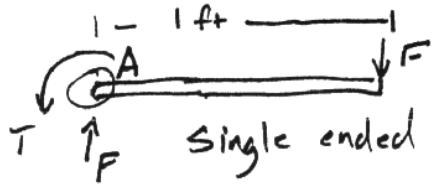


Due 11/17/99

①

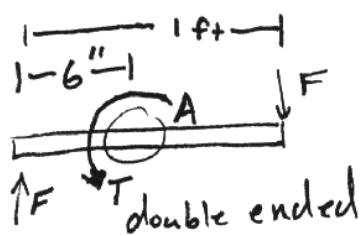
6-21

How many cycles of tightening before fatigue failure for (a) a single ended wrench and (b) a double ended wrench



$$T_{\max} = 100 \text{ ft-lbs} \quad d = 0.625''$$

$$T_{\min} = 0 \text{ ft-lbs} \quad S_{ut} = 60 \text{ ksi}$$



Max bending stress occurs near the point A in both cases.

### Case (a)

$$M = F L = T = 100 \text{ ft-lbs}$$

$$I = \frac{\pi d^4}{64} = 0.00749 \text{ in}^4$$

$$\text{Alternating: } M_a = \frac{T_{\max} - T_{\min}}{2} = 600 \text{ in-lbs}$$

$$M_m = \frac{T_{\max} + T_{\min}}{2} = 600 \text{ in-lbs}$$

$$\sigma_{x_a} = \frac{M_a C}{I} = 25,033 \text{ ksi}$$

$$\sigma_{x_m} = \frac{M_m C}{I} = 25,033 \text{ ksi}$$

No other stresses in the arm so

$$\sigma_I = \sigma_X \quad \sigma_{II} = 0 \quad \sigma_{III} = 0$$

Also, von Mises effective stress is

just:

$$\sigma_a' = \sigma_{xc} \quad \sigma_m' = \sigma_{xm}$$

Assume Case 3 loading and solve for the fatigue strength at which the wrench will fail  
(safety factor = 1)

$$N_f = 1 = \frac{S_f S_{ue}}{\sigma_a' S_{ut} + \sigma_m' S_f} \Rightarrow S_f = \frac{\sigma_a' S_{ut}}{S_{ut} - \sigma_m'}$$

Solving:  $S_f = 42.954 \text{ ksi}$

Calculate the unmodified endurance limit:

$$S_e' = 0.5 S_{ut} \rightarrow S_e' = 30 \text{ ksi}$$

Modifying factors:

$$C_{load} = 1$$

$$C_{surf} = A \left( \frac{S_{ut}}{\text{ksi}} \right)^b$$

$A = 39.9$
$b = -0.995$
as forged

$$C_{temp} = 1$$

$$= 0.679$$

$$C_{reliab} = 1 \quad (50 \text{ yr})$$

$$C_{size} = 0.869 \left( \frac{\text{degrev}}{\text{in}} \right)^{-0.097}$$

$$= 1.002 \approx 1$$

$$S_e = C_{load} C_{size} C_{surf} C_{temp} C_{reliab} S_e'$$

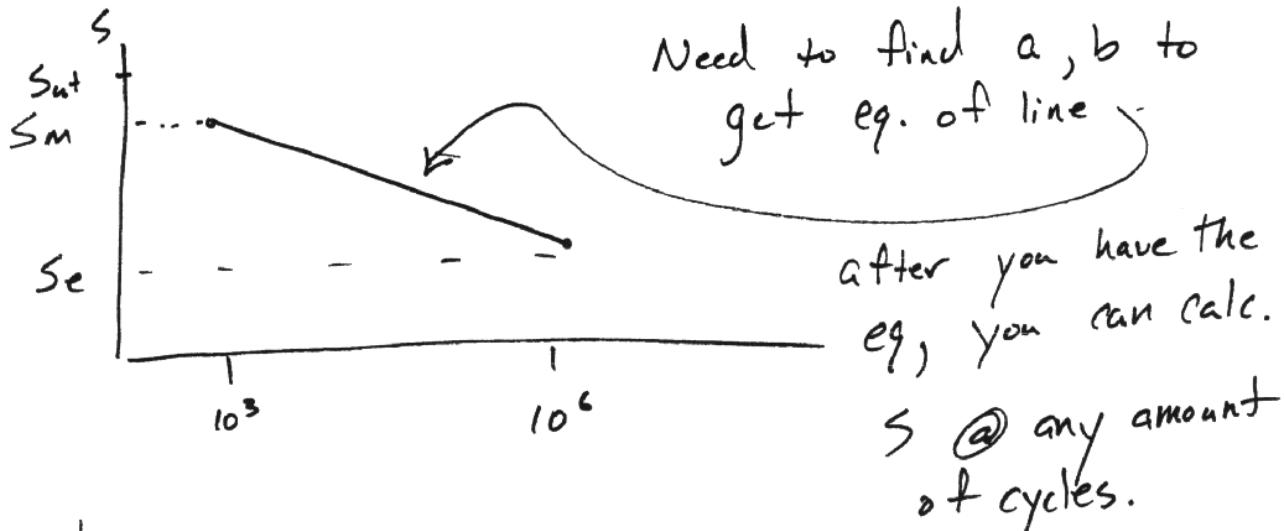
$$= (1)(1)(0.679)(1)(1)(30) = 20.398 \text{ ksi}$$

