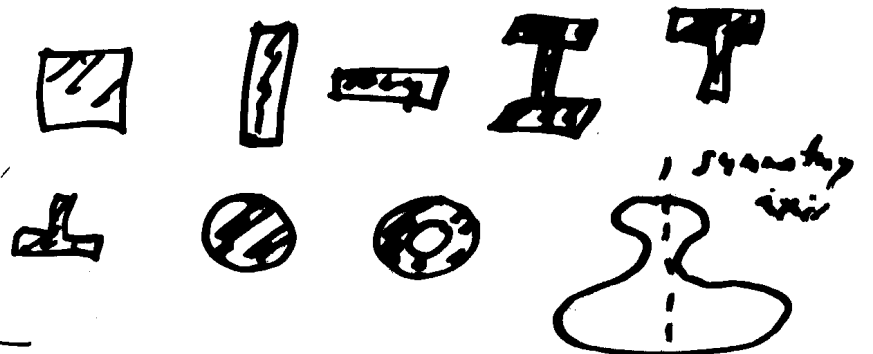
Theory of stresses in beams

Geometric assumption

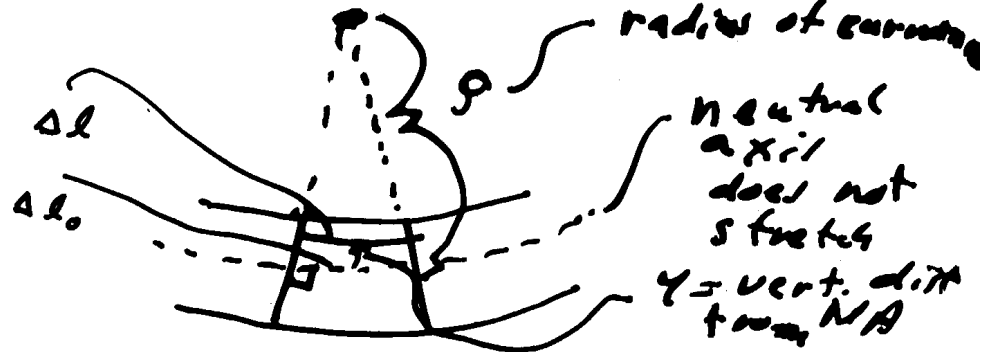
plane-normal sections remain plane & normal



aside
Cross sections of interest/reference



back to geometry



12/25/03

②

Similar Δ !

$$\frac{\Delta l}{l_0 - y} = \frac{\Delta l_0}{l_0} \Rightarrow \epsilon = \frac{\Delta l - \Delta l_0}{\Delta l_0}$$

$$\Delta l = \frac{l_0 - y}{l_0} \Delta l_0$$

$$\epsilon = \frac{\Delta l - \Delta l_0}{\Delta l_0} = \frac{\frac{l_0 - y}{l_0} \Delta l_0 - \Delta l_0}{\Delta l_0}$$

$$= \frac{l_0 - y - l_0}{l_0} = \frac{-y}{l_0}$$

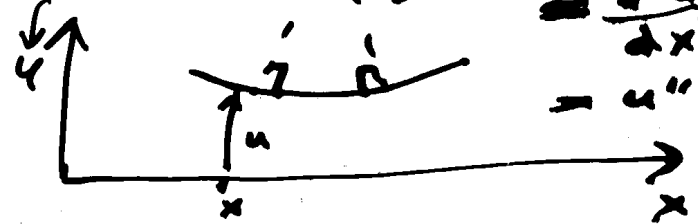
$$= 1 - \frac{y}{l_0} - 1 = -\frac{y}{l_0}$$