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$$S_m = 0.9 S_u + 54 \text{ ksi} \quad (N = 10^3 \text{ cycles})$$

$$S_f = a N^b \quad N = 10^6 \implies z = -3.000$$

### S-N diagram



$$b = \frac{1}{z} \log \frac{S_m}{S_e} \Rightarrow b = -0.1409$$

$$a = \frac{S_m}{(10^3)^b} \Rightarrow a = 142.955 \text{ ksi}$$

Now use:  $N_a = \left( \frac{S_f}{a} \right)^{\frac{1}{b}}$  to find # of cycles to failure

$$= \left( \frac{42.954}{142.955} \right)^{-0.1409} = \underline{\underline{5.1 \times 10^3 \text{ cycles}}}$$

(Case (b))

$$M = T/2 \Rightarrow \sigma_x = \frac{Mac}{I} = 12.5 \text{ ksi}$$

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as before,  $\sigma_a' = 12.5 = \sigma_m'$

$$S_f = \frac{\sigma_a' S_{ut}}{S_{ut} - \sigma_m'} \Rightarrow S_f = 15.9 \text{ ksi}$$

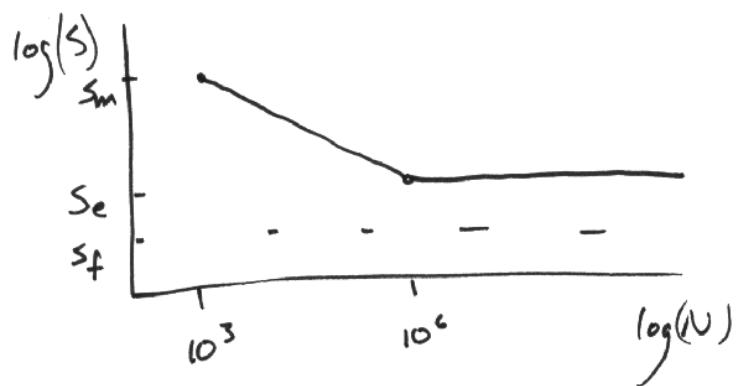
using a, b as found before

$$N_b = \left( \frac{S_f}{a} \right)^{\frac{1}{b}} = \left( \frac{15.9}{142.955} \right)^{\frac{1}{0.1409}}$$

$$\underline{N_b = 6 \times 10^6 \text{ cycles}}$$

Note: This answer is nonsense!!

Once you see that  $S_f < S_e$  you  
know that the life is infinite.



**PROBLEM 6-27**

**Statement:** A storage rack is to be designed to hold the paper roll of Problem 6-8 as shown in Figure P6-12. Determine a suitable value for dimension  $a$  in the figure for an infinite-life fatigue safety factor of 2. Assume dimension  $b = 100 \text{ mm}$  and that the mandrel is solid and inserts halfway into the paper roll.

- (a) The beam is a ductile material with  $S_{ut} = 600 \text{ MPa}$ .
- (b) The beam is a cast-brittle material with  $S_{ut} = 300 \text{ MPa}$ .

<b>Units:</b>	$N := \text{newton}$	$kN := 10^3 \cdot N$	$MPa := 10^6 \cdot Pa$	$GPa := 10^9 \cdot Pa$
<b>Given:</b>	Paper roll dimensions	$OD := 1.50 \cdot m$	Roll density	$\rho := 984 \cdot \text{kg} \cdot m^{-3}$
		$ID := 0.22 \cdot m$	Design safety factor	$N_{fd} := 2$
		$L_{roll} := 3.23 \cdot m$		
	Ductile tensile strength	$S_{uta} := 600 \cdot MPa$	Brittle tensile strength	$S_{utb} := 300 \cdot MPa$

**Assumptions:** The paper roll's weight creates a concentrated load acting at the tip of the mandrel. The mandrel's root fits tightly in the stanchion so it can be modeled as a cantilever beam. The mandrel is machined, reliability is 90%, and it operates at room temperature.

**Solution:** See Figure 6-27 and Mathcad file P0627.

1. Determine the weight of the roll and the length of the mandrel.

$$\text{Weight} \quad W := \frac{\pi}{4} \cdot (OD^2 - ID^2) \cdot L_{roll} \cdot \rho \cdot g \quad W = 53.9 \text{ kN}$$

$$\text{Length} \quad L_m := 0.5 \cdot L_{roll} \quad L_m = 1.615 \text{ m}$$

2. The maximum moment occurs at a section where the mandrel root leaves the stanchion and is

$$M_{max} := W \cdot L_m \quad M_{max} = 87.04 \text{ kN} \cdot m$$

4. The dynamic loading is repeated from 0 to  $M_{max}$  on each stress cycle, thus

$$M_{min} := 0 \text{ kN} \cdot m$$

5. Part (a) - Calculate the alternating and mean components of the bending moment.

$$M_a := \frac{M_{max} - M_{min}}{2} \quad M_a = 43520 \text{ eN} \cdot m$$

$$M_m := \frac{M_{max} + M_{min}}{2} \quad M_m = 43520 \text{ eN} \cdot m$$

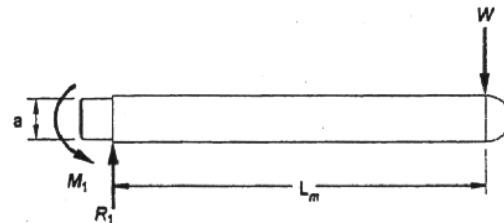
6. Determine the unmodified endurance limit.  $S'_e := 0.5 \cdot S_{uta}$   $S'_e = 300 \text{ eMPa}$

7. Calculate the endurance limit modification factors for a nonrotating rectangular beam.

$$\text{Load} \quad C_{load} := 1$$

$$\text{Size} \quad A_{95}(a) := 0.010462 \cdot a^2$$

$$d_{equiv}(a) := \sqrt{\frac{A_{95}(a)}{0.0766}}$$



**FIGURE 6-27**  
Free Body Diagram used in Problem 6-27

$$C_{size}(a) := 1.189 \cdot \left( \frac{d_{equiv}(a)}{mm} \right)^{-0.097}$$

Surface       $A := 4.51$        $b := -0.265$       (machined)

$$C_{surf} := A \cdot \left( \frac{S_{uta}}{MPa} \right)^b$$

$$C_{surf} = 0.828$$

Temperature       $C_{temp} := 1$

Reliability       $C_{reliab} := 0.897$       ( $R = 90\%$ )

8. Calculate the modified endurance limit.

$$S_e(a) := C_{load} \cdot C_{size}(a) \cdot C_{surf} \cdot C_{temp} \cdot C_{reliab} \cdot S'_e$$

9. We can now determine the minimum required diameter,  $a$ . Using the distortion-energy failure theory with the modified Goodman diagram, the bending stress will also be the only nonzero principal stress, which will also be the von Mises stress. Assuming a Case 3 load line, use equation (6.18e) to determine the factor of safety. Guess  $a := 100 \cdot mm$ .

$$\text{Bending stress} \quad \sigma = \frac{Mc}{I} = M \cdot \frac{a}{2} \cdot \frac{64}{\pi \cdot a^4} = \frac{32 \cdot M}{\pi \cdot a^3}$$

Given

$$N_{fd} = \frac{\pi \cdot a^3}{32} \cdot \frac{S_e(a) \cdot S_{uta}}{M_a \cdot S_{uta} + M_m \cdot S_e(a)}$$

$$a := \text{find}(a) \quad a = 186.864 \cdot mm$$

Round this up to the next higher even value       $a := 190 \cdot mm$

Using this value of  $a$ , the values of the functions of  $a$  are:

$$C_{size}(a) = 0.787 \quad S_e(a) = 175.371 \cdot MPa$$

The realized safety factor is

$$N_{fa} := \frac{\pi \cdot a^3}{32} \cdot \frac{S_e(a) \cdot S_{uta}}{M_a \cdot S_{uta} + M_m \cdot S_e(a)} \quad N_{fa} = 2.1$$

10. Part (b) - Determine the unmodified endurance limit.       $S'_e := 0.4 \cdot S_{utb}$        $S'_e = 120 \cdot MPa$

11. Calculate the endurance limit size modification factor for a nonrotating rectangular beam.

$$\text{Size} \quad A_{95}(a) := 0.010462 \cdot a^2 \quad d_{equiv}(a) := \sqrt{\frac{A_{95}(a)}{0.0766}}$$

$$C_{size}(a) := 1.189 \cdot \left( \frac{d_{equiv}(a)}{mm} \right)^{-0.097}$$

12. Calculate the modified endurance limit.

$$S_e(a) := C_{load} \cdot C_{size}(a) \cdot C_{surf} \cdot C_{temp} \cdot C_{reliab} \cdot S'_e$$

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13. We can now determine the minimum required diameter,  $a$ . Using the distortion-energy failure theory with the modified Goodman diagram, the bending stress will also be the only nonzero principal stress, which will also be the von Mises stress. Assuming a Case 3 load line, use equation (6.18e) to determine the factor of safety. Guess  $a := 100 \cdot mm$ .