$$\boxed{\epsilon = -\frac{y}{l_0}}$$

$$\boxed{\epsilon = -\gamma \kappa = -\gamma u''}$$

geometry in 4d

deflection



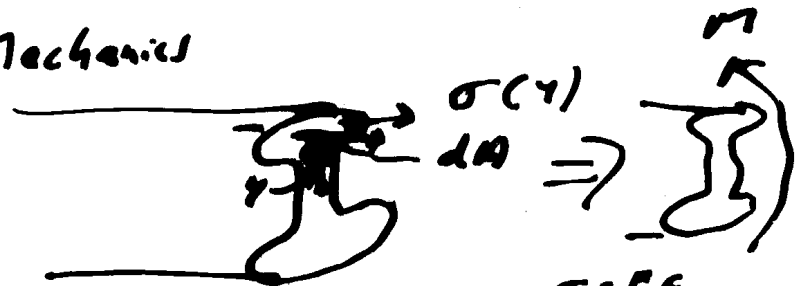
$$\kappa = \frac{1}{\rho} = \frac{d^2 y}{dx^2} = u''$$

Mat. Prop.

$$\boxed{\sigma = E \epsilon}$$

linear elastic

Mechanics



$$M = - \int y \sigma dA \quad \begin{matrix} \sigma = E \epsilon \\ \epsilon = -\gamma \kappa \end{matrix}$$

cross section

$$= - \int y E (\gamma \kappa) dA = \kappa E \int y^2 dA = I \kappa$$

$$\kappa = u''$$

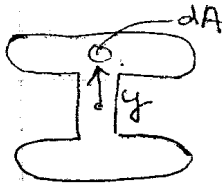
$$\boxed{M = EI u''}$$

Courtesy of:
Julia Ferullo

Geometry: $\epsilon = -y/\rho \rightarrow \frac{1}{\rho} = \kappa = u'' = \frac{d^2 u}{dx^2}$

Mat. Props: $\sigma = E\epsilon$; Mechanics: $M = EI/\rho = EI\kappa = EIu''$

$I =$ area moment of inertia $= \int y^2 dA$
cross section



Location of neutral axis: Net tension = 0 $\Rightarrow \int \sigma dA = 0$
 $\Rightarrow \int y dA = 0 \Rightarrow$ Neutral axis at centroid

of interest is stress: $\sigma = E\epsilon = E(-y\kappa)$, $\sigma = \frac{-My}{I}$

$\sigma_{max} = \left| \frac{My|_{max}}{I} \right| = \frac{M}{S}$ $S =$ geometric quantity = section modulus $= \frac{I}{|y|_{max}}$

Example: What is the yield stress of Andy's ruler?
bend ruler to radius of 2 in (1/50 in thick) to get permanent Δ .

$\rho = 2$ in, $t = 1/50$ in, $y =$ yield strain $= \left| \frac{y_{max}}{\rho} \right| = \frac{|1/50|/2}{2 \text{ in}}$ $E_y = 200$

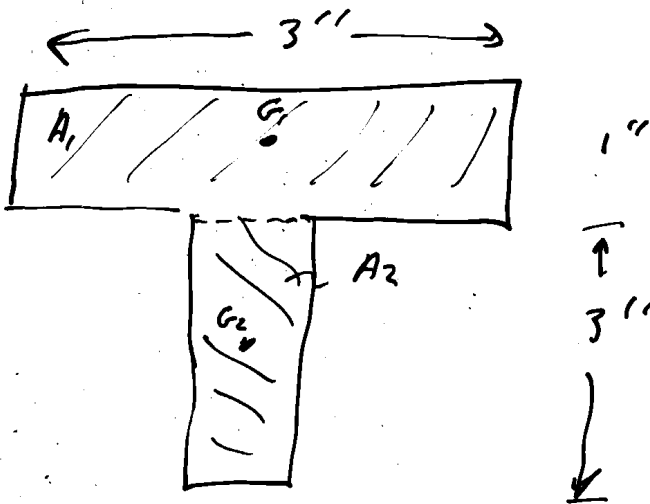
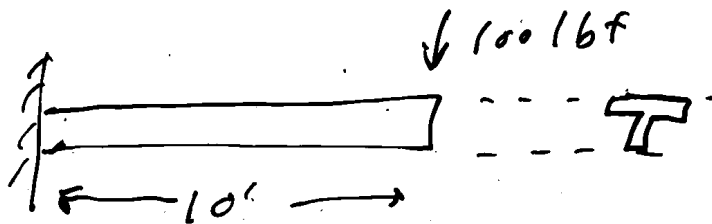
$\sigma_y =$ yield stress $= E \epsilon_y$

$\sigma_y = 10^9 \text{ Pa}$ or $30 \times 10^6 \text{ lbf/in}^2$ } for all steel $= 200 \times 10^9 \text{ Pa}$
 $= 150,000 \text{ lb/in}^2$ } hard steel

Tedious Skill: Calculating I for various cross sections

4/28/03

TODAY: I, deflection



$\uparrow z$

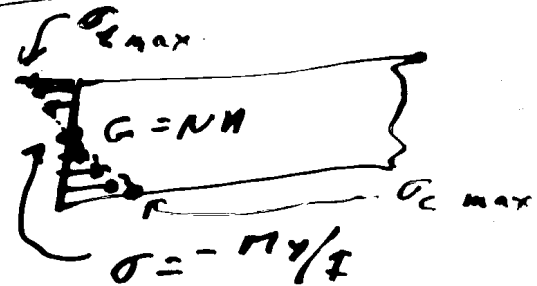
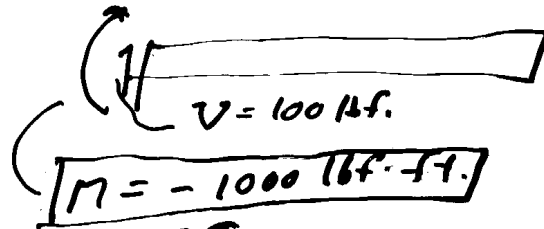
Question

$\sigma_{\text{tension max}} = ?$

$\sigma_{\text{comp max}} = ?$

First find M

Max M at left end
 $\downarrow 100 \text{ lbf}$



Need to find I , y values
 (find $G = N.A.$)

	Area	z_i	I_i
①	3 in^2	3.5 in	$\frac{1}{4} \text{ in}^4$
②	3 in^2	1.5 in	$\frac{9}{4} \text{ in}^4$

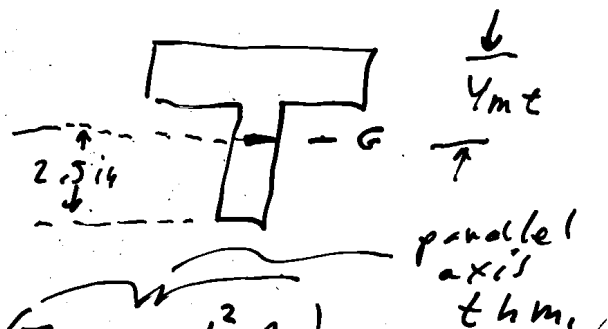
} = $\frac{1}{12} b h^3$

Area = area of region
 z_i = vert. loc. of centroid of area
 I_i = area moment of inertia about centroid

4/29/07 (2)

Where is G ? Z value of centroid of composite.

$$Z_G = \frac{\sum A_i z_i}{\sum A_i} = \frac{3i^2 [3.5i + 1.5i]}{(3+3)i^2}$$
$$= \frac{15}{6} i = 2.5 i.$$



$$I = \sum (I_i + d_i^2 A_i)$$

$\uparrow |z_i - z_G|$

dist. from total G to the individual G_i .

$$= \left[\left(\frac{1}{4} + 1^2 \cdot 3 \right) + \left(\frac{9}{4} + 1^2 \cdot 3 \right) \right] i^4$$

$$I = \frac{17}{2} i^4$$

For tension $|y_{max}| = 1.5 i$
for comp. $|y_{max}| = 2.5 i.$

$$\sigma_{tmax} = \left| \frac{M_{max} y_{max,t}}{I} \right|$$

= arithmetic

$$|\sigma_{cmax}| = \left| \frac{M_{max} y_{max,c}}{I} \right|$$

= arith. $\rightarrow |\sigma_{cmax}|$

because bottom surface is further from G than top surface

[4/28/03] (3)