Bending stress

$$\sigma = \frac{M \cdot c}{I} = M \cdot \frac{a}{2} \cdot \frac{64}{\pi \cdot a^4} = \frac{32 \cdot M}{\pi \cdot a^3}$$

Given

$$N_{fd} = \frac{\pi \cdot a^3}{32} \cdot \frac{S_e(a) \cdot S_{utb}}{M_a \cdot S_{utb} + M_m \cdot S_e(a)}$$

$$a := \text{find}(a)$$

$$a = 251.687 \cdot mm$$

Round this up to the next higher even value

$$a := 252 \cdot mm$$

Using this value of  $a$ , the values of the functions of  $a$  are:

$$C_{size}(a) = 0.766$$

$$S_e(a) = 68.253 \cdot MPa$$

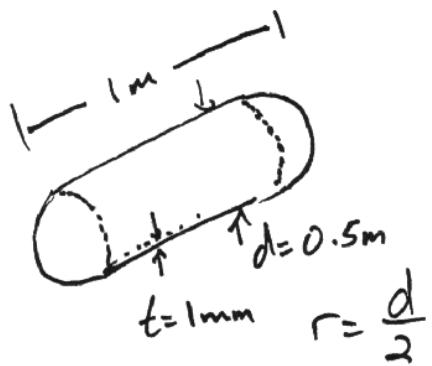
The realized safety factor is

$$N_{fb} := \frac{\pi \cdot a^3}{32} \cdot \frac{S_e(a) \cdot S_{utb}}{M_a \cdot S_{utb} + M_m \cdot S_e(a)} \quad N_{fb} = 2.0$$

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A cylindrical tank w/ hemispherical ends is required to hold 150 psi of air at room temp. The pressure cycles from zero to maximum. Steel has  $S_{UT} = 500 \text{ MPa}$ . Find infinite life safety factor if  $d = 0.5 \text{ m}$  and 1 mm wall thickness,  $L = 1 \text{ m}$



• Thin wall pressure vessel

$$\sigma_t = \frac{P r}{t} = 258.55 \text{ MPa}$$

$$\sigma_r = 0 \text{ MPa}$$

$$\sigma_a = \frac{P r}{2t} = 129.28 \text{ MPa}$$

$P = 150 \text{ psi} = 1034 \text{ kPa}$

These are already principals (shear = 0)

$$\text{so } \sigma_1 = \sigma_t \quad \sigma_2 = \sigma_a \quad \sigma_3 = \sigma_r$$

Find von Mises effective stresses

$$\sigma'_{\min} = 0 \text{ MPa} \quad \sigma'_{\max} = \sqrt{\sigma_1^2 - \sigma_1 \sigma_2 + \sigma_2^2} = 224.3 \text{ MPa}$$

$$\text{So } \sigma'_a = \frac{\sigma'_{\max} - \sigma'_{\min}}{2} = 112.151 \text{ MPa}$$

$$\sigma'_M = \frac{\sigma'_{\max} + \sigma'_{\min}}{2} = 112.151 \text{ MPa}$$

Calculate the unmodified endurance limit:

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$$S_e' = 0.5 S_{ut} = 250 \text{ MPa}$$

modification factors:

$$C_{load} = 0.7 \text{ (axial)}$$

$$C_{reliab} = 0.659 \text{ (99.999%)}$$

$$C_{size} = 1.0 \text{ (axial)}$$

$$C_{surf} = A \left( \frac{S_{ut}}{\text{MPa}} \right)^b$$

$$C_{temp} = 1$$

$$= 0.869$$

$$\boxed{\begin{aligned} A &= 4.51 \\ b &= -0.261 \\ \text{Machined} \end{aligned}}$$

$$\begin{aligned} S_e &= C_{load} C_{size} C_{surf} C_{temp} C_{reliab} S_e' \\ &= (0.7)(1.0)(0.869)(1)(0.659)(250) = 100.2 \text{ MPa} \end{aligned}$$

$$N_f = \frac{S_e S_{ut}}{\sigma_a' S_{ut} + \sigma_m' S_e} \quad (\text{case 3})$$

$$N_f = 0.74$$