What is shear stress in beam?

$$\bar{\tau} = \text{ave. shear stress}$$

$$= \frac{V}{A} = \int \tau dA$$

not const. in cross section



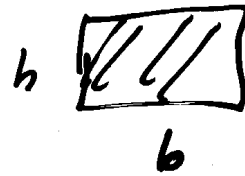
varies according to which is correct in lab but not HW or lecture

$$\approx \frac{VQ}{It}$$

This formula not on final exam.

(4)

ex)



$$I = \frac{1}{12} b h^3$$

ex)



$$I = \frac{\pi}{4} r^4$$

$$= \frac{1}{2} J$$

Deflection in beams

$$\kappa = \frac{M}{EI}$$

$$\Rightarrow \kappa = \frac{M}{EI}$$

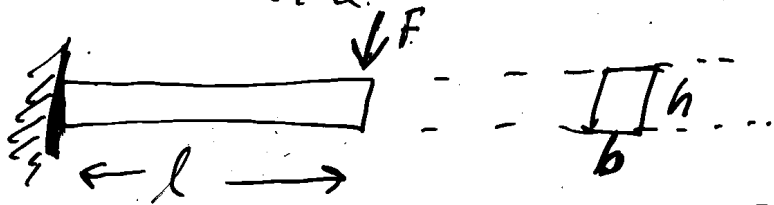
$$\frac{d^2 u}{dx^2} = \frac{M}{EI}$$

$$u'' = M/EI$$



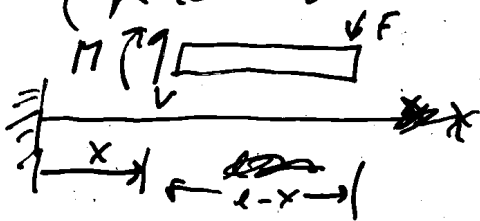
15/02/03 | ①

TODAY (1) Deflection of beams
 2) something you've always wanted.



deflection at end = ?

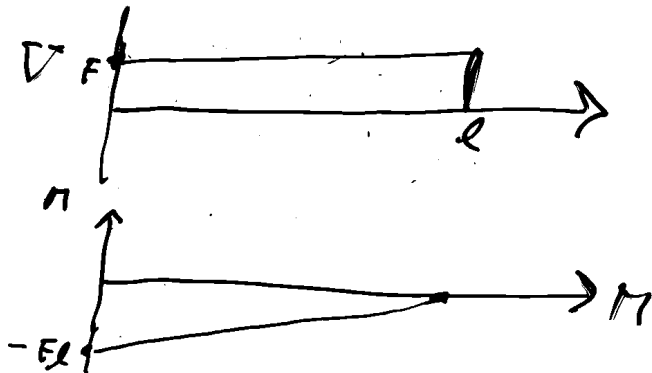
(know all moduli)



$$\sum F_y = 0 \Rightarrow V = F$$

$$+\circlearrowleft \sum M_x = 0 \Rightarrow -F(l-x) - M = 0$$

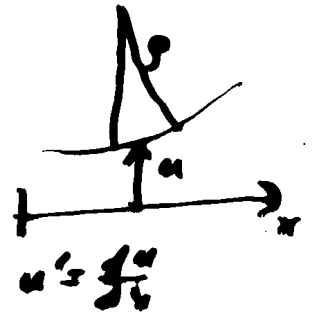
$$M = -F(l-x)$$



$$\int \frac{1}{\rho} = \frac{M}{EI}$$

$$\frac{h}{\rho} = \frac{M}{EI}$$

$$\boxed{u'' = \frac{M}{EI}}$$



$$u'' = \frac{1}{EI} (Fx - Fl)$$

$$u' = \int_0^x u''(x') dx' + u(0)$$

$$= \frac{1}{EI} \left(\frac{Fx^2}{2} - Flx \right) + 0$$

B.C.: slope at left end is zero

$$u(x) = \int_0^x u'(x') dx' + u(0)$$

$$= \frac{1}{EI} \left(\frac{Fx^3}{6} - \frac{Flx^2}{2} \right) + 0$$

B.C.: value of u at left end = 0

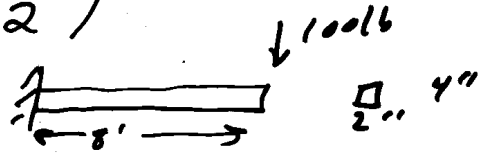
15/2/03

②

Deflection at end = $u(l)$.

$$u(l) = \frac{F}{EI} \left(\frac{l^3}{6} - \frac{l^3}{2} \right)$$

$$u(l) = \frac{-F l^3}{3EI}$$



ex) $l = 8'$, $b = 2''$, $h = 4''$
 $F = 100 \text{ lb}$ Real, as opposed to "Nominal", 2x4.
 $E = 1,000,000 \text{ lb/in}^2$

$$I = \frac{1}{12} b h^3 = \frac{1}{12} 2 \cdot 4^3 \text{ in}^4$$

$$= \frac{32}{3} \text{ in}^4$$

$$u(l) = \frac{(100 \text{ lb}) (8 \text{ ft})^3}{3 \cdot (10^6 \frac{\text{lb}}{\text{in}^2}) \cdot (\frac{32}{3} \text{ in}^4)}$$

$$= \frac{-100 \cdot 8^3}{3 \cdot 10^6 \cdot \frac{32}{3}} \frac{\text{ft}^3}{\text{in}^2} \left(\frac{12 \text{ in}}{1 \text{ ft}} \right)^2$$

$$= \frac{-100 \cdot 2^9 \cdot 3^3 \cdot 2^6 \text{ in}}{10^6 \cdot 2^5}$$

$$= -10^{-4} \cdot 2^{10} \cdot 3^3 \text{ in}$$

$2^{10} \approx 10^3$
 $3^2 \approx 10$

$$u = -10^{-4} \cdot 10^3 \cdot 10 \cdot 3 \text{ in}$$

$$= -3 \text{ in}$$

ex) same 2x4 sideways



Key variable is I
more floppy by factor
of $I_{\text{new}}/I_{\text{old}}$.

$$= \frac{0.4 \cdot 2^3 / 12}{2 \cdot 4^3 / 12}$$

$$= \frac{2^7}{2^7}$$

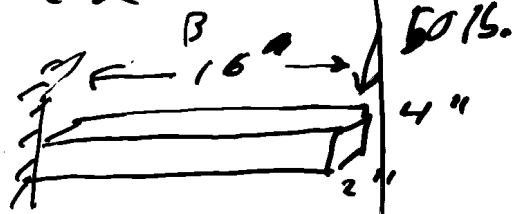
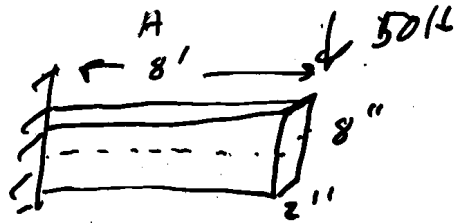
$$= \frac{1}{4}$$

\Rightarrow 4 times floppier.

5/02/07

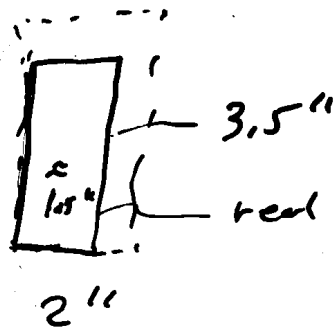
③

How much bigger is deflection of case B than case A.



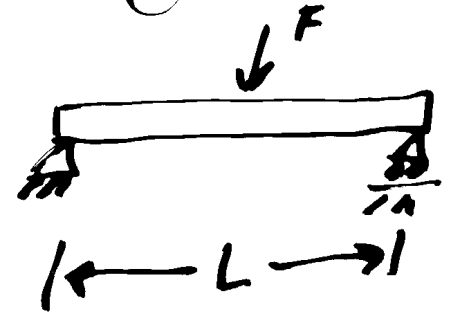
Aside: nominal 2x4

4"



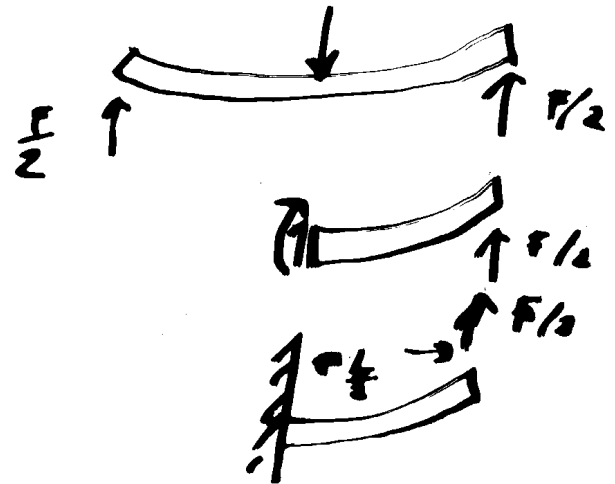
④

ex)



deflection in middle

Trick: symmetric



5/02/07

④

Cantilever $u(l) = \frac{F l^3}{3 EI}$

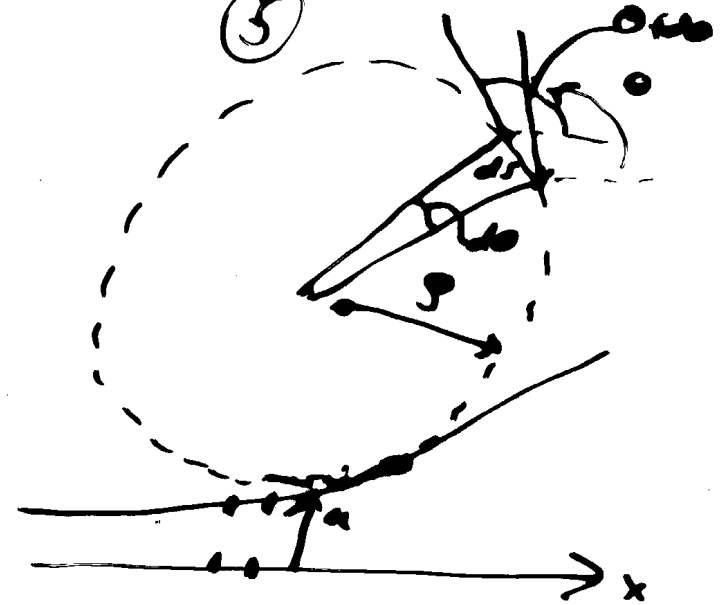
simply supported load in middle : $u\left(\frac{l}{2}\right) = \frac{\left(\frac{F}{2}\right)\left(\frac{l}{2}\right)^3}{3 EI}$
 $= \frac{F l^3}{48 EI}$

Q & A

big formula

$$\kappa = \frac{d^2 u / dx^2}{\sqrt{1 + \left(\frac{du}{dx}\right)^2}}$$

⑤



small slope

$ds \approx dx$

$\theta \approx \tan \theta = u'$

circle: $\rho d\theta = ds$

$\frac{1}{\rho} = \frac{d\theta}{ds}$

$\frac{1}{\rho} = \frac{d}{dx} \left(\frac{du}{dx} \right) = u''$